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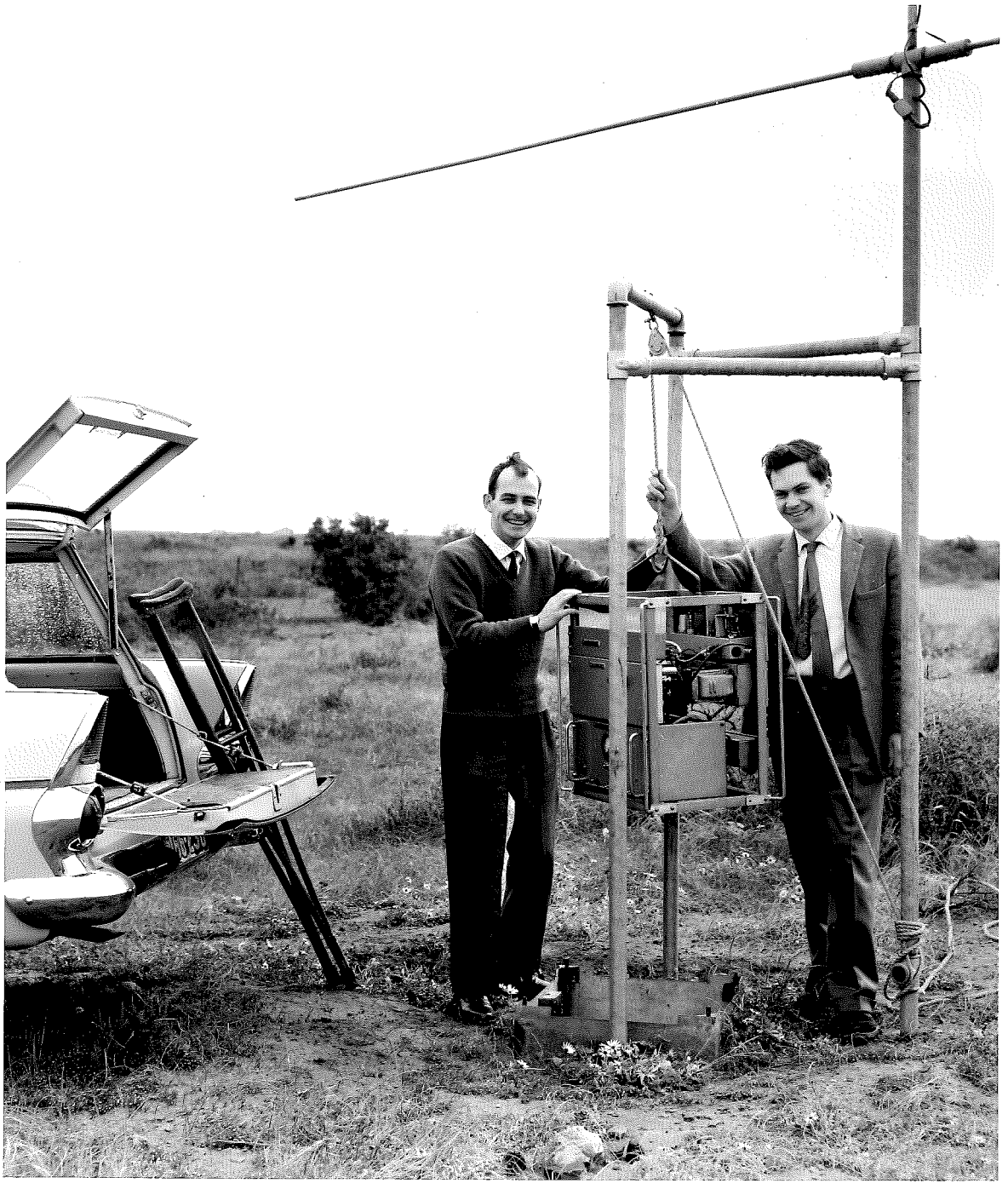
ATMOSPHERIC TURBULENCE  
IN THE  
METEOR REGION

by

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Frontispiece - The author ( left ) and colleague

C. S. Nilsson with the Direk outstation.

SUMMARY

A re-examination is made of work published by others on turbulence in the upper atmosphere at heights ranging from 80 to 100km. These results are found to be explicable in terms of a modified Kolmogoroff spectrum.

A method is described for obtaining data for the investigation of turbulence in the meteor region from the line of sight drifts of ionized meteor trails as measured by a spaced-station radio technique. The application and limitations of this technique are discussed in some detail.

The results of a thirteen month survey of atmospheric turbulence at meteor heights from December, 1960 to December, 1961 are presented. Direct observations are made on the scales from 100 metres to 3km, and of the characteristic velocity of the energy bearing eddies. The results from this survey are also found to follow the Kolmogoroff law, modified by the effects of a vertical shear and an eddy subrange arising from the action of buoyancy forces. At scales less than 1km the buoyancy subrange predominates. If the measured wind velocity differentials pertaining to this small-scale region are interpreted in terms of a Kolmogoroff spectrum, the turbulent dissipation rate  $\epsilon$  is found to be approximately 50

ergs/gm/sec, which is of the same order as the 70 ergs/gm/sec found by others from the photographic observations of the diffusion of sodium trails and visual meteor trails (The maximum scale size that can be measured by this photographic technique is limited by visibility to about 1km). Once allowance is made for the buoyancy subrange, the scales to 6km are probably isotropic, and subject to a small vertical shear, while the large scales, whose characteristic size is determined by extrapolation from the scales observed, are distinctly anisotropic, with a vertical scale of 6km and a horizontal scale of the order of 100km.

The turbulent dissipation rate calculated from the change in velocity difference with separation for the scales from 1 to 3km shows a marked seasonal variation, with a maximum of approximately 350 ergs/gm/sec in spring and autumn, and a minimum of about 180 ergs/gm/sec in summer and winter. An attempt has been made to ascribe an absolute value to the turbulent dissipation rate via an expression which involves the height shear. Because of the low echo rate achieved, no information about the diurnal variation in  $\epsilon$  is forthcoming. However, since the diurnal variation of the characteristic velocity of the energy bearing eddies is of the order of 2:1, a diurnal variation in the turbulent dissipation rate is expected. The dissipation mechanism is

somewhat obscure, since energy is extracted from all scales by the buoyancy subrange. The origin of the large-scale anisotropic eddies is still in doubt, although there is some correlation between the characteristic velocities of these eddies and the diurnal variation of the mean wind.

A preliminary estimate of the turbulence Reynolds number appropriate to the meteor region is given as  $7 \times 10^3$ .

## Preface

To the best of the author's knowledge and belief, this thesis contains no material previously published or written by another person, except when due reference is made in the text.

Portions of Chapters III, VI, VII and VIII are claimed as original work, as are Chapters IX, X and XI. The author was responsible for the design and construction of the Three-station Wind Recording Rack and the equipment housed at the outstations, for the design of the main station receivers, for the reading of the three-station wind films, and for the reduction of the data so obtained. All programming of the mean wind and turbulence data for execution on the University's IBM 1620 electronic computer was the work of the author.

This thesis contains no material which has been presented by the author for the award of any other degree or diploma in any University.

### Acknowledgements

In a project as large as the 1961 orbit and turbulence survey, considerable teamwork is required to ensure its satisfactory execution. The survey was suggested by and carried out under the supervision of Dr. W. G. Elford and Dr. A. A. Weiss, with both of whom the author has had many helpful discussions. The ancillary data pertaining to the specification of the geometry of each meteor trail used in the survey was provided by Carl S. Nilsson, who was also responsible for the housing of the outstation equipment and, together with the author, for most of the equipment maintenance throughout the year. The mean wind measuring equipment was constructed and installed at St. Kilda by Mr. E. J. Welsby, who also assisted with the maintenance of the transmitters. Many points of equipment design procedure were clarified by Dr. E. L. Murray. The drawing of the diagrams presented in this thesis was the work of Miss Mary Chapman, who, together with Miss Joan Allister, processed the 46,000 feet of film used in the 1961 survey, and read and computed that portion of the data pertaining to the mean winds. The cooperation of Mr. G. Tomlinson and Miss Ann Millbank of Photographic Services is grate-



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## CHAPTER I.

### Turbulence in the Ionosphere

#### 1.1 Historical

Ionospheric irregularities were first studied by Ratcliff and Pawsey at Cambridge some 30 years ago. Recent discussion of turbulence in the ionosphere dates from the discovery about ten years ago of the phenomenon of VHF scatter transmission. Until then, all irregularities were assumed to have a Gaussian distribution of electron density, and to occur in an inherently stable ionosphere. While experiments were being carried out in propagation of radio waves beyond the visible horizon at VHF, scatter from the troposphere was evident; this was readily explained in terms of atmospheric turbulence. Since ionospheric scatter also occurs at VHF, ionospheric phenomena were further investigated in the light of turbulence theory.

#### 1.2 General

Mechanisms of two types can lead to turbulence. The breakdown of primary instabilities in ionization can give rise to turbulent "motions" of the ionization; secondary instabilities can arise from geophysical motions, which will also cause

turbulent motion of the ionization. The first gives rise to apparent motions, these being caused by changes in the ionization density, and not necessarily related to actual physical transportation of the reflecting particles. The second type, which represents a true motion, may or may not be that of the non ionized medium, perturbation of motion being possible if there are strong E or H fields present. In most of the above cases, the appearance of turbulence should be irregular in time, space, amplitude and spectrum.

Available theories treat only turbulent motions that are homogenized, "isotropized", quasi-stationary and associated with a single large scale motion. In forward scatter experiments, the question whether the statistical sampling along the path of the beam of a given wavelength corresponds to the theoretical model remains unanswered. With other causes of electron distribution irregularities besides turbulence (diffusing meteor trails, wave like motions) it does not appear surprising that theoretical spectra and those determined from experiment disagree. Nor is it surprising that there is no agreement amongst workers in the 80 - 100 KM region as to the interpretation of radio data in terms of the velocities of the small scale motions.

It is hoped that the contents of this thesis will help to resolve some of these anomalies in interpretation.

Before proceeding to a detailed discussion of the problems peculiar to the 80 to 100 KM region of the upper atmosphere, a general discussion of atmospheric turbulence is pertinent.

### 1.3 Atmospheric Turbulence

In recent years there has been an increasing acceptance by meteorologists of the view first put forward by Defant in 1921, that surface phenomena can be explained by an atmosphere which is essentially a quasi-horizontal turbulent flow field. Since 1948, when von Weizsacker and Heisenberg independently extended the theory of turbulence proposed by Kolmogoroff in 1941, considerable interest has been shown by fluid dynamicists in small scale turbulence. In 1953, Batchelor published his "Theory of Homogeneous Turbulence," giving a comprehensive account of the modern theory of homogeneous, and, in most applications, isotropic turbulence. Most of these amplifications of the Kolmogoroff theory were applied to turbulent flows as encountered in aerodynamics and hydraulic laboratories. Here, the turbulence scales were indeed small, their size being dictated by the dimensions of the apparatus. However, McCready (1953) and Taylor (1955) were able to show that much of this work could be extended to the investigation of turbulent fields occurring naturally in the atmosphere. They found that when autocorrelations in horizontal wind velocity

components at two points are measured in the lower atmosphere, they show the particular dependence on the distance between the two points which is characteristic of the inertial sub-range of turbulence scales, up to unexpectedly large separations - often as much as several times the height of observation.

The phenomena of the lower atmosphere have been summarized by Sheppard (1959). In the lower atmosphere, the large scale turbulence increases with height to the tropopause, and then decreases, whereas the small scale turbulence (as measured, for example, on an aircraft accelerometer) decreases with height to the upper tropopause, and then becomes more constant. The pattern of convection is partly responsible for the decrease with height of the small scale turbulence. The Kolmogoroff theory is unable to account for these changing relationships between large and small scale turbulence, and considerable caution must be exercised in inferring the properties of small scale turbulence from observations only on the large scale motions.

#### 1.4 The Stability of the Atmosphere

As pointed out by Stewart (1959), the classical Reynolds number is rarely an important parameter in geophysical motions. By and large it is stability that determines whether or not a given region of the atmosphere is turbulent. It is most un-

usual for the velocity gradient not to be great enough to sustain turbulence, although, of course, the turbulent intensity may be exceedingly low. Thus we may say that, if the atmospheric region under discussion is unstable, or in neutral stability on a large scale, turbulence will occur. Almost never will the velocity gradient be so small that turbulence is inhibited by too low a Reynolds number.

In regions where the structure is stable, as, for example, at altitudes above the mesopause at approximately 80KM, where there is the stabilizing influence of a positive temperature gradient, the inhibition of turbulence is possible, although turbulence will occur if the velocity gradient is great enough. The temperature increase with height will, of course, exert a stabilizing influence only on vertical motions. Large scale horizontal temperature gradients can and do exist, giving rise to geophysical winds that are predominantly horizontal. The magnitudes of these winds are of the order of tens of meters per second, and thus turbulence is expected to be a characteristic of these motions. Because of the vertical stability, the large scale eddies characterizing this turbulent motion would be expected to be distinctly anisotropic, and this has been verified experimentally (see Chapter 3).

### 1.5 Turbulence at Meteor Heights

To date, the application of turbulence theory to the

atmosphere at meteor heights has met with only diffident success. Some of the apparent inconsistencies are the result of variations in the interpretation of the statistical equations of homogeneous turbulence, and for this reason the derivation of the relationships used in the reduction of the Adelaide wind results is treated in some detail in Chapter II. The main text of this Chapter is taken from the excellent exposition given by Stanley Corrsin at the Symposium on the Fluid Mechanics of the Ionosphere (New York, 1959). Material is also drawn from Batchelor's "Theory of Homogeneous Turbulence" (Cambridge University Press, 1953).

## CHAPTER II.

### Turbulence Theory

#### 2.1 Definition of Turbulence

Unfortunately, there is no universally accepted definition of turbulence. For the purposes of this thesis, a definition of turbulence given by Stewart at the International Symposium on Fluid Mechanics in the Ionosphere (New York, 1959) seems most appropriate.

"A fluid is said to be turbulent if each component of the vorticity is distributed aperiodically in time and space, if the flow is characterized by the transfer of energy from larger to smaller scales of motion, and if the mean separation of neighbouring fluid particles tends to increase with time. This definition excludes all two dimensional flows, as well as such phenomena as vortex sheets, whirlpools, convection cells and internal waves."

The most commonly accepted criterion of the stability of fluid flow is the dimensionless Reynolds number, the ratio of the inertial forces to the viscous forces. If the viscous forces are of the order of, or exceed the inertial forces, then any fluctuation of the mean flow will be quickly damped



out by viscous dissipation. When the Reynolds number exceeds a certain critical value, the flow becomes unstable for small disturbances, and so a superimposed flow in the nature of a fluctuation with characteristic velocity and dimensions is generated. As mentioned in Chapter I, Section 4, the critical value of the Reynolds number in the atmosphere depends not only on the inertial and viscous forces, but also on other factors influencing the stability e.g. a positive temperature gradient with height, which severely inhibits vertical motion.

Kolmogoroff (1941) applies the fluctuation principle to each superimposed flow, so that with increasing Reynolds number each in turn becomes unstable, generating further disturbances. At sufficiently high Reynolds number, fluctuations of all magnitudes are present. Further, as the energy of the larger superimposed flows passes down through the smaller scale disturbances, the direct influence of the larger scales is lost, so that all scales smaller than a certain critical size have statistical properties which, to a large extent, are purely random. That is, these smaller scales are both homogeneous and isotropic.

## 2.2 The Kolmogoroff Turbulence Spectrum

This may be summarized as follows:-

1. The large scale fluctuations, which carry the turbulent energy extracted from the mean flow. These are included

in the inertial range of scales, the term "inertial" implying that energy is received from higher scales and transferred to lower scales with negligible loss.

2. The isotropic inertial region. If the Reynolds number is high enough, a range of scales will exist at the lower end of the inertial region which does not contribute to the energy dissipation, but has random statistical properties. Such an isotropic inertial region rarely exists in wind tunnel and hydraulics experiments, because the Reynolds number involved is usually too low. However, one may expect to find such a region in a turbulent atmosphere.

3. The region of viscous dissipation. The energy of the inertial region is passed on to these smaller scale fluctuations, and is dissipated by viscous forces. The statistical properties of this state depend only on two parameters, viz.

$\epsilon$ , the rate at which energy is handed down from the larger scales, and

$\nu$ , the kinematic viscosity, which determines the rate at which the kinetic energy of the viscous scales can be converted to heat.

### 2.3 The Turbulent Energy Spectrum $\epsilon$

This is the most important property of any turbulent region; however, direct measurement of energies is usually not physically possible, and therefore the energy spectrum is

deduced from the turbulent velocity and scale. To establish the relationship between velocity and scale, use is made of the spacial correlation function

$$\mu(\xi) = \overline{u(x) u(x+\xi)}$$

2.3.1

where  $u(x)$  is the turbulent velocity at  $x$ , and  $u(x+\xi)$  the turbulent velocity at a point distance  $\xi$  from  $x$ . The bar symbol denotes an average taken over a time large compared with the time constant of the turbulence scale characterized by  $\mu$ . This time constant or characteristic time of a turbulent flow field is the time taken for the turbulent energy carried by the large scale fluctuations to be passed down through the inertial range and finally dissipated in the viscous region.

Note that  $\mu$  is compounded from turbulent velocities only; all mean flow and shear components must be subtracted from the measured velocity to define the  $\mu$  field. The energy density spectrum is the Fourier transform of the spacial correlation function, taken over all real space. i.e.

$$\mathcal{E}(\kappa) = \frac{1}{8\pi^3} \iiint_{-\infty}^{\infty} \mu(\xi) e^{-i(\kappa \cdot \xi)} dV_{\xi}(\xi)$$

2.3.2

Here, the spectral density has been defined in phase space of dimension  $\kappa$ . It is often convenient, in theoretical arguments, to refer to the energy associated with an eddy of

size  $\kappa^{-1}$  (Dimension of length in physical space). In the terminology of those working on problems in turbulence, an eddy simply means a volume of fluid moving more or less coherently with respect to the mean flow. The eddy motion need not be, and usually is not of a rotating character. The term eddy can frequently be inter-changed with the more cumbersome expression "scale of motion".

#### 2.4 Practical Correlation Functions

In experiments conducted on turbulent flow fields, it is usual to employ the one dimensional lateral and transverse correlation functions  $f(\xi)$  and  $g(\xi)$ , relating to the turbulent velocity components parallel and perpendicular respectively to the separation  $\xi$ , where  $\xi$  now becomes a one dimensional length.

By definition

$$g(\xi) = \frac{\overline{u(x) u(x+\xi)}}{\left[ \overline{u^2(x) u^2(x+\xi)} \right]^{\frac{1}{2}}} \quad 2.4.1$$

where  $u$  are turbulent velocity components measured normal to  $\xi$ .

The accuracy of such a determination for a given  $\xi$  is given by

$$\partial g = \frac{1-g^2}{\sqrt{(n-1)}}$$

where  $n$  is the number of velocity pairs having separation  $\xi$ .

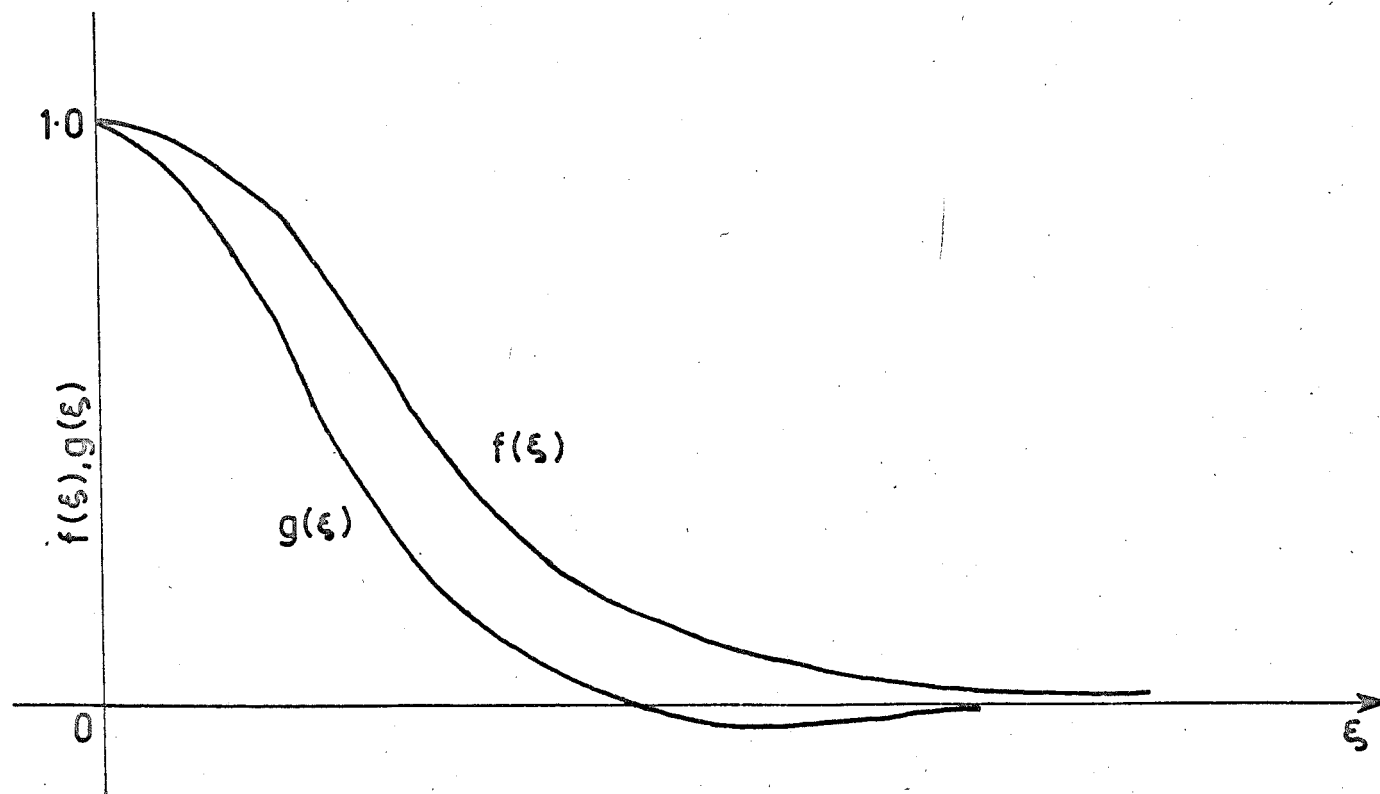


FIG.1 - GENERAL SHAPE OF CORRELATION CURVES.

Similarly

$$f(\xi) = \left[ \frac{\overline{v(x) v(x+\xi)}}{\overline{v^2(x) v^2(x+\xi)}} \right]^{\frac{1}{2}}$$

2.4.2

for the components  $v$  parallel to  $\xi$ .

The form of these correlation functions is shown graphically in Fig. 1.

For these spacial correlation functions to be readily applied, certain restrictive conditions are necessary. The definition of  $g(\xi)$  confines the turbulent velocity vector only to the perpendicular plane, and so the transverse correlation assumes specificity for random sampling only if the turbulent field is isotropic. If the direction of  $\xi$  is fixed in space, two dimensional isotropy will suffice. In isotropic turbulence

$$\left[ \overline{u^2(x) u^2(x+\xi)} \right]^{\frac{1}{2}} = \left[ \overline{v^2(x) v^2(x+\xi)} \right]^{\frac{1}{2}} \\ = U_0^2.$$

2.4.3

$U_0$  is the velocity characteristic of the energy bearing eddies.

The introduction of the equation of continuity for incompressible fluids leads to the relation

$$g(\xi) = f(\xi) + \frac{1}{2} \xi \frac{\partial f}{\partial \xi}$$

2.4.4

between the functions  $f$  and  $g$ ,

or, if the turbulence is isotropic in two dimensions only

$$g(\xi) = f(\xi) + \xi \frac{\partial f}{\partial \xi} \quad 2.4.5$$

[von Karman and Howarth (1938)]

From 2.4.1 and 2.4.2, we have for isotropic turbulence

$$f(0) = g(0) = 1 \quad 2.4.6$$

this being the maximum value of  $f$  and  $g$ .

i.e.

$$\left[ \frac{\partial f}{\partial \xi} \right]_{\xi=0} = \left[ \frac{\partial g}{\partial \xi} \right]_{\xi=0} = 0 \quad 2.4.7$$

## 2.5 The Characteristic Dissipation Length $\lambda$ .

Expansion of  $f(\xi)$  in the neighbourhood of  $\xi=0$  in the form of a Taylor's Series yields

$$f(\xi) = 1 + \frac{1}{2} f_0'' \xi^2 + o(\xi^4) \quad 2.5.1$$

where the dash symbol denotes differentiation with respect to  $\xi$ . Since the dimensions of  $f''(\xi)$  is  $L^{-2}$ , it is common practice to introduce the length  $\lambda$ , called the characteristic dissipation length, and defined by

$$f_0'' = - \frac{1}{\lambda^2}$$

Thus 2.5.1 becomes

$$f(\xi) = 1 - \frac{1}{2\lambda^2} \xi^2$$

2.5.2

in the neighbourhood of  $\xi=0$ .

Similarly, from 2.4.4,

$$g(\xi) = 1 - \frac{1}{\lambda^2} \xi^2$$

2.5.3

near  $\xi=0$ .

The significance of the parameter  $\lambda$ , which is a measure of the energy characteristic of the large scale eddies, is treated in more detail in Section 7.

## 2.6 The One Dimensional Turbulent energy spectrum E.

Using the one dimensional length  $\xi$ , we may formulate a one dimensional energy spectrum equivalent to 2.3.2, using  $f(\xi)$  and  $g(\xi)$  as the velocity correlation functions.

Since

$$f(\xi) = \frac{\mu(\xi)}{U_0^2}$$

the longitudinal one dimensional spectrum function becomes

$$E(\kappa) = \frac{1}{2\pi} \int_{-\infty}^{\infty} U_0^2 f(\xi) \cos(\kappa \cdot \xi) d\xi \quad 2.6.1$$

where  $\kappa$  now defines a one dimensional phase space.



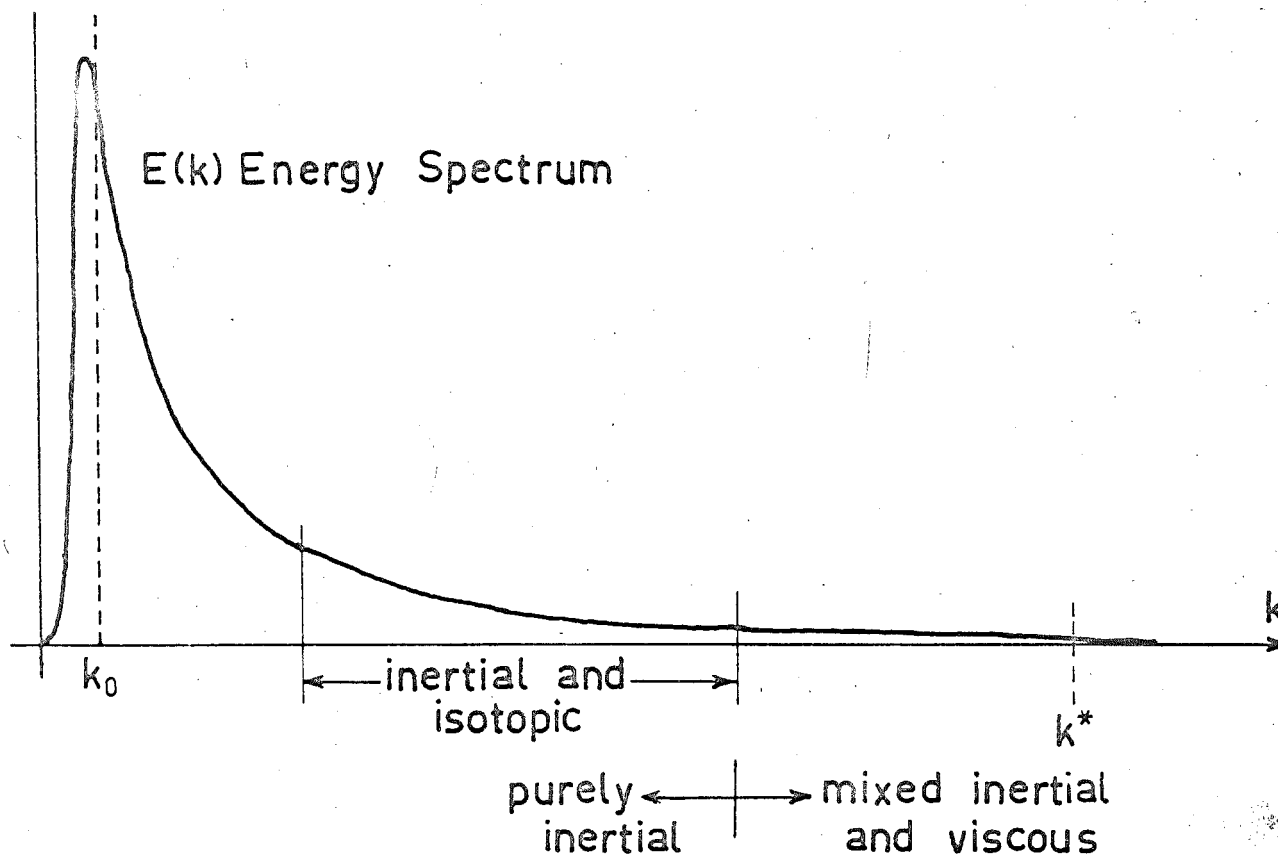


FIG.2 - SPECTRAL RANGES IN TURBULENCE OF MODERATE REYNOLDS NUMBER.

Similarly, the lateral one dimensional spectrum function is

$$E(\kappa) = \frac{1}{2\pi} \int_{-\infty}^{\infty} \bar{u}_0^2 g(\xi) \cos(\kappa \cdot \xi) d\xi$$

2.6.2

The form of the one dimensional energy spectrum for moderate Reynolds number is shown in Fig. 2. The peak of the curve occurs at  $\kappa = \kappa_0$ , and this is the characteristic scale of the energy bearing eddies. G. I. Taylor has defined the size of these energy bearing scales in terms of the characteristic integral scales

$$L_0 = \frac{1}{\bar{u}_0^2} \int_0^{\infty} \mu(\xi) d\xi$$

Inserting  $f(\xi)$  and  $g(\xi)$ , with regard given to 2.4.4 we have

$$L_0 = \int_0^{\infty} f(\xi) d\xi = 2 \int_0^{\infty} g(\xi) d\xi$$

2.6.3

Thus an alternative definition of  $L_0$  is that it is the maximum scale over which velocities are appreciably correlated.

Batchelor (1953) has shown that 2.6.3 can be expressed as

$$L_0 = \frac{3\pi}{4} \frac{\int_0^{\infty} \kappa^{-1} E(\kappa) d\kappa}{\int_0^{\infty} E(\kappa) d\kappa}$$

which rationalizes the term "average eddy of size  $\kappa^{-1}$ " as used

by fluid dynamicists.

At the other end of the turbulence spectrum, at extremely large  $\kappa$ , is the region of viscous dissipation. Kolmogoroff has defined  $\kappa^*$ , the characteristic eddy of this region, as

$$\kappa^* = \left[ \frac{\epsilon}{\nu^3} \right]^{\frac{1}{4}}$$

2.6.4

This assumes that at this characteristic eddy size all the energy transferred from the inertial region is being dissipated in purely viscous forces. In physical space, the size of the characteristic eddy of the Kolmogoroff microscale is denoted by  $\eta$  ( $=\kappa^{*-1}$ ).

If the lower boundary of the mixed inertial-viscous region is at very large  $\kappa$ , there may exist a region both inertial and isotropic. For such a region Kolmogoroff postulated that  $E(\kappa)$  depends only on the rate of energy flux through that part of the spectrum. This must equal the total dissipation rate  $\epsilon$ .

The only form dimensionally possible is

$$E(\kappa) = \alpha \epsilon^{\frac{2}{3}} \kappa^{-\frac{5}{3}}$$

2.6.5

where  $\alpha$  is of the order of unity.

Laboratory experiments indicate that this law holds up to wave numbers about as high as

$$\kappa_{\text{lim}} = 0.2\kappa^*$$

2.6.6

If the actual spectrum at extremely high Reynolds numbers is replaced by a purely inertial spectrum as defined (equation 2.5.8) truncated at  $\kappa_0 (= L_0^{-1})$  and at  $\kappa^*$  ( $=\eta^{-1} \gg L_0^{-1}$ ) the total dissipation

$$\epsilon = 2\nu \int_0^{\infty} \kappa^2 E(\kappa) d\kappa \quad 2.6.7$$

can be estimated as

$$\epsilon \approx 2\alpha\nu\epsilon^{\frac{2}{3}} \int_{\kappa_0}^{\kappa^*} \kappa^{\frac{1}{3}} d\kappa$$

Neglecting  $\kappa_0^{\frac{4}{3}}$  relative to  $\kappa^{*\frac{4}{3}}$  we have

$$\alpha \approx \frac{2}{3}$$

This paradoxical estimation of dissipation from a non-dissipative form is justifiable as long as the true  $E(\kappa)$  decreases sufficiently rapidly for  $\kappa > \kappa^*$ .

Having determined the form of  $E(\kappa)$  for the inertial range of eddy sizes, the form of the correlation function appropriate to this region can now be established.

Since

$$E(\kappa) = \frac{1}{2\pi} \int_{-\infty}^{\infty} U_0^2 f(\xi) \cos(\kappa \cdot \xi) d\xi \quad (\text{equation 2.6.1})$$

it follows that

$$f(\xi) = \frac{1}{U_0^2} \int_0^{\infty} E(\kappa) \cos(\kappa \cdot \xi) d\kappa$$

which is, on substitution of 2.5.8 with  $\alpha = \frac{2}{3}$ ,

$$f(\xi) = \frac{2}{3} \frac{\epsilon^{\frac{2}{3}}}{U_0^2 K_0} \int_{K_0}^{\infty} K^{-\frac{5}{3}} \cos(K_0 \xi) dK$$

and has a solution of the form

$$1 - f(\xi) = a \xi^{\frac{2}{3}}$$

2.6.8

where  $a$  is a constant.

Introduction of the relations

$$g(\xi) = f(\xi) + \frac{1}{2} \xi \frac{\partial f}{\partial \xi}$$

(equation 2.4.4)

yields

$$g(\xi) = 1 - \frac{4}{3} a \xi^{\frac{2}{3}}$$

for three dimensional isotropy, or from 2.4.5

$$g(\xi) = 1 - \frac{5}{3} a \xi^{\frac{2}{3}}$$

if the turbulence is isotropic in two dimensions only.

The ratio

$$\frac{1 - f(\xi)}{1 - g(\xi)}$$

2.6.9

is thus 0.75 or 0.60, according as the turbulence is two or

three dimensionally isotropic.

## 2.7 The Rate of change of Energy $\epsilon$ .

Using the dynamical equation first derived by von Karman and Howarth (1938), Batchelor has shown (1953) that the rate of change of energy

$$\epsilon = \frac{d^3 U_0^2}{dt} = 15\nu U_0^2 f'' = -15\nu U_0^2 / \lambda^2$$

2.7.1

The proportional rate of decrease of energy is

$$\frac{1}{U_0^2} \frac{dU_0^2}{dt} = -10\nu / \lambda^2$$

whence the term "dissipation length parameter" for  $\lambda$ .

## 2.8 Turbulence Reynolds Numbers.

For the isotropic inertial region there are no characteristic mean flow velocities or lengths, but a turbulence Reynolds number may be defined in terms of turbulence parameters. This will be of assistance in determining the scales of the viscous region, since, in the Kolmogoroff theory, the smaller scales see the larger scales as a form of mean motion.

A turbulence Reynolds number due to G. I. Taylor is

$$R_\lambda = U_0 \lambda / \nu$$

2.8.1

where  $U_0$  is the velocity characteristic of the energy

bearing eddies, and  $\lambda$  is the dissipation length parameter.

### 2.9 The Relationships between Eddy Sizes.

From 2.6.4

$$\eta = \left[ \frac{\nu^3}{\epsilon} \right]^{\frac{1}{4}}$$

Substituting for  $\epsilon$  from 2.7.1

$$\frac{\eta}{\lambda} = \left[ \frac{\nu^3}{15U_0^2 \lambda^2} \right]^{\frac{1}{4}}$$

In terms of the Reynolds number  $R_\lambda$ ,

$$\frac{\eta}{\lambda} = \frac{1}{15^{\frac{1}{4}} R_\lambda^{\frac{1}{2}}} \approx \frac{0.5}{R_\lambda^{\frac{1}{2}}}$$

2.9.1

A relation between  $\lambda$  and the energy bearing eddies  $L_0$  ( $=\kappa_0^{-1}$ ) can be calculated from the energy spectrum  $E(\kappa)$ , using a method similar to that employed in the determination of  $\alpha$  (equation 2.6.7 et sequ.).

The total energy

$$\frac{1}{2} U_0^2 = \int_0^\infty E(\kappa) d\kappa$$

2.9.2

Substituting 2.7.1 for  $\epsilon$ , and neglecting  $\kappa_0^{*-2/3}$  relative to  $\kappa_0^{-2/3}$  gives

$$\frac{\lambda}{L_0} = \frac{8}{R_\lambda}$$

2.9.3

## 2.10 The Relationship Between Velocity Differences and Separation in Isotropic Turbulence.

Since velocity differences are the most easily measured parameters of a turbulence spectrum, some knowledge of their behaviour with separation is advantageous.

Batchelor has shown (1953) that, if there exists an isotropic inertial region, then in that region

$$\overline{[u(x+\xi, t) - u(x, t)]^2} = 4 \int_0^{\infty} E(\kappa) \left\{ 1 - \frac{\sin(\kappa \cdot \xi)}{\kappa \cdot \xi} \right\} d\kappa$$

This may be written

$$\overline{[u(x+\xi) - u(x)]^2} = 4 \int_0^{\infty} E(\kappa) \left\{ 1 - \frac{\sin(\kappa \cdot \xi)}{\kappa \cdot \xi} \right\} d\kappa \quad 2.10.1$$

where the bar symbol now denotes an average over a time large in comparison with the time constant of the energy bearing eddies.

The integrand is suppressed by the small value of  $(1 - \frac{\sin(\kappa \cdot \xi)}{\kappa \cdot \xi})$  when  $\kappa = \kappa_0$ , and is suppressed by small  $E(\kappa)$  for  $\kappa = \kappa^*$ . The behaviour is thus dominated by the inertial range of  $\kappa$ , and thus we may write

$$\overline{[u(x+\xi) - u(x)]^2} = 4 \int_0^{\infty} \frac{2}{3} \epsilon^{\frac{2}{3}} \kappa^{-\frac{5}{3}} \left\{ 1 - \frac{\sin(\kappa \cdot \xi)}{\kappa \cdot \xi} \right\} d\kappa$$



$$= 9/5 \left[ \Gamma\left(\frac{1}{3}\right) [\epsilon \xi]^{\frac{2}{3}} \right]$$

$$= 3.2 (\epsilon \xi)^{\frac{2}{3}}$$

i.e.

$$\overline{[u(x+\xi) - u(x)]^2} = 3.2 (\epsilon \xi)^{\frac{2}{3}}$$

2.10.2

In terms of the RMS velocity differences  $\overline{\Delta v}$ ,

$$\overline{\Delta v} = 1.8 (\epsilon \xi)^{\frac{1}{3}}$$

2.10.3

### 2.11. The Time Dependent Velocity Correlation Function.

In the single station work of meteorology, the time correlation function  $g(\tau)$  is a more convenient quantity than the spacial correlation  $g(\xi)$ .

By definition, the autocorrelation function

$$g(\tau) = \frac{\overline{v(t)v(t+\tau)}}{\overline{v^2(t)}}$$

2.11.1

and similarly

$$f(\tau) = \frac{\overline{u(t)u(t+\tau)}}{\overline{u^2(t)}}$$

2.11.2

where  $v$  and  $u$  are the NS and EW components respectively of the turbulent wind.

The equivalence of the spacial and time correlation functions has provided some contentious arguments in the past, but the experience of meteorologists has been that the equivalence is established for regions sufficiently removed from the effects of orography, and Hutchings (1955) has demonstrated this from radar-sonde balloon readings taken in the upper levels of the lower atmosphere in New Zealand.

A similar power law relating the Eulerian time correlation function and the time of measurement as that determined for spacial correlations and separations has been verified by Hutchings, viz.

$$1 - f(\tau) \sim \tau^{\frac{2}{3}}$$

2.11.3

This relationship assumes, however, that the turbulent velocity is small in relation to the velocity of the mean flow. By a consideration of the one dimensional model of small scale turbulence suggested by von Weizsacker (1948) (reproduced in Section 6), Ogura (1953) has shown that, in the range where the space correlation is given by 2.6.8,  $f(\tau)$  may be expressed in the form

$$1 - f(\tau) \sim \tau^m$$

2.11.4

$m$  being a complicated function of the mean flow  $U$  and RMS turbulent velocity  $[\overline{u^2}]^{\frac{1}{2}}$  in the direction of the mean flow.

For

$$U \gg \gg [\overline{u^2}]^{\frac{1}{2}} \quad m = \frac{2}{3}$$

and for

$$U \ll \ll [\overline{u^2}]^{\frac{1}{2}} \quad m = 1$$

Either  $f(\tau)$  or  $g(\tau)$  may be used in the relation 2.11.4, the general equation being written

$$1 - R(\tau) \sim \tau^m$$

2.11.5

## 2.12 The Richardson Number.

In well developed turbulence, the gas is influenced by a force arising from instability associated with the velocity shear.

This yields a turbulent velocity

$$v_t = l_m \left[ \frac{d\bar{v}}{dh} \right]$$

where  $l_m$  is the momentum mixing length.

If a small volume of air is brought up (or down) over a distance  $l_h$  by such turbulent forces, it will descend (ascend) and arrive at its initial level with a terminal velocity

$$v_h = \text{const.} \cdot l_h \left[ \frac{g}{\Theta} \frac{d\Theta}{dh} \right]^{\frac{1}{2}}$$

where  $\Theta$  is the potential temperature of the gas.

For turbulent motions of this type to occur

$$v_t^2 \geq v_h^2$$

A generalized Richardson number  $Ri^*$  may be postulated, such that

$$\begin{aligned} Ri^* &= \text{const.} \frac{v_h^2}{v_t^2} \\ &= \left[ \frac{l_h}{l_m} \right]^2 \frac{g}{\Theta} \frac{d\Theta/dh}{(dv/dh)^2} = \left[ \frac{l_h}{l_m} \right]^2 Ri \end{aligned}$$

2.12.1

where  $Ri$  is the normal definition of the Richardson number.

Turbulence will occur when  $Ri^*$  is smaller than some critical value.

Considerable simplification of the relation 2.12.1 results from the substitution

$$\frac{1}{T} \left[ \frac{dT}{dh} + \Gamma \right] = \frac{1}{\Theta} \frac{d\Theta}{dh}$$

where  $\Gamma = \frac{\gamma - 1}{\gamma} \frac{g\mu}{R}$ ,  $\gamma = \frac{C_p}{C_v}$   
and  $\mu$  is the molecular weight.

For a dry atmosphere at sea level  $\Gamma \approx 9.9 \times 10^{-5} \text{K/cm}$ , and is somewhat less at 100KM. The precise value is not critical in this discussion.

In practice,

$$Ri^* = \left[ \frac{l_h}{l_m} \right]^2 \frac{g}{T} \left[ \frac{dT}{dh} + \Gamma \right] \left[ \frac{d\bar{v}}{dh} \right]^2$$

Theoretically, one expects turbulence to occur for  $Ri^* < 1$ . In the copious literature on this subject, values from .04 up to 1.0 have been quoted; the lower values usually assume that the momentum mixing length  $l_m$  is equal to  $l_h$ , which is the characteristic size of the eddies arising from the Richardson type instability. In the theory of homogeneous turbulence, the analogue to the momentum mixing length is the characteristic dissipation length  $\lambda$ ; thus

$$Ri^* = \left[ \frac{l_h}{\lambda} \right]^2 \frac{g}{T} \left[ \frac{dT}{dh} + \Gamma \right] \left[ \frac{d\bar{v}}{dh} \right]^2$$

2.12.2

### 2.13 Turbulence in Shear Flow.

As has been mentioned previously, the meteor region of the atmosphere is stabilized by a positive temperature gradient with height. Such stabilization can produce a stratification of motion, and this in turn can give rise to shear flow. Some knowledge of the modifications of the theories of homogeneous turbulence in the presence of shear could be useful.

Tchen (1954) has investigated the turbulent spectrum associated with shear flow, and his findings may be summarized

as follows:-

For small shear fields, the one dimensional energy spectrum in wave number space is of the form

$$E(\kappa) \sim \kappa^{-\frac{1}{3}} \quad 2.13.1$$

and for large shear

$$E(\kappa) \sim \kappa^{-1} \quad 2.13.2$$

The spacial spectrum  $E(\xi)$  thus follows

$$E(\xi) \sim \xi^{\frac{4}{3}} \quad 2.13.3$$

for small shear,

and

$$E(\xi) \sim \ln \frac{1}{\xi} \quad 2.13.4$$

for large shear.

If no preferential direction is prescribed for  $\xi$ , then for small shears the spectrum is usually given by the Kolmogoroff isotropic relation

$$E(\kappa) \sim \kappa^{-\frac{5}{3}} \quad (2.6.5)$$

i.e.

$$E(\xi) \sim \xi^{\frac{2}{3}}$$

2.13.5

If  $\xi$  is the direction specifying the shear, ( a vertical shear is defined as a variation of horizontal wind velocity with height ), then, for small shear, 2.13.1 and 2.13.3 apply. For large shears, the relations 2.13.2 and 2.13.4 are appropriate for all  $\xi$ .

## CHAPTER III.

### Measurements of

### Atmospheric Turbulence at Meteor Heights

#### 3.1 General

The first attempt to apply the Kolmogoroff Theory to turbulent motions in the meteor region was made by Booker in 1956. The results as presented (Booker and Cohen, 1956) were not in accordance with the known behaviour of meteor trails, and the paper contributed little if anything to the understanding of this region. However, it did stimulate the investigation of turbulence at these heights, and resulted in papers by Manning (1959) and Greenhow (1959) which provided an adequate qualitative treatment of the motions of meteor trails subject to random winds. A paper by Greenhow and Neufeld (1959) on work carried out at Jodrell Bank detailed the results of a spaced station radar survey of wind velocity differentials measured from the drift of meteor trail ionization. (For details of the spaced station technique, see Chapter VII). However, several anomalies exist in this paper, and the value derived for  $\epsilon$ , the turbulent dissipation energy, is almost an order greater than that determined from rocket



and photographic meteor data (see Chapter XII). Their work is, however, worthy of detailed consideration, especially that portion dealing with the larger scale motions.

### 3.2 Discussion of the work of Greenhow and Neufeld.

The analysis of velocity correlations in this paper is not particularly rigorous. Because of the limitations imposed by station separations the maximum height difference observed was 4.5KM, at which value the correlation function relating the transverse turbulent velocity to height separation fell to 0.22 (see Fig. 3). The curve is then extrapolated to zero at 6.5KM, this extrapolation being justified by Greenhow and Neufeld from results published by Liller and Whipple (1954), Fig. 4, which show loss of correlation between the horizontal velocity of points separated by some 6 - 7KM in height. Figure 5 is an actual photograph of a trail undergoing distortion. However, in general the zero of the transverse correlation does not determine the scale for which velocities are no longer correlated. A measure of the separations over which velocities are appreciably correlated is given by equation 2.6.3, viz.

$$L_p = 2 \int_0^{\infty} g(\Delta h) d\Delta h$$

(using Greenhow's notation)

From the theoretical shape of the transverse correlation

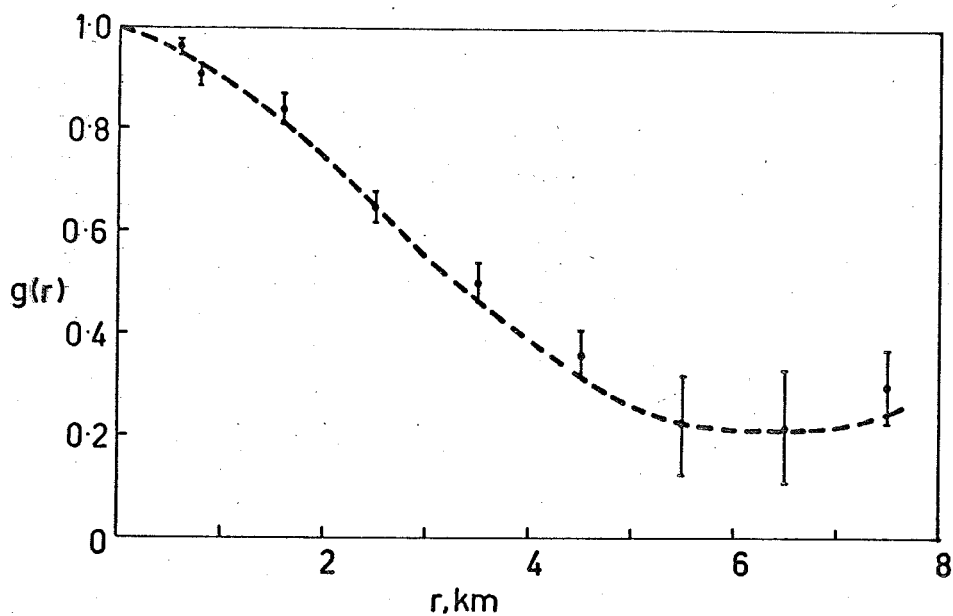


FIG.3a-Variation of velocity correlation  $g$  with spacial separation of the reflecting points  $r$ .

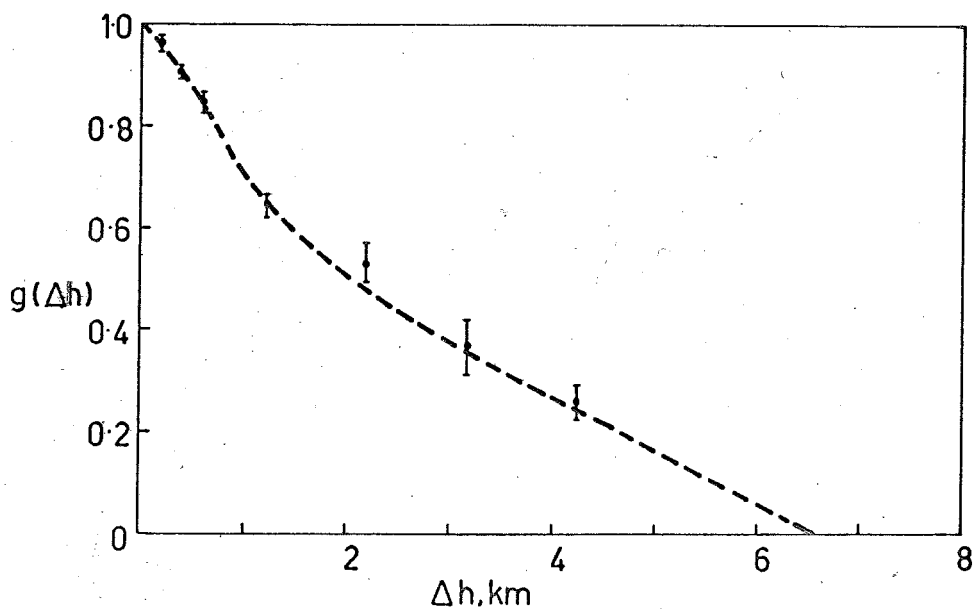


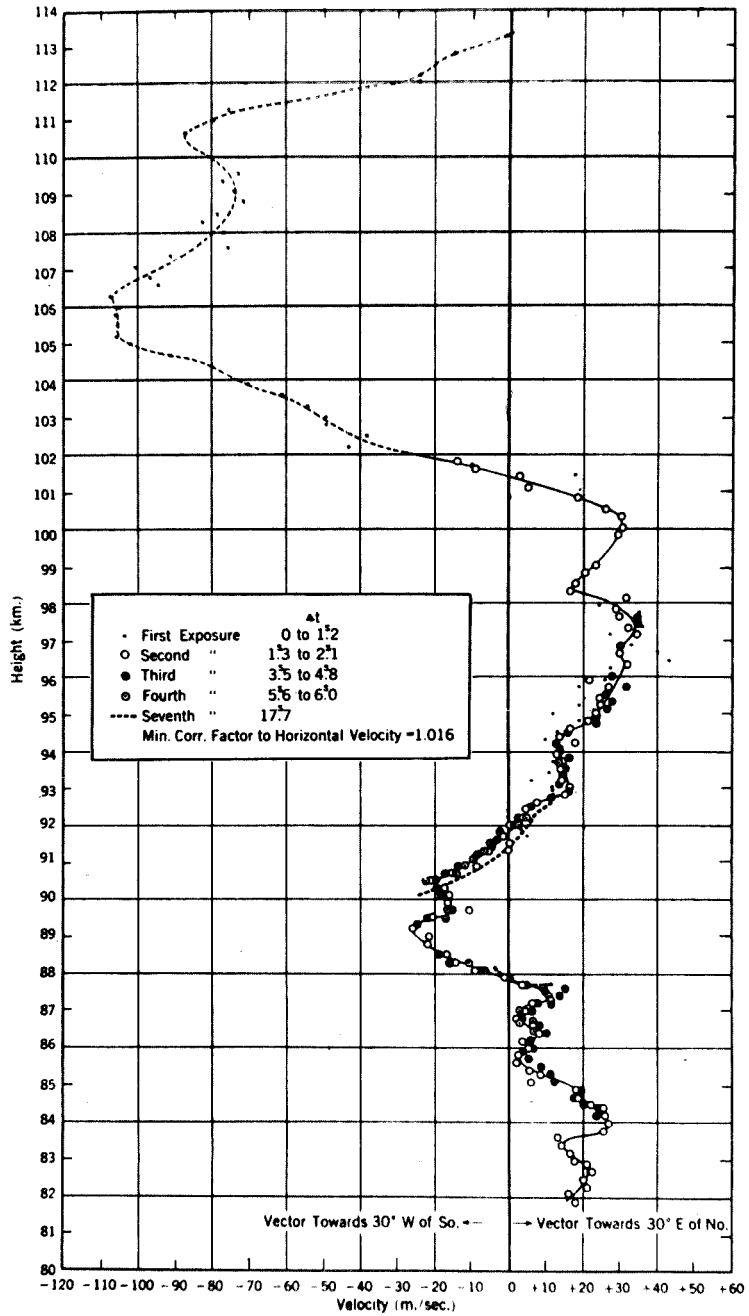
FIG.3b-Variation of  $g$  with height separation  $\Delta h$  of the reflecting points.

curve (see Fig. 1), the area under the curve up to the zero value is an upper limit of the scale of appreciable correlation. By graphing the transverse correlation function of Fig. 3 on squared paper, and counting squares to determine the area under the curve

$$2 \int_0^{6.5 \times 10^5} g(\Delta h) d\Delta h = 5.0 \times 10^5 = 5 \text{ km.}$$

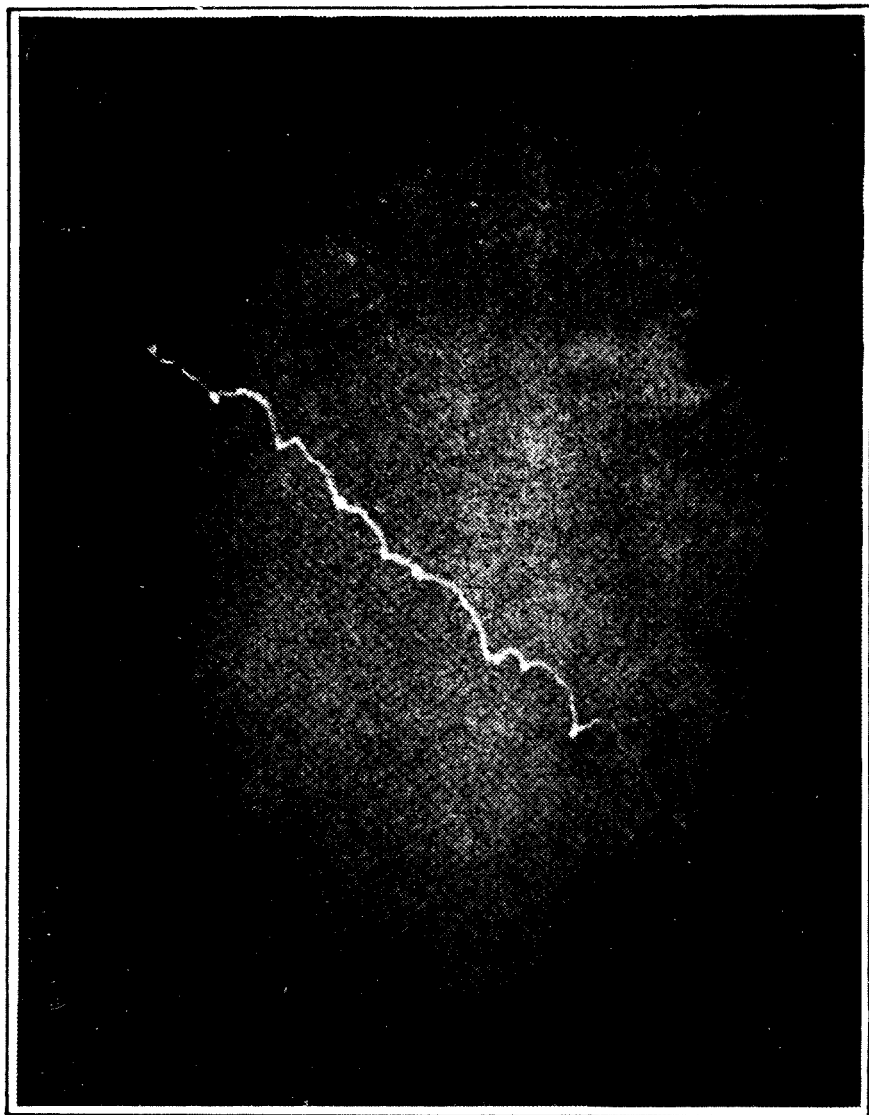
This upper limit of appreciable correlation is in keeping with the zero correlation height difference of 6 - 7 KM found by Liller and Whipple. Thus although the scale determined by the zero of the transverse correlation function is not justifiable on theoretical grounds as the scale of zero velocity correlation, it does appear to be a useful indication in this particular application.

The turbulent velocity correlations as derived by Greenhow and Neufeld are not characteristic of the complete flow field, but of the 6KM scale only. To allow for the prevailing and periodic motions of the atmosphere, the mean horizontal wind during any hour was obtained by averaging approximately 100 individual velocities obtained each hour. The turbulent component for echoes in that hour was obtained by subtracting the radial component of this steady wind from each individual echo. One of the restrictions placed on the



A typical example of meteor train drifts determined photographically at the Harvard Observatory by Liller and Whipple.

FIG 4



Photograph of a meteor train taken about 60 seconds after passage of the meteor. Length of train approximately  $19^\circ$ . Photo by F. L. Trube, May 3, 1939.

FIG 5

spacial correlation function  $g(\xi)$  is that the turbulent velocity components must be averaged over a time large in comparison with the characteristic time of the energy bearing eddies. Greenhow and Neufeld determined this time constant to be 100 minutes (see Section 4 this Chapter). Their averaging interval for mean flow determination is obviously too small, and the correlation function thus determined is not characteristic of the complete turbulent flow. This accounts for the marked effects of the height correlation on the spacial function.

If the correlations of Fig. 3 are graphed according to the relations established in Chapter 2 Section 11

viz.

$$1 - R(\xi) \text{ against } \xi$$

Fig. 6 results.

Because of the anisotropic nature of the large scale eddies, the spacial correlation, parameter  $V_T = \xi$  of which contains components of the horizontal, exhibits the anomalous slope of 1; linear extrapolation of this portion of the curve to "zero correlation" ( $1 - R(\xi) = 1$ ) yields a scale of 7KM.

The correlation against  $\Delta h$ , however, exhibits the slope characteristic of the inertial subrange of the Kolmogoroff turbulence spectrum. Extrapolation of this line to "zero correlation" yields a vertical scale of 6KM.

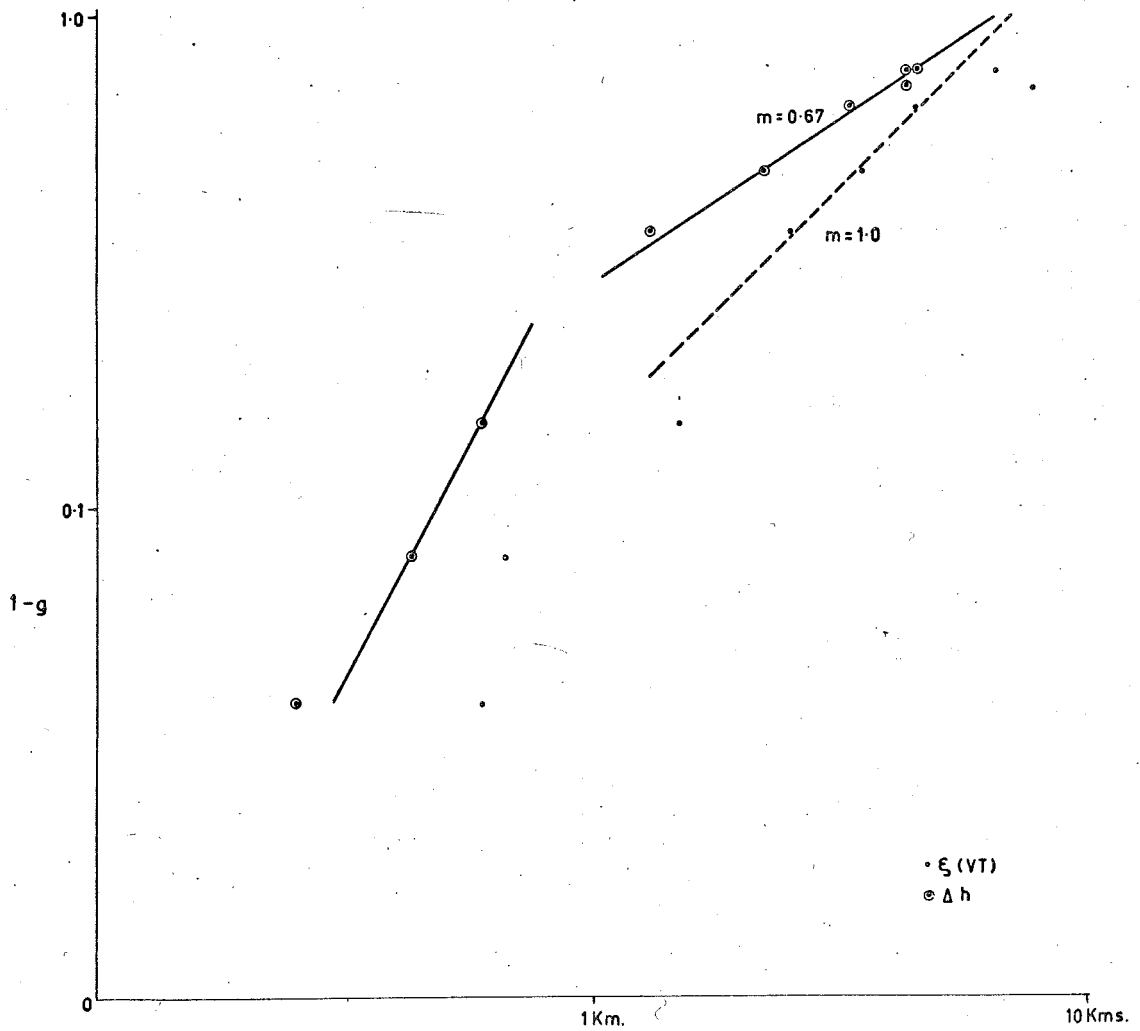


FIG.6 - SPACIAL AND HEIGHT CORRELATIONS  
(AFTER GREENHOW)

At separations less than 1KM, the slopes of the correlations change markedly. This breakdown could be explained as due to the onset of viscous dissipation at scales less than 1KM. Since from hydraulics and wind tunnel experiments the spectrum behaviour of the inertial subrange can be expected to hold down to scales five times the characteristic eddy size of the Kolmogoroff microscale (equation 2.6.6), this would indicate a minimum eddy size

$$\eta = 200 \text{ metres.}$$

The error associated with Greenhow and Neufelds' determination of  $g(0)$  is of the order of 0.01. This limits the accuracy of direct determination of the smaller scales to some 140 metres.

### 3.3 Richardson Type Instabilities.

Another explanation for the inertial subrange breakdown at 1KM is the extraction of kinetic energy from the turbulence and its conversion to potential energy by bouyancy forces as suggested by Bolgiano (1959). In this, the generalized Richardson number of Chapter 2 Section 12 is the relevant parameter.

Since the height correlation function appears to follow the Kolmogoroff spectrum, the fitting of a parabola of the form

$$g(\xi) = 1 - \xi^2/\lambda^2$$

(equation 2.5.3)



to the correlation curve in the neighbourhood of  $\xi = 0$  should yield a dissipation length parameter  $\lambda$  characteristic of the vertical scale of turbulent motion. The value of the dissipation length thus determined is

$$\lambda = 1.4 \text{ KM}$$

Substitution of this value in equation 2.12.2, viz.

$$Ri^* = \left[ \frac{l_h}{\lambda} \right] \frac{g}{T} \left[ \frac{dT}{dh} + \Gamma \right] \left[ \frac{d\bar{v}}{dh} \right]^{-2}$$

using the ARDC (1959) Model Atmosphere in determining  $T$ ,  $\frac{dT}{dh}$  and  $\Gamma$  and taking Greenhow's figure of 10m/sec/km for  $\frac{d\bar{v}}{dh}$  yields

$$Ri^* = \left[ \frac{l_h}{\lambda} \right]^2 \times 4$$

For  $Ri^* < 1$

$$l_h < 700 \text{ metres}$$

i.e. the scale of eddies arising from Richardson type instabilities is less than 700 M. Further comments on this value in relation to the small scale turbulence appear in Chapter IX.

### 3.4 The Large Scale Irregularities.

In determining the size of the large scale eddies, Greenhow and Neufeld (1959 a) used the single station time correlation function:

$$g(\tau) = \frac{\overline{v(t)v(t+\tau)}}{v^2(t)}$$

Their results as published have been regraphed according to the relation

$$1 - R(\tau) \sim \tau^m$$

(equation 2.11.5)

The best least squares fit line to these points (Fig.7) has slope 0.66, indicating a turbulent velocity less than that of the mean flow (see Chapter 2 Section 11). This is not readily explained since, according to Greenhow and Neufeld, "there was no correlation between the RMS turbulent velocity for any hourly interval and the mean wind speed or mean height gradient in that interval, turbulent velocities of up to 40m/sec being measured even when the mean wind speed was zero." Further comments on this in the light of measurements made at Adelaide appear in Chapter IX.

From the RMS turbulent velocity, measured as 25m/sec, and the time constant of  $6 \times 10^3$  secs (being the time of "zero correlation"), the scale of these eddies is given as some 150KM.

From several time correlations plotted from data obtained over several 24 and 48 hour periods, a minimum scale of 60KM

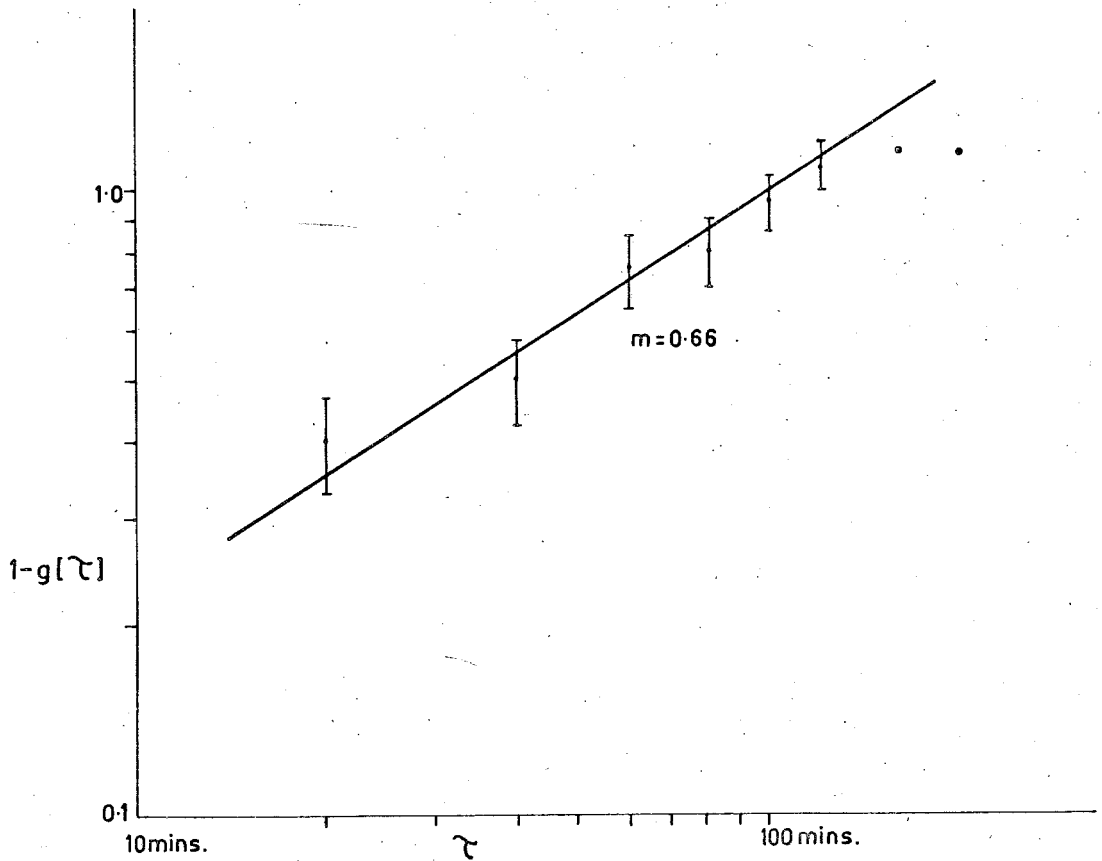


FIG.7 - TIME CORRELATION (AFTER GREENHOW)

and a maximum of 250KM was observed. (Greenhow and Neufeld, 1960). These larger scales are given further consideration in Chapter IX.

### 3.5 Eddy Diffusion and Photographic Meteors.

The effects of eddy diffusion on meteor trails at heights above 90KM have been investigated by Greenhow by using visual and photographic observations summarized by Millman. The trail radii, defined as the standard deviation  $r$  of an assumed Gaussian distribution of intensity across any diameter, were estimated by Greenhow and compared with the theoretical variation of  $r$  for various values of ambipolar diffusion and eddy diffusion. The behaviour of such diffusion of trails has been adequately summarized by Booker (1956) as follows. Immediately after trail formation, dispersion proceeds under the influence of ambipolar diffusion, given by

$$r^2 = 4Dt$$

where  $D$  is the ambipolar diffusion coefficient.

After a time  $t_1$ , the time constant of the characteristic eddy of the Kolmogoroff turbulence spectrum, the trail radius increases as

$$r^2 = \frac{4}{3}ct^3$$

up to a time  $t_2$ , the time constant of the large scale turbulence. After  $t_2$  the radius once more increases in the

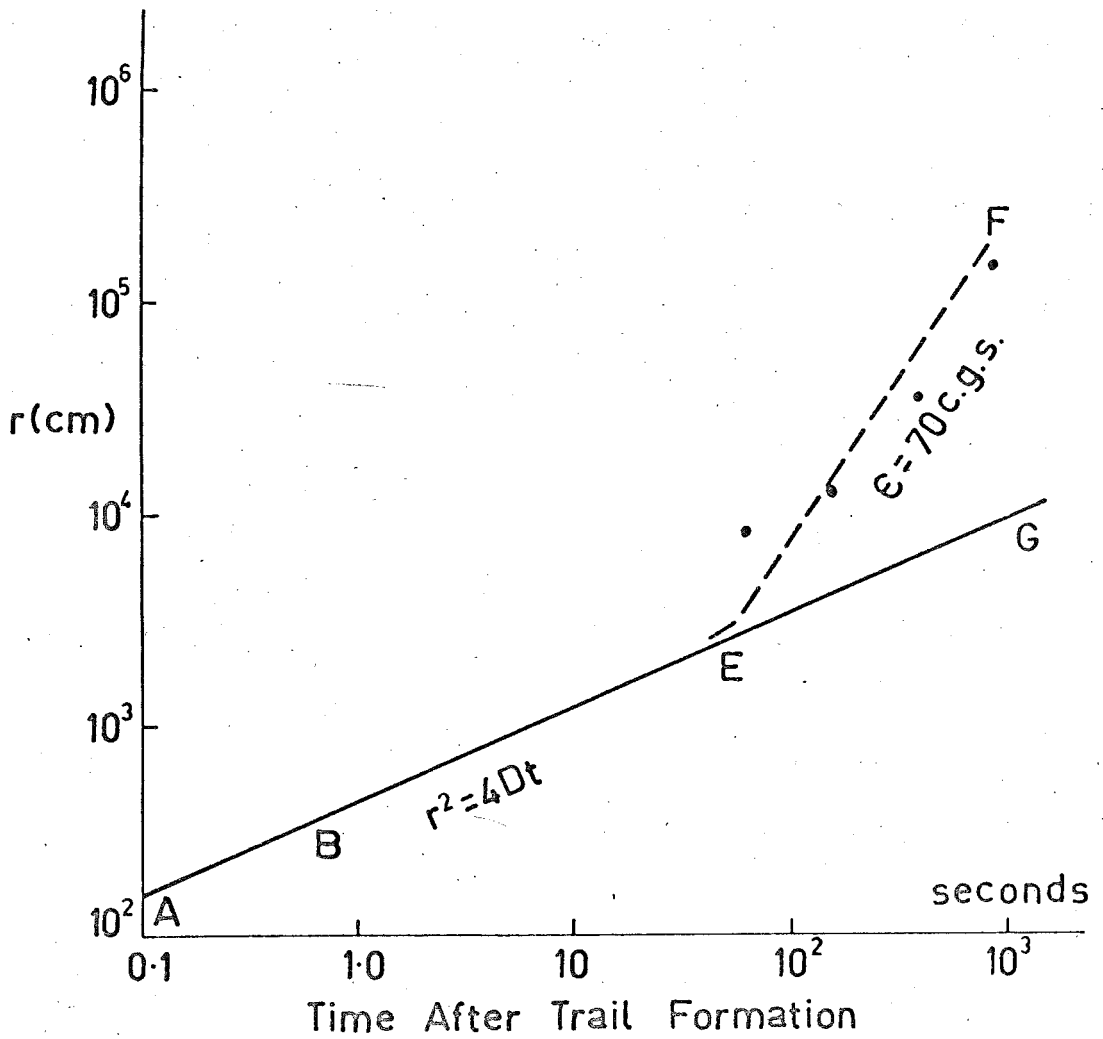


FIG.8 - TRAIL RADIUS  $r$  AS A FUNCTION OF TIME.

manner

$$r^2 = 4D_e t$$

where  $D_e$  is the eddy diffusion coefficient.

$D_e = VL$ , where  $V$  and  $L$  are the velocity and scale of the large scale turbulence.

Millman's results graphed by Greenhow according to these relations are shown in Fig. 8. AG is the expected growth of trail radius under the influence of ambipolar diffusion alone ( $D = 4 \times 10^4$  sq cm/sec at 90 KM). The measured values of  $r$  shown on the figure lie about the line EF, for which  $r^2 = \frac{4}{3}\epsilon t^3$ , and correspond to a turbulence power of only 70 ergs/gm/sec. They give a value of 30 secs for the time constant of the small scale turbulence, far in excess of the value of 0.4 secs determined by Booker and more in keeping with the observed behaviour of radio echoes from meteor trails, which often endure without any observable distortion of reflection for periods of several seconds.

The value of  $\epsilon$  thus determined is an order smaller than that determined by Greenhow from spaced station observations of meteor trails.

### 3.6 Rocket Results.

From photographic observations of the diffusion of sodium trails ejected from a Veronique rocket, Blamont and De

Jager (1961) calculated by similar means the turbulent dissipation  $\epsilon$  to be approximately 70 ergs/gm/sec below 102 KM. Above this height, the observed diffusion was found to be completely explained by molecular diffusion. Height stratification of atmospheric motion with dimension 6 KM was also in evidence, the observed winds being predominantly horizontal, with a vertical maximum of  $\pm 10$  m/sec.

Groves (1959) and Rofe (1961) have carried out similar experiments at Woomera, with both sodium trails and grenades. The work of Rofe is of particular interest in that small scale vertical motions are clearly indicated (Fig. 9a). The presence of such vertical motion is of prime importance in the explanation of anomalies arising from the application of the Kolmogoroff theory to small scales at meteor heights (vide Section 3 this Chapter, and Section 5 of Chapter IX).

### 3.7 An Examination of Velocity Differentials with Height.

Zimmerman (1962) has applied Tchen's theory of shear turbulence to the results of the velocity/height differential analysis performed by Blamont and De Jager (1961) from their sodium trail experiments. Excellent agreement with the Tchen modifications of the Kolmogoroff theory was obtained. For preferential specification of  $\xi$  as  $z$ , the sodium trail data gives

$$E(z) \sim z^{\frac{4}{3}}$$

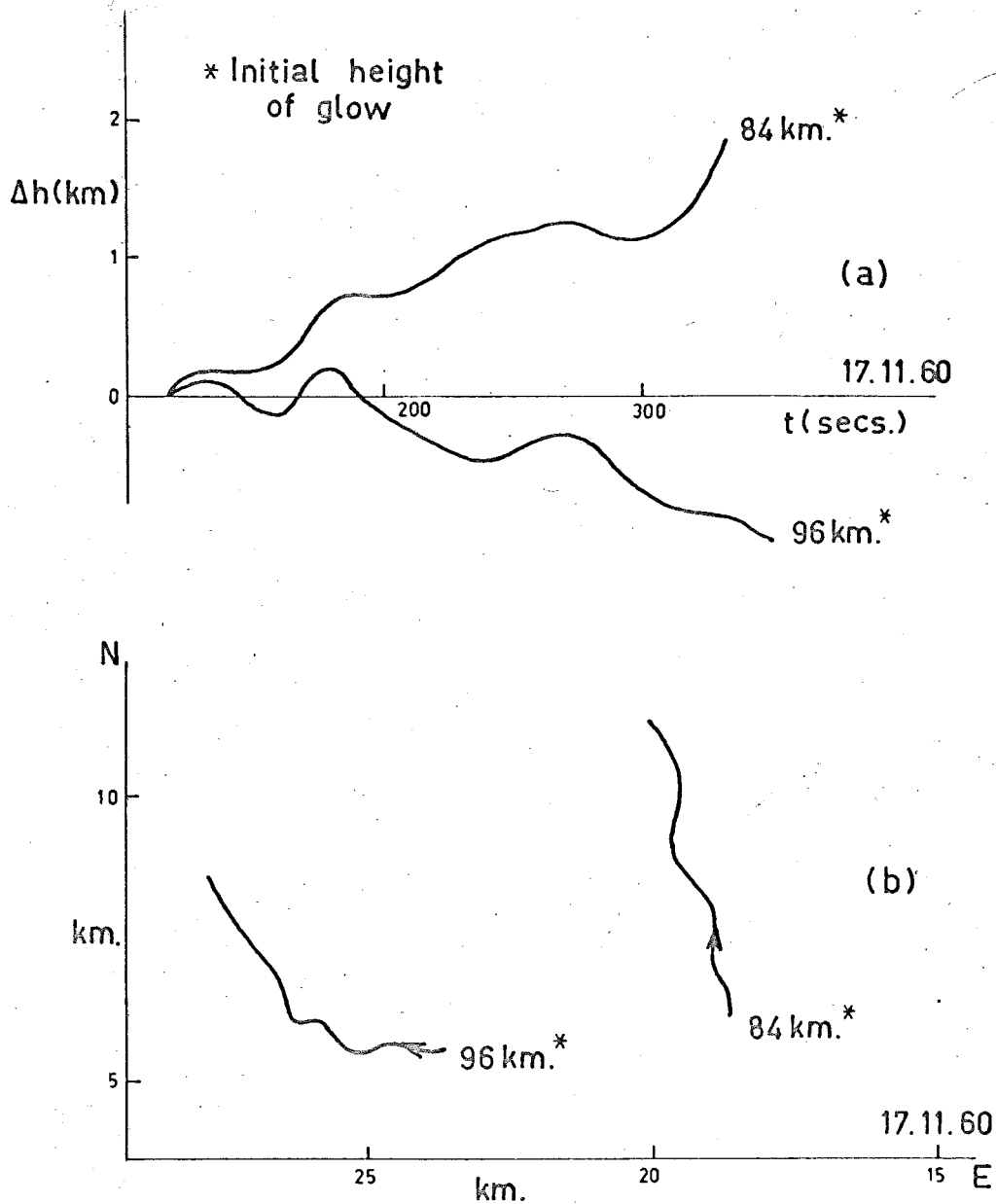


FIG.9 - DISTANCE TIME PLOTS OF GRENADE GLOWS. (ROFE, 1961)



up to 6KM height separation, and

$$E(z) \sim \ln \frac{1}{z}$$

for separations greater than 6KM. Zimmerman reforms the height differentials to make the separation  $\xi$  isotropic, and finds a Kolmogoroff spectrum

$$E(\xi) \sim \xi^{\frac{2}{3}}$$

pertains up to 6KM separation, with again

$$E(\xi) \sim \ln \frac{1}{\xi}$$

for separations greater than 6KM.

Thus the Blamont / De Jager measurements at 80 - 100 KM are applicable to a region of small shear at separations up to 6KM, with large shears predominating at greater separations.

## CHAPTER IV.

### Radio Reflections from Meteor Trails

Before the applications and limitations of radio methods of meteor tracking as used to determine atmospheric motions can be fully appreciated, the mechanism of reflection must be understood.

#### 4.1 Historical

When the serious study of the ionosphere began in about 1925, it was discovered that, in addition to the ionization produced by the ultraviolet radiation of the sun, subsidiary sources must exist, particularly in the E region where abnormal increases in ionization occurred during the night.

Skellett (1931) suggested that meteors might be responsible, at least in part, for the E layer anomaly, and, together with Schafer and Goodall (1932) carried out pulse type soundings of the ionosphere during the Leonid meteor shower of 1931. Their results showed that E layer ionization was increased during meteor showers, and that transient increases of up to a few seconds duration sometimes accompanied individual meteors.

In 1941, Chamanlal and Venkataraman in India reported

weak whistles on an unmodulated 7 MCS carrier received at a site located some 10 miles from the transmitter. These whistles lasted approximately a second, usually descended in pitch, and coincided with visual meteors.

With the advances in radio techniques made during the war years, it was possible, in 1945, to begin a systematic study of radio reflections from meteors. The pioneering work in this field was carried out by Hey and Stewart (1947) using a military disposals radar equipment operating on a frequency of 70 MCS. By 1949, radio observations of Meteors were being carried on at Manchester (Lovell, Clegg, Aspinall, Almond, Davies, Ellyet), Stanford (Manning, Villard and Petersen) and Ottawa (McKinley).

More recent developments in this field have already been discussed in Chapter III, and a detailed account of the Adelaide work is given in Chapter VI.

The theoretical discussion which follows is due to Herloffsen, Lovell and Clegg, and Kaiser.

#### 4.2 Theory of Trail Formation.

Because of its high velocity, a meteor in orbit has an energy of the order of several hundred electron volts. As the meteor enters the upper atmosphere, collision with air molecules results in the transformation of some of the kinetic energy of the meteor into heat. The ionized trail is produced

in the 70 to 120 km height range by evaporation of the meteor, molecules leaving the meteor at thermal velocities relative to the meteor of the order of a few electron volts. ( Herloffsen 1948 )

The rate of evaporation is given by

$$n = (\mu H)^{-1} m v \cos \chi \frac{p}{p_{\max}} \left[ 1 - \frac{1}{3} \left\{ \frac{p}{p_{\max}} \right\} \right]^2 \quad 4.2.1$$

where

$n$  = number of meteor atoms evaporated per second

$m$  = initial mass of the meteor

$\mu$  = mass of an individual meteor atom

$v$  = meteor velocity

$H$  = atmospheric scale height

$p_{\max}$  = atmospheric pressure at point of max. rate of evaporation

$p$  = atmospheric pressure at point where  $n$  occurs

$\chi$  = zenith angle of the meteor radiant

( Kaiser 1954 )

The electron line density  $\alpha$  produced by the meteor is

$$\alpha = \frac{\beta n}{v} \quad 4.2.2$$

where  $\beta$  is the probability of a single evaporated meteor atom giving rise to a free electron.

Substituting for n

$$\alpha = \frac{\beta}{\mu H} m \cos \chi \frac{p}{p_{\max}} \left[ 1 - \frac{1}{3} \left\{ \frac{p}{p_{\max}} \right\} \right]^2 \quad 4.2.3$$

Visual meteors have  $\alpha > 10^{10}$  electrons per centimeter.

At formation, the diameter of the trail is of the order of 10 - 100 cm, being a function of the mean free path.

#### 4.3 Decay of Ionized Trails.

After formation, the trail decays under the influence of diffusion, recombination and attachment. Diffusion is the most important; recombination and attachment only assume significance for long duration echoes.

Radial ambipolar diffusion leads to a Gaussian electron density distribution at time t of the form,

$$n_e = \frac{\alpha}{4\pi Dt} e^{-\frac{r^2}{4Dt}} \quad 4.3.1$$

where

$\alpha$  = initial line density of ionization

D = ambipolar diffusion coefficient (  $D = 2 D_{\text{ionic}}$  )

r = radius of the trail.

#### 4.4 Scattering of Radio Waves by Meteor Trails. The Radar case.

The classical theory of reflection from meteor trails is that due to Lovell and Clegg (1948). This was subsequently

amplified by Kaiser (1955).

The non-relativistic scattering cross section for a single electron is given by

$$\sigma = \frac{8\pi}{3} \left[ \frac{e^2}{mc^2} \right]^2$$

Each electron behaves as a hertzian dipole, with power gain 1.5 over an isotropic radiator. Thus the equivalent echoing area per electron

$$= 4\pi \left[ \frac{e^2}{mc^2} \right]^2$$

For an element of trail length  $d\xi$ , electron line density  $\alpha$ , the echoing area becomes

$$dA = 4\pi \left[ \frac{e^2 \alpha d\xi}{mc^2} \right]^2 = 4\pi \left[ \frac{e^2}{mc^2} \right]^2 \alpha^2 d\xi^2$$

If  $P_t$  is the power output of the transmitter in watts,  $G$  the aerial power gain over an isotropic radiator in the direction of the trail,  $R$  the range of the echo and  $\lambda$  the wavelength, then the received echo power from element  $d\xi$  is

$$\begin{aligned} dP_r &= \frac{P_t G^2 \lambda^2}{(4\pi)^3 R^4} dA \text{ watts} \\ &= \frac{P_t G^2 \lambda^2}{64\pi^3 R^4} 4\pi \left[ \frac{e^2}{mc^2} \right]^2 \alpha^2 d\xi^2 \text{ watts} \end{aligned}$$

The total reflected power from the trail is given by integration of the contributions from all  $d\xi$  with regard to

the phase of each, and is given in terms of the Fresnel integrals  $C$  and  $S$ .

Total power of the echo received from the trail is

$$P_r = \frac{P_t G^2 \lambda^2}{64\pi^3 R^4} 4\pi \left[ \frac{e^2}{mc^2} \right]^2 \frac{R\lambda}{4} \left[ C_t^2(\xi) + S_t^2(\xi) \right] \text{ watts.}$$

where  $C_t(\xi)$  and  $S_t(\xi)$  are the values of the Fresnel integral over the complete trail, whence

$$P_r = \frac{P_t G^2 \lambda^3}{32\pi^2 R^3} \left[ \frac{e^2}{mc^2} \right]^2 \alpha^2 \text{ watts}$$

since

$$C_t^2(\xi) + S_t^2(\xi) = 2$$

or

$$P_r = 3.3 \times 10^{-28} \frac{P_t G^2 \lambda^3}{R^3} \alpha^2 \text{ watts.}$$

4.4.1

If separate receiving and transmitting aerials are used (as, of necessity, in the CW method)

$$P_r = 3.3 \times 10^{-28} \frac{P_t G_t G_r \lambda^3}{R^3} \alpha^2 \text{ watts.}$$

4.4.2

An alternative form of this equation, which is more readily applicable to the consideration of the partly formed trail, was obtained by Kaiser by postulating a reflection

coefficient  $g = g(\xi)e^{i\phi}$ , where  $\xi$  is the trail length parameter measuring the distance of the meteor from the  $t_0$  or specularly reflecting point, and  $\phi$  is the phase of the reflected wave, such that equation 4.4.2 above becomes

$$P_r = \frac{P_t G_t G_r \lambda^3}{32\pi^4 R_0^3} gg^*$$

Before proceeding further, it is necessary to differentiate between the two types of trail reflection, viz. that produced by underdense and overdense trails. These types are in practice recognized by their echo durations.

#### 4.5 Echo Duration.

Radial diffusion of the trail (Section 3) produces a trail radius  $r$  tending to and then exceeding the transmitted wavelength  $\lambda$ . The received power from such a trail is progressively reduced by interference, and the echo decays exponentially with a decay time (duration)

$$\tau = \frac{\lambda^2}{16\pi^2 D}$$

4.5.1

where  $D$  is the ambipolar diffusion coefficient.

This is the typical decay pattern of underdense trails. If, however, the initial trail electron line density exceeds a certain critical value, reflection will occur from the surface of an expanding cylindrical column, and the echo will



persist until the electron line density falls below this critical value, when the echo will decay exponentially as before. Kaiser and Closs (1952) determined this critical electron line density to be approximately  $2.4 \times 10^{12}$  electrons/cm.

If  $\alpha \ll 2 \times 10^{12}$  electrons per cm, then

$$g = \pi \alpha \left[ \frac{e^2}{mc^2} \right] e^{-4k^2 Dt} \quad 4.5.2$$

$$k = \frac{2\pi}{\lambda}$$

for a fully formed trail.

#### 4.6 The partly formed trail.

If, however, the trail is only partly formed, the reflection coefficient is given by

$$g(\xi) = C \int_{-\infty}^{\xi} F(x) \exp \left[ -ik \left\{ \frac{x^2}{R_0} + 2(\xi - x) \frac{u}{v} \right\} \right] dx \quad 4.6.1$$

where  $\xi = vt$  ( $t = 0$  when meteor is at the specular reflection point  $t_0$ )

$F(x)$  = reflection coefficient of the trail when meteor is at  $x$  relative to the  $t_0$  point

$u(x)$  is the line of sight wind component

$R_0$  is the range of the  $t_0$  point

$v$  is the meteor velocity

and  $k = \frac{2\pi}{\lambda}$  as before

(Kaiser, 1955)

The constant  $C$  is chosen so that when  $F(x) = \text{constant}$ ,  $u = 0$  and  $\xi \rightarrow \infty$ ,  $g = F = \text{value for infinite column of ionization}$  (equation 4.5.2)

i.e.

$$C = \left[ \frac{2}{R_0 \lambda} \right]^{\frac{1}{2}} e^{i\pi/4}$$

#### 4.7 The effect of wind shear on reflection.

In the special case where  $F = \text{constant}$ ,  $u = u_0 + u'x$  (mean wind plus constant wind shear of gradient  $u'$ ), and putting  $\delta = \frac{2u'R_0}{v}$ , equation 4.6.1 reduces to

$$g(\xi) = 2(1-\delta)^{\frac{1}{2}} \exp \left[ i \left\{ \frac{\pi}{4} - \theta \right\} \right] I(\nabla) \quad 4.7.1$$

where

$$I(\nabla) = \int_{-\infty}^{\nabla} e^{-i\frac{\pi}{2}\nabla^2} d\nabla$$

i.e. Fresnel's Integral

This has solution

$$\nabla = \frac{2-\delta}{[R_0 \lambda (1-\delta)]^{\frac{1}{2}}} v \left[ t - \frac{2u_0 R_0}{v^2 (2-\delta)} \right] \quad 4.7.2$$

$$\theta = k \left\{ u_0 t \left( \frac{2-\delta}{1-\delta} \right) - \frac{\delta^2 v^2 t^2}{4(1-\delta)R_0} - \frac{R_0}{(1-\delta)} \frac{u_0^2}{v^2} \right\} \quad 4.7.3$$

The phase shift

$$\dot{\theta} = k \left[ u_0 \left\{ \frac{2-\delta}{1-\delta} \right\} - \frac{\delta^2 v^2 t}{2(1-\delta)R_0} \right] \quad 4.7.4$$

corresponds to an apparent time dependent drift velocity.

When compared with the phase of the transmitted wave, the reflected wave will produce a doppler beat of frequency

$$f = \frac{\dot{\theta}}{2\pi}$$

This beat results from the apparent velocity  $u_a$  of the trail with respect to the T/R system, where

$$u_a = \frac{f\lambda}{2}$$

i.e.

$$u_a = \frac{\theta\lambda}{4\pi} = \frac{u_0(2-\delta)}{2(1-\delta)} \frac{\delta^2 v^2 t}{4(1-\delta)R_0}$$

4.7.5

If  $u' = 0$  (i.e. uniform wind, no shear) then

$$\delta = 0 \text{ and } u_a = u_0$$

i.e. the apparent line of sight drift velocity of the trail is equal to the line of sight component of the wind. It should be mentioned here that the drift of the trail is that of the neutral air. At this height the influence of the background ionization and magnetic field on the ionized trail is negligible.

If  $\delta \ll 1$  i.e.  $2u'R_0 \ll v$  which is usually the case, then equation 4.7.5 reduces to

$$u_a = u_0 - u'^2 R_0 t$$

4.7.6

By measuring velocities  $u_1$  and  $u_2$  at times  $t_1$  and  $t_2$  respectively, we can find  $u'$  since

$$u_1 - u_2 = -u'^2 R_0 (t_1 - t_2)$$

i. e.

$$u'^2 = - \frac{u_1 - u_2}{R_0 (t_1 - t_2)}$$

This gives values of  $u'$  in cm/sec/cm. The conventional units used are m/sec/km, and in these units

$$u'^2 = - \frac{u_1 - u_2}{R_0 (t_1 - t_2)} \times 10^3 \text{ m/sec/km.} \quad 4.7.7$$

where  $u$  is in m/sec

and  $R_0$  in km.

The significance of this equation is best demonstrated by considering the physical picture. Neglecting the translation due to the mean wind component, the effect on the trail of a linear gradient  $u'$  m/sec/km (measured perpendicular to and along the trail) is to cause a rotation  $\omega$  such that

$$\omega = u' \times 10^{-3} \text{ radians/sec}$$

To maintain specular reflection, the reflection point will, in time  $t$ , run along the trail from  $R_1$  to  $R_2$ , a distance of  $R_0 \omega t$  (see Fig. 10). If the line of sight drift velocity is  $u_1$  at  $R_1$ , and  $u_2$  at  $R_2$ , then

$$u_1 - u_2 = u' R_0 \omega t = u'^2 R_0 t \times 10^{-3}$$

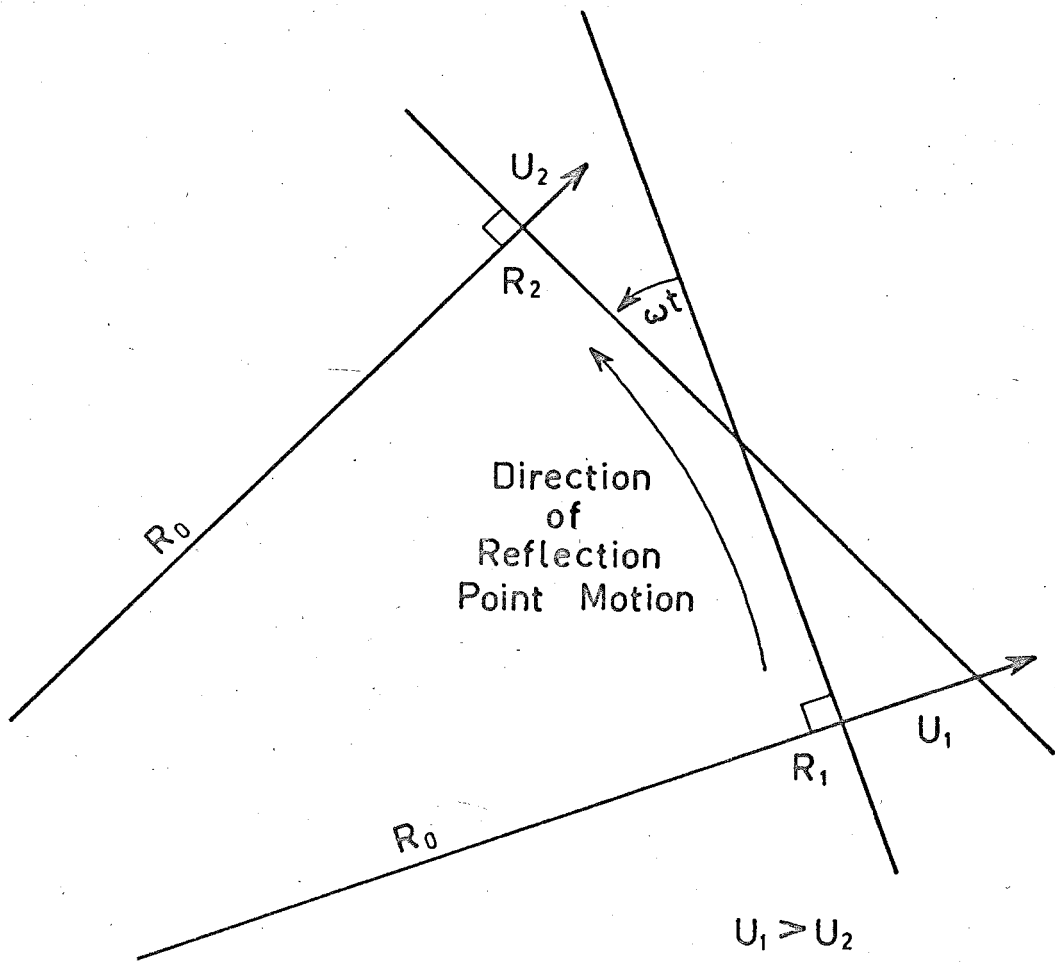


FIG. 10

i. e.

$$u'^2 = \frac{u_1 - u_2}{R_0 t} \times 10^3$$

The rate of motion of the reflection point is given by

$$T = R_0 u' \times 10^{-3} \quad \text{kilometres/sec}$$

4.7.8

Rao (1958) and Rao and Armstrong (1958) considered this motion of the reflection point to be dependent on the rate of decay of ionization at the reflection point (accompanied by a shift of the reflection point towards the position of maximum ionization  $\alpha$  max.) The explanation presented here, which was first proposed by Elford (1954) with different mathematical reasoning, is considered to be more plausible.

#### 4.8 The effect of uniform wind and shear on the measurement of $t_0$ points.

The amplitude function for uniform wind (no gradient) is given by

$$\nabla = K v \left[ t - \frac{u_0 R_0}{v^2} \right]$$

Kaiser. (See equation 4.7.2)

At the  $t_0$  point, if no wind is present,  $\nabla = 0$ . However, in the presence of a wind  $u_0$ , the measured value of  $t_0$  is delayed by

$$\partial t_0 = \frac{u_0 R_0}{v^2}$$

4.8.1

If the one trail is being observed from three spaced receiving sites, the property of specular reflection will assign a different  $t_0$  point to each station (see Chapter 7 Section 1). With a uniform wind, there will be a similar delay for each reflection point, and thus no overall effect on the time differences between  $t_0$  points at the different sites.

If, however, a linear wind gradient is also present, then the line of sight component of the velocity will be different at each  $t_0$  point, and will thus produce a different shift in each point.

This delay at each station is given by

$$\partial t_{01} = \frac{u_1 R_1}{v^2}$$

$$\partial t_{02} = \frac{u_2 R_2}{v^2}$$

$$\partial t_{03} = \frac{u_3 R_3}{v^2}$$

In practice,

$$R_1 \approx R_2 \approx R_3 \approx R_0$$

where  $u_1$ ,  $u_2$  and  $u_3$  are the line of sight drift velocities measured at  $t_{01}$ ,  $t_{02}$  and  $t_{03}$  respectively.

Thus the true time taken for passage of the meteor from  $t_{01}$  to  $t_{02}$  is

$$\Delta t_0 = \Delta t - \frac{u_1 R_0}{v^2} + \frac{u_2 R_0}{v^2}$$

where  $\Delta t$  is the measured interval.

i. e.

$$\Delta t_0 = \Delta t - \frac{R_0}{v^2} (u_1 - u_2).$$

But

$$u_1 - u_2 = u' v \Delta t$$

where  $u'$  is the wind gradient along the trail.

$$\therefore \Delta t_0 = \Delta t \left\{ 1 - \frac{u' R_0}{v} \right\} = \Delta t (1 - \delta/2) \quad 4.8.2$$

$$\text{where } \delta = 2u'R_0/v$$

This is the same as the expression derived by Kaiser (1955) for the time displacement for spaced stations along a trail  $x$  drifting under the influence of a line of sight wind component  $u = u_0 + u'x$ .

#### 4.9 The effect of wind gradients on the measurement of meteor velocities.

As shown in 4.8, a uniform wind affects only the time displacement of the echo, and not the velocity measurement. If a uniform gradient  $u'$  is present, however, then from equation 4.7.2 the measured velocity is given by

$$v_a = \frac{1}{2}v \frac{2 - \delta}{(1 - \delta)^{1/2}} \quad 4.9.1$$

$$= v \left\{ 1 - \frac{\delta^2}{8} \right\} \quad \text{for } \delta \ll 1$$



i.e. the effect is small and may usually be neglected.

#### 4.10 Discussion.

The foregoing represents, of course, an idealized set of conditions. Trail ionization is certainly not constant, but increases as the meteor descends through the atmosphere, reaching a maximum only a very short distance before the eventual complete evaporation of the meteor. The meteors with highest velocities will, in general, ionize and disintegrate at higher levels in the atmosphere. Here, the diffusion coefficient is greater, and the trail will therefore decay more rapidly. Because of this limitation in available recording time, there may be some exclusion of faster meteors from our investigation. The simple wind velocity correction applied to  $t_0$  measurements is also subject to the limitation that wind gradients may not be linear shears. However, inclusion of this correction does reduce the scatter in radiant for known shower meteors, and is thus worthwhile.

Because of the possible motion of the reflection point along the trail in the presence of wind shears, a limit is placed on the minimum useful separation of reflection points for calculation of wind gradients from separated stations. This is in addition to the uncertainty of location of the reflection "point" within the first Fresnel zone, which makes the maximum contribution to the reflection. These factors

are discussed in greater detail in later chapters, where some results and order of magnitude calculations of these effects are presented.

The theory of velocity determination as presented applies for radar echoes only, and is modified in the CW case by the presence of the coherent ground wave. As shown by Mc Kinley (1949), the continual presence of a reference phase causes a whistle before the  $t_0$  point (as in the radar case a post  $t_0$  whistle occurs also). Before the meteor velocity can be calculated for such an echo, a knowledge of the relative phases of the sky wave and groundwave is required for all the Fresnel zones used in the calculation. Mainstone (1960) developed a technique for this, utilizing the doppler beat which occurs as the trail drifts (after formation) under the action of atmospheric motion. The minima of the resultant waveform are produced when the skywave and groundwave are  $\pi$  out of phase, and, provided the doppler beat frequency is measurable and constant, the phase can be extrapolated back along the echo.

The presence of the pre  $t_0$  whistle is indeed fortunate, since in most cases, the post  $t_0$  whistle is lost in the resultant "body doppler" caused by the drift of the trail as a whole.

As shown by equation 4.7.4, the presence of a wind grad-

ient will produce a time dependent velocity of trail drift. Because of this, a linear extrapolation of doppler phase back into the pre  $t_0$  whistle can lead to errors in velocity and  $t_0$  point determination. These were estimated by Mainstone as less than 2%. The present investigation has not seriously modified this figure.

Note added in proof.

No mention has been made of the meteor particle as a source of turbulent energy. This question was discussed at some length at the Symposium on the Fluid Mechanics of the Ionosphere (New York 1959). The general concensus of opinion was that whereas the turbulent wake could be of importance for the larger visual meteors, the contribution of radio meteors to the turbulent energy is negligible.

## CHAPTER V.

### The Adelaide Equipment for Wind Measurement

#### 5.1 General

The original apparatus used for the continuous, automatic tracking of meteor trails at Adelaide was developed in the early 1950's by Elford, Liddy and Robertson. The receiving equipment was installed in a disused ammunition loading store at Salisbury, the transmitter being operated in the Physics Department in Adelaide. Receiver and transmitter separation was necessary since a continuous wave technique was used, and the separation of 20KM produced a groundwave at the receivers of approximately the same level as the skywave reflected by a typical meteor trail.

By the end of 1957, the site at Salisbury was becoming more and more unsuitable because of interference caused by increased industrialization of the area. Also, the equipment, while still producing worthwhile results, had become obsolete and required considerable maintenance. It was decided, therefore, to build a new receiving station at St. Kilda, an electrically "quiet" area some 14KM to the west of Salisbury, and in 1958 the building of equipment for the new

station commenced. The development of a refined technique for meteor velocity determination from CW echoes by Mainstone (1960) about this time initiated a project designed to measure the orbits of individual meteors. This involved the construction and installation of an additional two outstations; these are treated in detail in Chapter VIII. Inherent in the orbit determination is the measurement of the separation distance between the reflection points associated with any two receivers, and this information is utilized in the turbulence investigation (see Chapter VII).

## 5.2 The Transmitters

The transmitters used for the 1961 survey have been in operation for several years, doing routine wind measurements. The CW transmitter was modified by Mainstone in 1958, and produces an output power of approximately 300 watts at a frequency of 26.773 MCS (wavelength = 11.2 metres). This power is radiated within a cone  $\pm 45^\circ$  about the zenith by a three element Yagi array suspended above the Physics building at the University. A similar array is used to radiate the 10KW peak power pulse from the radar transmitter, which operates on the same frequency and is phase coherent with the CW. Pulse width is 10 $\mu$  sec; repetition frequency 100 pulses per second. The arrays were raised and lowered until a suitable groundwave level (some 8  $\mu$ v CW) was received at St. Kilda.



MAIN STATION, ST. KILDA

This CW groundwave is used as the reference phase in the drift measurement by the doppler method, and the radar groundwave triggers the range marker generator at the receiving site.

The main receiving station at St. Kilda was operational in July, 1960, and all facilities, including the outstations, were producing results by December of that year.

### 5.3 The Main Station Mean Wind Measuring Equipment.

The method of determining atmospheric mean motions at meteor heights from the line of sight drifts of trails measured by the doppler shift on CW signals reflected from such trails is due to Manning, Villard and Petersen (1950); as used at Adelaide it contains some modifications in the direction finding and sense of drift determinations. The two basic assumptions, that of aspect sensitive reflection from the trails, and a predominantly horizontal mean atmospheric drift, have been well substantiated both theoretically and experimentally.

The mean wind survey requires the measurement of four parameters for each echo. These are

- 1) the line of sight drift of the trail
- 2) the direction cosines of the reflection point relative to NS and EW axes
- 3) the height of the reflection point

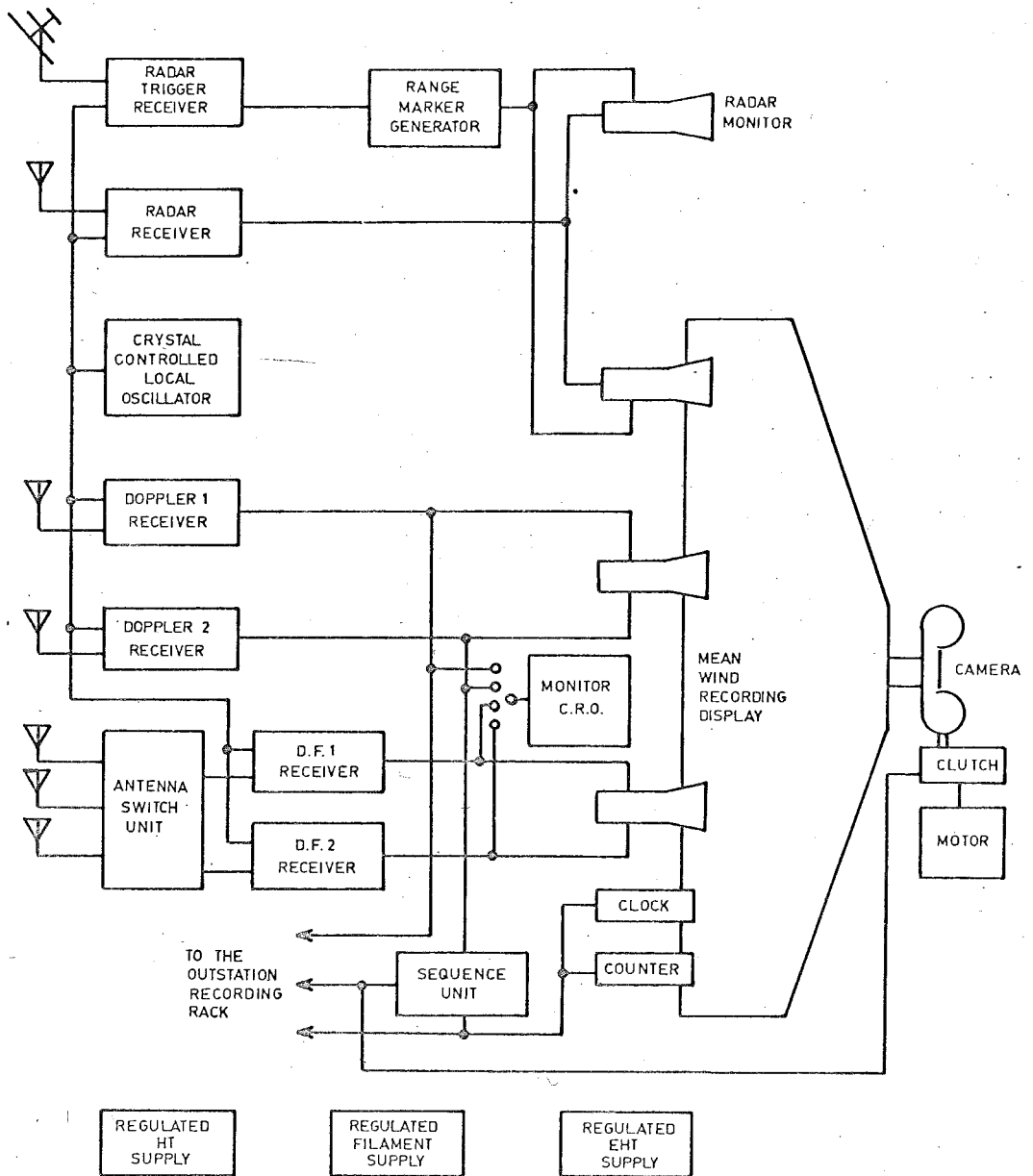


FIG.11 - MAIN STATION MEAN WIND MEASURING EQUIPMENT.



and 4) the time of occurrence of the echo.

Continuous wave techniques are utilized in the determination of 1) and 2), and 3) is found by using the supplementary pulsed transmitter and radar receiver to determine the slant range of the trail which, together with 2), yields the height of reflection. Fig. 11 is the block diagram of the receiving section of the Mean Wind Measuring Equipment (henceforth referred to as the Wind Equipment).

Five spaced dipoles, each  $\lambda/4$  above ground, and four narrow band receivers are used to determine the direction of arrival of the reflected wave. The orientation of these aeri-als is such that the groundwave always presents a plane wave front to the dipoles. The relative phases of the sky-wave and groundwave are established at the aeri-als in terms of the relative phases of the low frequency doppler beats. Therefore, there is no necessity to preserve the R.F. phases through the receivers. Two of the receivers, labelled D.F.1 and D.F.2 in Fig. 11, are switched between three of the aeri-als. Their outputs are D.C. amplified and displayed on a double beam C.R.T in the Wind display. The traces are identified by asymetry of the switching intervals, giving "light" and "dark" traces.

The two Doppler receivers continuously monitor the C.W. level. The sharp rise (or fall) of the signal level which

occurs when a suitably oriented trail is formed in the atmosphere, triggers the sequence unit which is connected to the output of one of these receivers. The sequence unit brightens the oscilloscopes in the Wind display, and energizes the magnetic clutch on the Wind camera. The normal recording interval is of the order of one second. The sequence unit also provides a triggering pulse for the equipment which records information required for turbulence and orbit measurements.

The receivers and D.C. amplifiers are the same as those used in the outstation equipment, and are described in detail in Chapter VIII.

The range of the reflection point is recorded using a conventional intensity modulated A-scope display. A wide band receiver is connected to a sixth dipole, and its output intensity modulates another C.R.T in the Wind display. The time base and range marker generator is triggered by the output of an additional broadband receiver connected to a horizontal three element Yagi array beamed on Adelaide. This ensures that the timebase is always triggered on the groundwave pulse even when the skywave amplitude (as received on a dipole) exceeds the groundwave level.

A clock which is illuminated every quarter hour, and a counter and electronic flash which are fired on each echo

completes the display line-up.

A typical echo as filmed is shown in Fig. 13.

The sense of trail drift (towards or away from the receivers) can be determined from the relative phases of the doppler outputs of the four narrow band receivers, but this is a rather tedious process of elimination, and a much better method which gives the sense of drift on sight from any one output trace was devised by Robertson in 1952. The CW transmitter at Adelaide is phase modulated by a sawtooth wave synchronized with the 50 cycle mains which slowly advances the R.F. phase of the transmitter output and then suddenly retards it by  $90^\circ$  fifty times per second. Normally, this phase modulation produces no receiver output, since the receivers all employ conventional A.M. diode detectors. However, during the presence of an echo the phase retarded groundwave reaches the receivers at St. Kilda about one millisecond before the phase retarded skywave, so that during this brief period the vector triangle RGS in Fig. 12 is replaced by  $R'G'S'$  in which  $G'$  makes an angle of  $-90^\circ$  with G, but the phase of the skywave is unchanged. The brief change from R to  $R'$  is recorded as a spike (of duration  $\sim 1$  millisecond) on the doppler beat curve at  $1/50$  sec. intervals. It can be seen from the record of Fig. 13 that the ends of these spikes trace out a "phantom" beat curve which in this case

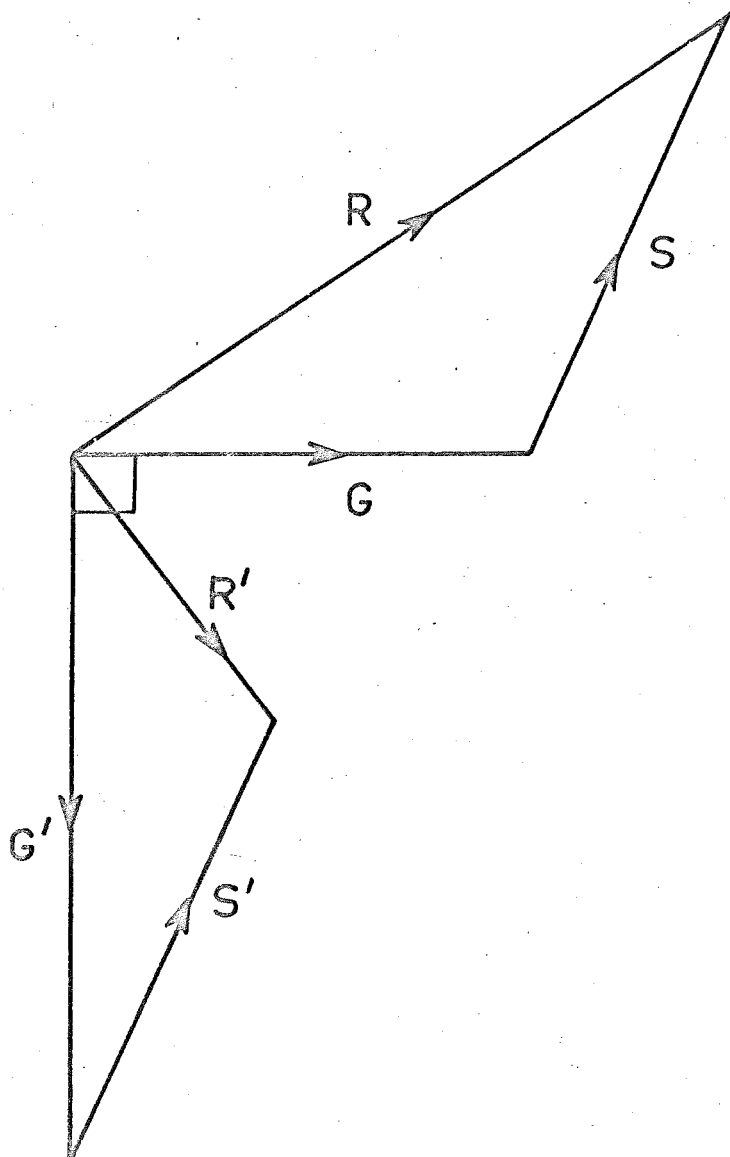


FIG. 12

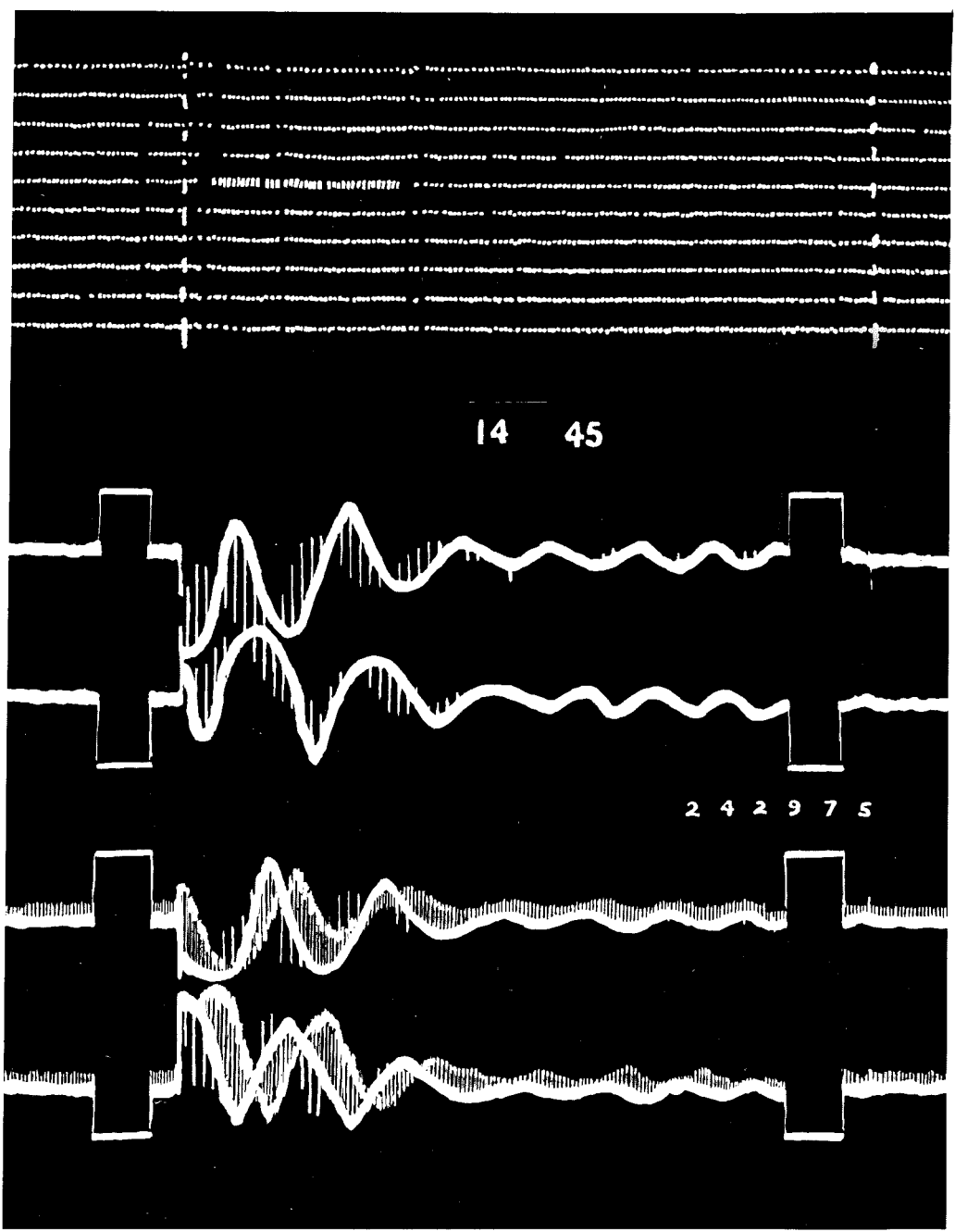


FIG 13-MEAN WIND RECORD



FIG 14 - MEAN WIND RECORDING RACK

leads the main doppler beat curve.

The phase of the phantom beat curve either leads or lags the phase of the main doppler beat curve according as the phase path of the sky wave is either increasing or decreasing. With reference to Fig. 12, if the phase path is increasing, S is rotating clockwise and R is approaching its maximum value as  $R'$  is approaching its mean value. Thus R leads  $R'$  in phase. Similarly if S is rotating anticlockwise R is approaching its mean value as  $R'$  is approaching its minimum value so that R lags  $R'$  in phase. In the actual record of Fig. 13, R lags  $R'$ . This represents a decrease in the phase path, and the line of sight component of the drift velocity of the meteor is directed towards the T/R system. These sense spikes on the traces also give convenient 20 millisecond time markers by means of which the doppler frequency can be measured.

In the interests of stability, all H.T. power supplies are electronically regulated, and all valve filaments are fed from a constant voltage transformer. The receiver and display power supplies use a highly stable circuit devised by Attree (1955). The output impedance of these supplies is less than 0.1 ohm from D.C. to 50kc, and ripple is less than 10 millivolts at a high tension output of 200ma at 250v.

Fig. 14 is a photograph of the Mean Wind Recording Rack,

housed in the station at St. Kilda.



## CHAPTER VI.

### Preliminary Investigations at Adelaide

#### 6.1 Methods of Mean Wind Data Reduction.

For the years 1952 to 1962 inclusive, equipment similar to that described in Chapter V has been recording winds in the meteor region above Adelaide. Almost from the commencement of the wind project it was realized that velocity gradients with height existed, and the method of data reduction included arbitrary stratification of the region into three bands 10km thick, centred on 80, 90 and 100km respectively. Stratification with narrower height grouping was contemplated, but was not commensurate with either the accuracy of ranging or the echo rate, and was therefore not implemented.

The echoes are grouped in hourly intervals. Because of the echo rate, it has been necessary to record for several days each month, and to lump the data to give winds for a "typical day" of the month.

For any given height group and time, at least two echoes are required to specify the appropriate mean wind velocity vector, since only the line of sight component of drift can be recorded for each echo. If more than two echoes of the

appropriate height and time are recorded, a least squares fit to the resultant NS and EW vector components is used to obtain the required wind vector. This analysis is produced in the Elford et al (1953) paper.

The data read from the wind films is punched onto cards and processed on an IBM 1620 computer to give the resultant NS and EW wind components. The appropriate IBM FORTRAN programme appears in Appendix I.

The mean hourly winds for a given height range of the typical day for the given month are then subjected to a Fourier Analysis to determine the prevailing, diurnal and semi-diurnal components. This is also carried out on the 1620 computer (programme, Appendix I).

## 6.2 Random Winds.

An early investigation by Elford et al of trails assuming aspect sensitivity close to the zenith showed that vertical drift velocities were small in comparison with the horizontal drifts, and that the mean motion was confined to within  $\pm 15^\circ$  of the horizontal. Vertical motions of up to 10 metres / second with a vertical mean drift of 1 to 2 metres / second were found, associated with horizontal winds of 80 metres / second.

One of the most interesting features of the winds as recorded is the reproduction year after year of the same

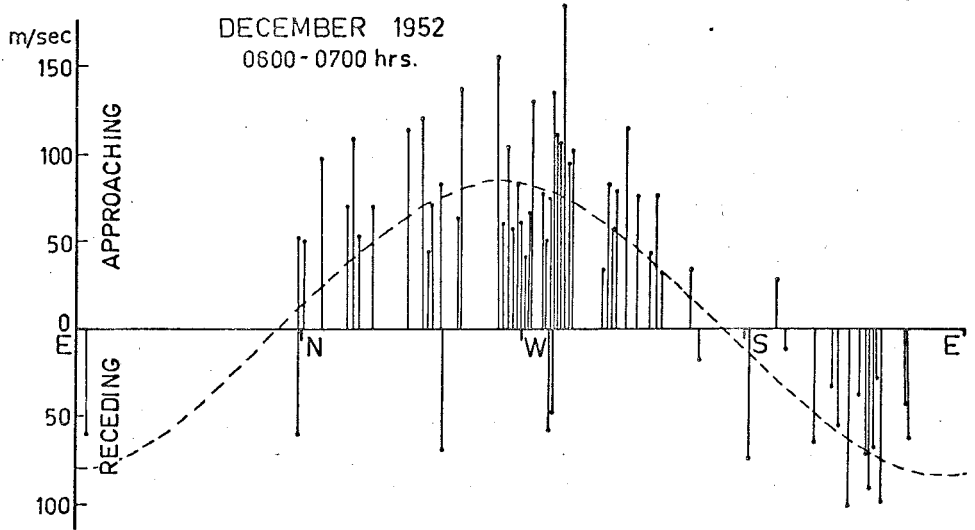


FIG.15a - MEASURED WINDS

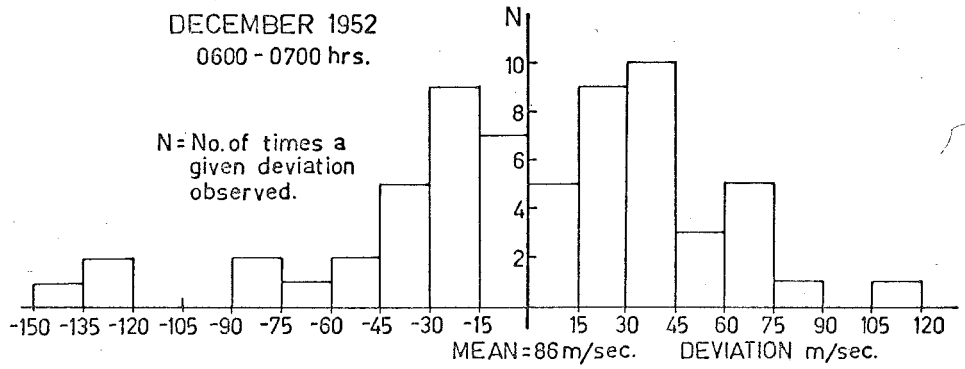


FIG.15b - VELOCITY DEVIATIONS FROM THE MEAN WIND

basic wind patterns for a given month. This leads to the conclusion that turbulent motions, if present, are subjugate to the mean flow.

To obtain some indication of the probable magnitude of the turbulent velocities, the mean wind was extracted from the records for 0600 to 0700 hours, December 1952 (Fig. 15a). The result is the histogram of Fig. 15b. The value of the mean wind at the time was 86m/sec, and the most probable turbulent velocity of the order of 30m/sec. This agrees reasonably well with the RMS velocity of 50m/sec for a mean wind of 100m/sec (Manning, 1959), and also with the 25m/sec turbulent velocity found by Greenhow to be an average regardless of the mean wind speed.

The size of the eddies responsible for this turbulent velocity cannot be deduced from the Adelaide single station observations. A narrow transmitting and receiving beam width is required to determine time correlations (an area of the meteor region small in comparison with the size of the disturbance must be illuminated), and the Adelaide system has always used broad all sky antennae.

### 6.3 The Effect of Wind Shears on Long Duration Echoes.

The effect on the meteor trail of a linear shear has been shown in Chapter IV Section 7 to produce a rotation of the trail which results in motion of the reflection point

along the trail. Since in the determination of the wind gradients along a trail the separation distance is of prime importance, any factor which may effect reflection point motion is worthy of consideration.

The rate of motion of the reflection point in kilometres/sec is given by equation 4.7.7, viz.

$$T = R_0 u' \times 10^{-3}$$

where  $R_0$  is the slant range of the echo in km and  $u'$  is the trail gradient expressed in m/sec/km. The assumption of a linear gradient is, of course, an oversimplification, but it is useful for determining order of magnitude values for the quantity  $T$ . If the gradient were truly linear, then there would be no relative motion with time of any two points on the trail, and errors would be incurred only in the absolute and not in the relative drift velocity determinations.

Velocity gradients, when encountered, are characterized by a time dependent drift velocity, observed on the recording film as a doppler beat increasing or decreasing with time.

The actual drift velocity at the  $t_0$  point is related to the observed drift velocity at a time  $t$  after the meteor has reached the  $t_0$  point by equation 4.7.6, viz.

$$u_a = u_o - u'^2 R_0 t$$

If the gradient  $u'$  is zero, then the observed drift is equal to the actual drift.

Since by far the greater majority of long duration echoes exhibit this change in doppler frequency with time, it is reasonable to suppose that the trail is always subject to deformation by some form of wind gradient. There is always a finite delay after trail formation before the echo is recorded, and the execution of each doppler cycle in itself introduces a further time interval over which measurement of the doppler is being made. Thus there is always some difference between the measured and actual values of the line of sight drift vector.

To determine the typical order of magnitude of the time dependent drift velocity variation and reflection point motion, 70 echoes for December, 1960, were examined in considerable detail. The average line of sight drift vector for these echoes is 42 metres/sec; the RMS value of the rate of velocity change with time is 9.1 metres/sec/sec. Allowing for an average time after the  $t_0$  passage of the meteor before measurement of 0.1 secs, the average absolute velocity error is only of the order of 1 metre/sec, and is of the same order as the normal film reading errors. However, the rate of reflection point motion is of more consequence. Fig. 16 shows the number of echoes observed with a given reflection point motion (calculated from the doppler variation with time, and assuming a linear gradient). Whereas the average motion of

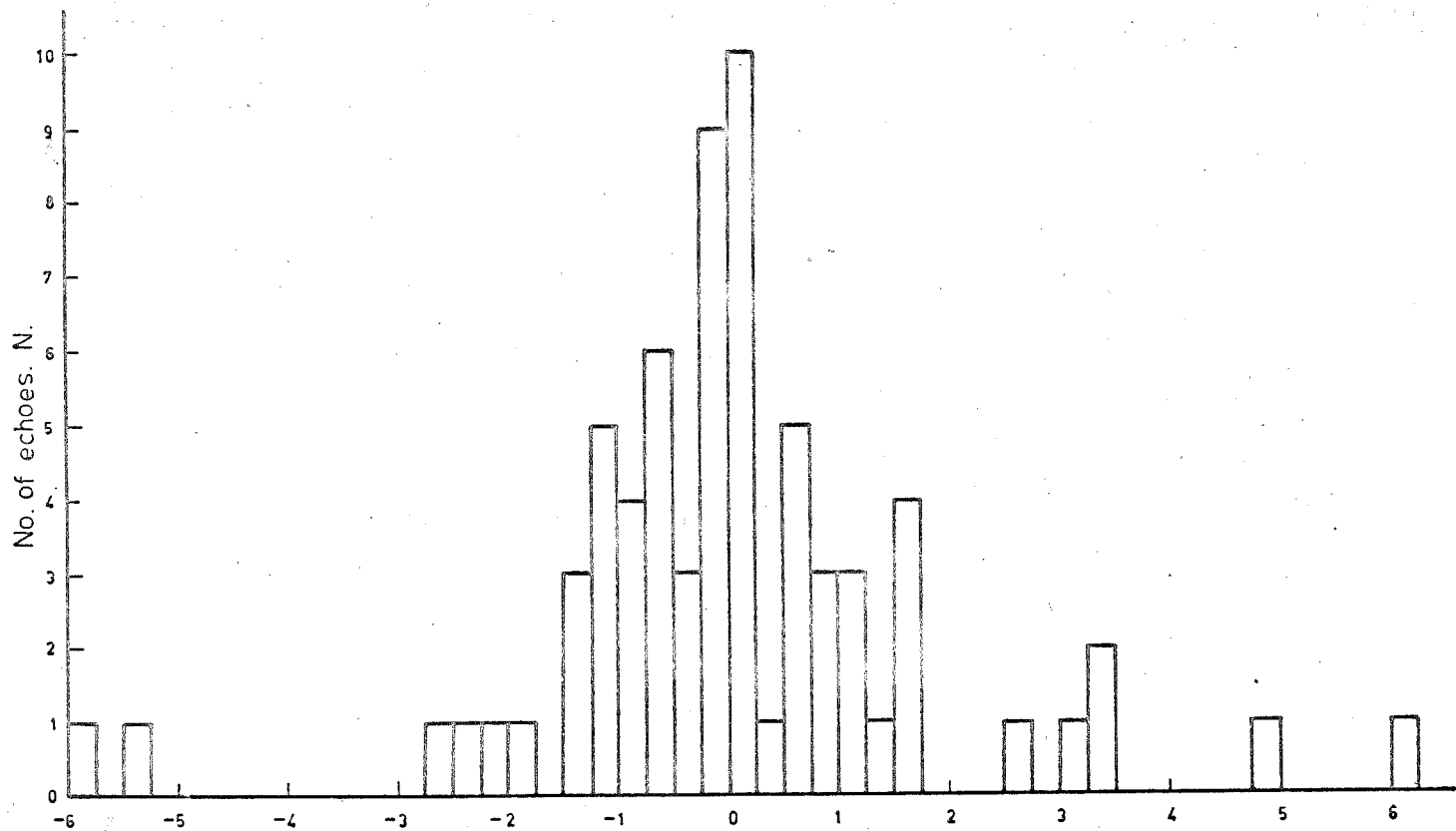


FIG.16 - Motion of reflection point along trail, (km./sec.)

DECEMBER 1960.

the reflection point along the trail is zero, values up to 6 km/sec are found, the RMS value being  $\sim 1$  km/sec. Thus the drift velocity measured 0.1 sec after time  $t_0$  is, on the average, the drift velocity of a point 100 metres away from the  $t_0$  point. If the wind gradient along the trail is not linear (which is usually the case, as is shown by reference to Fig. 17a, where the change in drift velocity with time is by no means constant), then the separation of any two reflection points on the trail will be highly time dependent. This places a rather severe limitation on the measurement of trail shears at small separations ( $\sim 200$  metres).

If the echo duration is sufficient, the action of non-linear shears may cause more than one portion of the initially straight trail to be normal to the line of sight direction and hence give equally strong reflections. If these portions are moving at different relative velocities, interference between the reflected waves will be observed at the receiver. This has proved to be a complication for methods of investigation of meteor trails based on the measurement of the passage over the ground of diffraction patterns caused by reflection of radio waves from such trails. The effect is of little consequence with a doppler method for short duration echoes.

A typical beat envelope type of long duration echo is



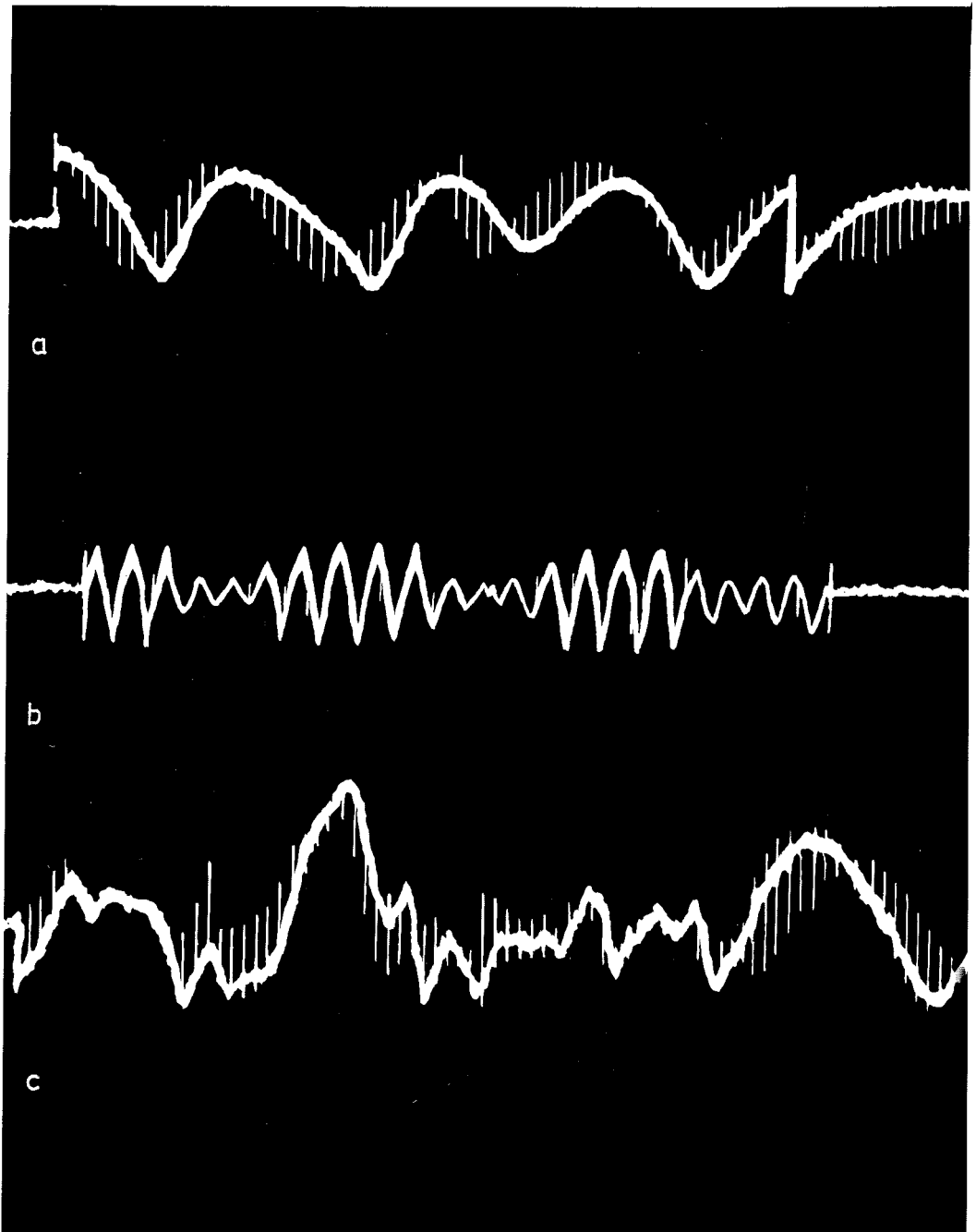


FIG 17- TYPICAL ECHO TYPES

shown in Fig. 17b. This presumably has only two reflecting centres, moving with small relative velocity. The echo of Fig. 17c, however, depicts what one might almost describe as a chaotic situation.

Anomalous echoes can also be caused by the fragmentation of meteors. This can give rise to blobs of high ionization lying along the path of the meteor. These regions simultaneously produce reflections, and the trail is no longer a line source. Relative motion of these blobs gives rise to a beating type echo.

Occasionally, more than one meteor echo will occur at a given time, the received signal being a combination of the two doppler beats. This type of echo is usually readily recognized because of its associated multiple range.

A further investigation of the interesting properties of long duration echoes is contemplated.

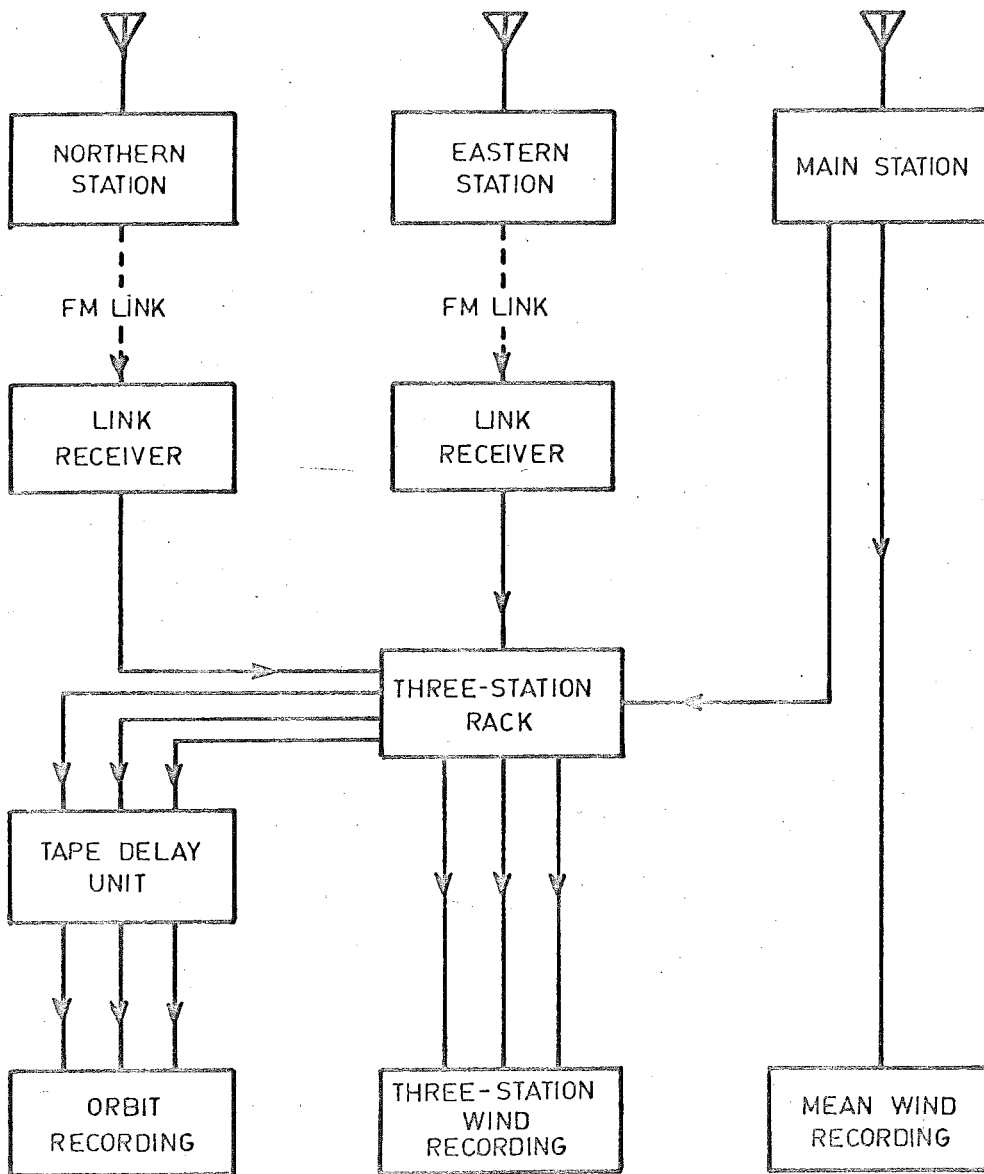


FIG.18 - THE THREE-STATION SYSTEM.

## CHAPTER VII.

### The Adelaide Three Station System

#### 7.1 General Description of the System.

Since the reflection of radio waves from a meteor trail after formation is, in the first instance, specular, the spacing of receivers on the ground will result in each receiving information from a separate point on the trail. In the Adelaide system three receiving stations are used. The operation of two of these stations is completely automatic in that they run unattended except for routine maintenance, and continuously transmit the signals received back to the main receiving station at St. Kilda.

Fig. 18 is a block diagram of the overall system. One outstation is located at Sheedys' Farm, 4.72 km north of the St. Kilda receiving site, and the other outstation is at Direk, a disused railway siding 4.81 km to the east of the main station (see Fig. 19). For convenience in the specification of the geometry of the system for purposes of data reduction, the transmitting and receiving sites determine a rectilinear system of coordinate axes (see Fig. 20). The transmitters at Adelaide, the St. Kilda main station and the

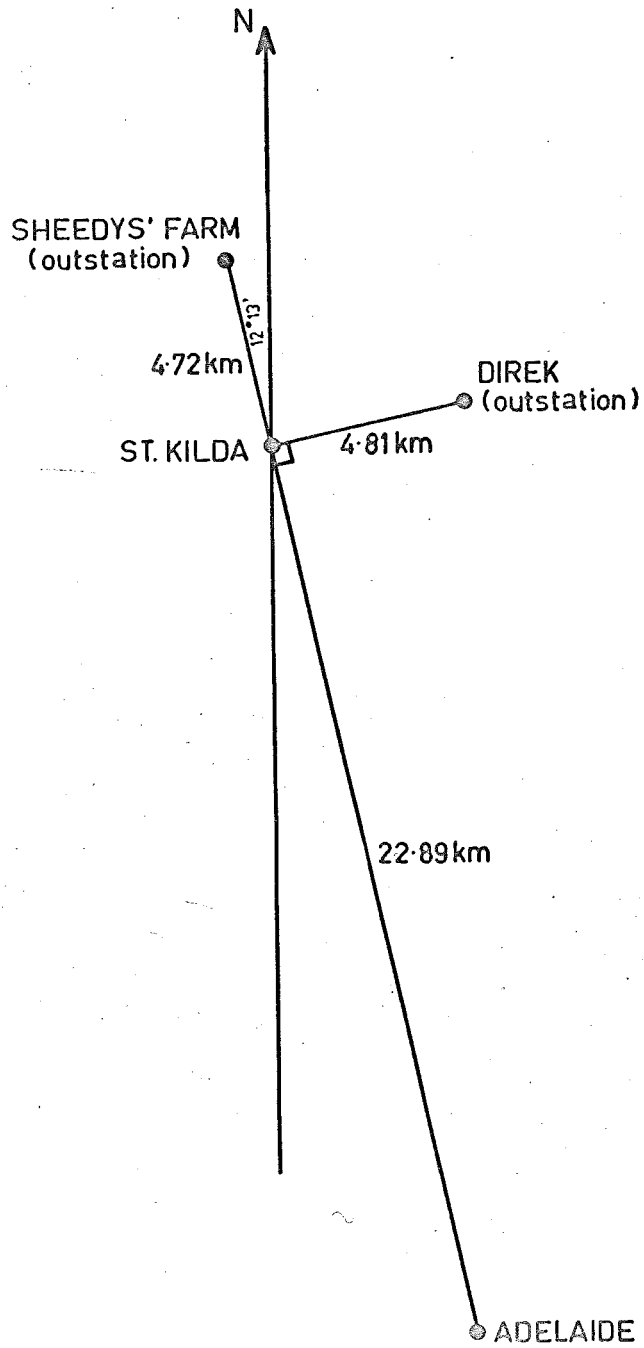


FIG.19 - PLAN of TRANSMITTING and RECEIVING SITES - 1961.

Sheedys' Farm outstation are colinear, and the St. Kilda - Direk line is perpendicular to this axis.

To simplify the correlation of echoes received at the various stations, all recording is performed at the main station. Three sets of film records are obtained, each concerned with a different aspect of the investigation of the astronomical properties of meteors, and the effects of atmospheric motions on meteor trails.

By recording the initial whistle accompanying trail formation as observed from each outstation, not only can the meteor velocity be determined (from the rate of formation of the Fresnel zone pattern) but also the time of onset of specular reflection,  $t_0$ , appropriate to each station can be specified. From the meteor velocity and the different times of commencement of the echo at each station, the spacial separation of the appropriate reflection points is readily calculated. This spacial separation is the parameter  $\xi$ , which, together with the line of sight trail drift velocity as determined from the body doppler received at each station, provides the basic data for the turbulence investigation.

As has already been shown in Chapter V, the record taken from the main station receivers uniquely specifies the position of one point on the trail. The coordinates of this point, together with the appropriate  $\xi$  values and a knowledge

of the outstation system geometry is all that is required to determine the direction cosines and the position of the meteor trail in the upper atmosphere. From these direction cosines and the velocity of formation of the trail, the orbit of the meteor particle producing the trail can be determined. The basic measurements of meteor velocity and spacial separation of the reflection points were made by a co-worker, Carl S. Nilsson, who completed, concurrent with the turbulence programme, a thirteen month survey of the orbits of shower and sporadic meteors as observed from the southern hemisphere.

#### 7.2 The Choice of the Receiving Station Separations.

For the investigation of atmospheric turbulence from spaced station radio echo observations of meteor trails, there are two main factors which govern the choice of station separation. These are the finite lengths of the trails formed by meteors in the upper atmosphere, and the scale of the turbulence. From the point of view of the orbit project, as large a station separation as possible is desirable, since the accuracy of determination of the direction cosines of the trail increases with increase of reflection point separation. If only a small number of zones of the Fresnel diffraction pattern are discernable on the orbit film record for a given meteor, there is an inherent inaccuracy in the absolute spec-

ification of the  $t_0$  points appropriate to each receiving station. If the time difference between the  $t_0$  points are large, the effect of this inaccuracy of individual  $t_0$  determination on the computed direction cosines of the trail will be small.

The upper limit of station separation is determined by the lengths of the trails formed in the atmosphere. Trail lengths are, on the average, about 20km. If large station separations are used, the number of meteors producing echoes at all three stations will be small. The station separation of 5km used in the Adelaide three station system represents a good compromise between accurate trail direction cosine determination and useable echo rate.

It can be shown geometrically that the maximum separation of reflection points on a suitably oriented trail is half the station separation (See Section 4 of this Chapter). Thus the maximum  $\xi$  value associated with a station separation of 5km is 2.5km. Since most of the controversy associated with investigations of atmospheric turbulence in the meteor region has been concerned with the scales of size 1km or less, a station separation of 5km can be regarded as almost ideal for the determination of the properties of small scale turbulence in the height range from 80 to 100km.



### 7.3 The Use of Shower Meteors in a Turbulence Investigation.

While the turbulence and orbit projects were being developed, it was not altogether certain that the radiants of sufficient meteors would be measured in any one month to give a significant sample for the purpose of turbulence measurements. For this reason, an alternative system was devised, based on the known properties of shower meteors.

The extraction of shower meteors from the sporadic background is made possible by the aspect sensitivity of reflecting trails. A northerly radiant produces trails which assume aspect sensitivity only to the south of the receivers. For a given shower radiant, at any one time the reflection points of such shower meteor trails lie on a horizontal line perpendicular to the radiant direction. (This assumes a flat earth, and that all meteors form trails at the same height, both reasonable approximations). Thus, if the time of occurrence of the echo is recorded, and its reflection point lies on the line appropriate to the radiant at that time, then it is highly probable that the meteor belonged to the given shower. A clock in the wind equipment display records quarter hourly, and this interval is sufficiently small to provide quite accurate grouping. From the time of the echo and the known radiant of the shower, the appropriate trail direction cosines  $\lambda, \mu, \gamma$  can be calculated. To calculate the reflection point

separations, consideration must be given to the geometry of the trail with respect to the spaced stations.

#### 7.4 The Outstation Geometry. (Fig. 20).

If the trail  $\xi$  has direction cosines  $\lambda$ ,  $\mu$ ,  $\nu$ , and  $A_0P_0$  is perpendicular to  $P_1P_0$  (specular reflection) then, putting  $q = R_0A_0$

$$(lr_0 + q)\lambda + mr_0\mu + nr_0\nu = 0$$

i.e.

$$r_0(l\lambda + m\mu + n\nu) = -q\lambda$$

7.4.1

A first order approximation is given by

$$q = + \frac{d}{2}$$

i.e.  $A_0$  is the midpoint of  $R_0T$

This approximation is adequate for the order of accuracy required in the turbulence investigation. The true solution is obtained by making the distance  $R_0P_0 + P_0T$  a minimum, and is used in the orbit calculations where a higher degree of accuracy is required.

Substituting  $q = \frac{d}{2}$  in 7.4.1 gives

$$r_0(l\lambda + m\mu + n\nu) = - \frac{d}{2}\lambda$$

7.4.2

The equation of the trail, in terms of the parameter  $\xi$ , is

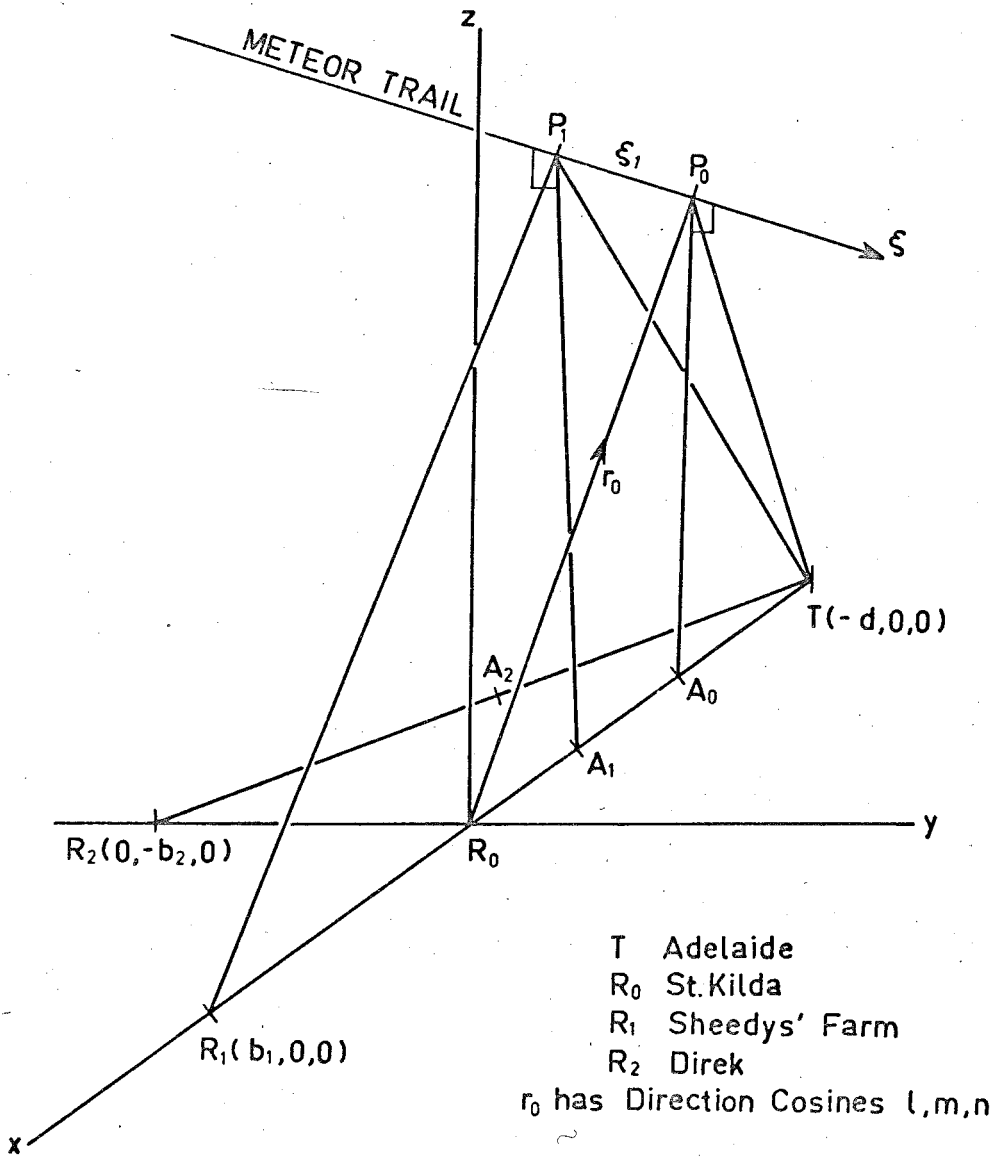


FIG.20 - THE GEOMETRY OF THE SYSTEM

$$x = lr_0 + \lambda\xi$$

$$y = mr_0 + \mu\xi$$

$$z = nr_0 + \nu\xi$$

For the first outstation  $R_1$ ,

$$A_1 \text{ has coords. } \left(-\frac{d-b_1}{2}\right), 0, 0$$

and at  $\xi = \xi_1$ , the point of specular reflection from T to  $R_1$ , the direction cosines of  $A_1P_1$  are

$$\frac{1}{A_1P_1} \left[ lr_0 + \lambda\xi_1 + \frac{d-b_1}{2}, mr_0 + \mu\xi_1, nr_0 + \nu\xi_1 \right]$$

$A_1P_1$  is perpendicular to the trail

$$(lr_0 + \frac{d-b_1}{2} + \lambda\xi_1)\lambda + (mr_0 + \mu\xi_1)\mu + (nr_0 + \nu\xi_1)\nu = 0$$

i.e.

$$\lambda(lr_0 + \frac{d-b_1}{2}) + \mu(mr_0) + \nu(nr_0) + \xi_1 = 0$$

7.4.3

From 7.4.3 and 7.4.2,

$$\lambda \left(\frac{b_1}{2}\right) + \xi_1 = 0$$

i.e.

$$\xi_1 = -\frac{b_1}{2}\lambda$$

For the second outstation  $R_2$

$$A_2 \text{ has coords. } -\frac{d}{2}, -\frac{b_2}{2}, 0$$

Specular reflection from T to  $R_2$  requires

$$(lr_0 + \lambda\xi_2 + \frac{d}{2})\lambda + (mr_0 + \mu\xi_2 + \frac{b^2}{2})\mu + (nr_0 + \nu\xi_2)\nu = 0$$

i. e.

$$\lambda(lr_0 + \frac{d}{2}) + \mu(mr_0 + \frac{b^2}{2}) + \nu(nr_0) + \xi_2 = 0$$

7.4.4

From 7.4.4 and 7.4.2,

$$\mu\left(\frac{b^2}{2}\right) + \xi_2 = 0$$

i. e.

$$\xi_2 = -\frac{b^2}{2}\mu$$

The distance  $b_1$  is the distance from the main station at St. Kilda to the outstation at Sheedys' Farm, viz. 4.72 KM;  $b_2$ , the distance from the main station to the Direk outstation, is 4.81 KM.

Substitution of the appropriate  $\lambda$ ,  $\mu$  for each shower meteor thus yields the separations  $\xi_1$  and  $\xi_2$ .

Because of the reasonably high sampling rate achieved with all orbit information available, this method has not been used to date.

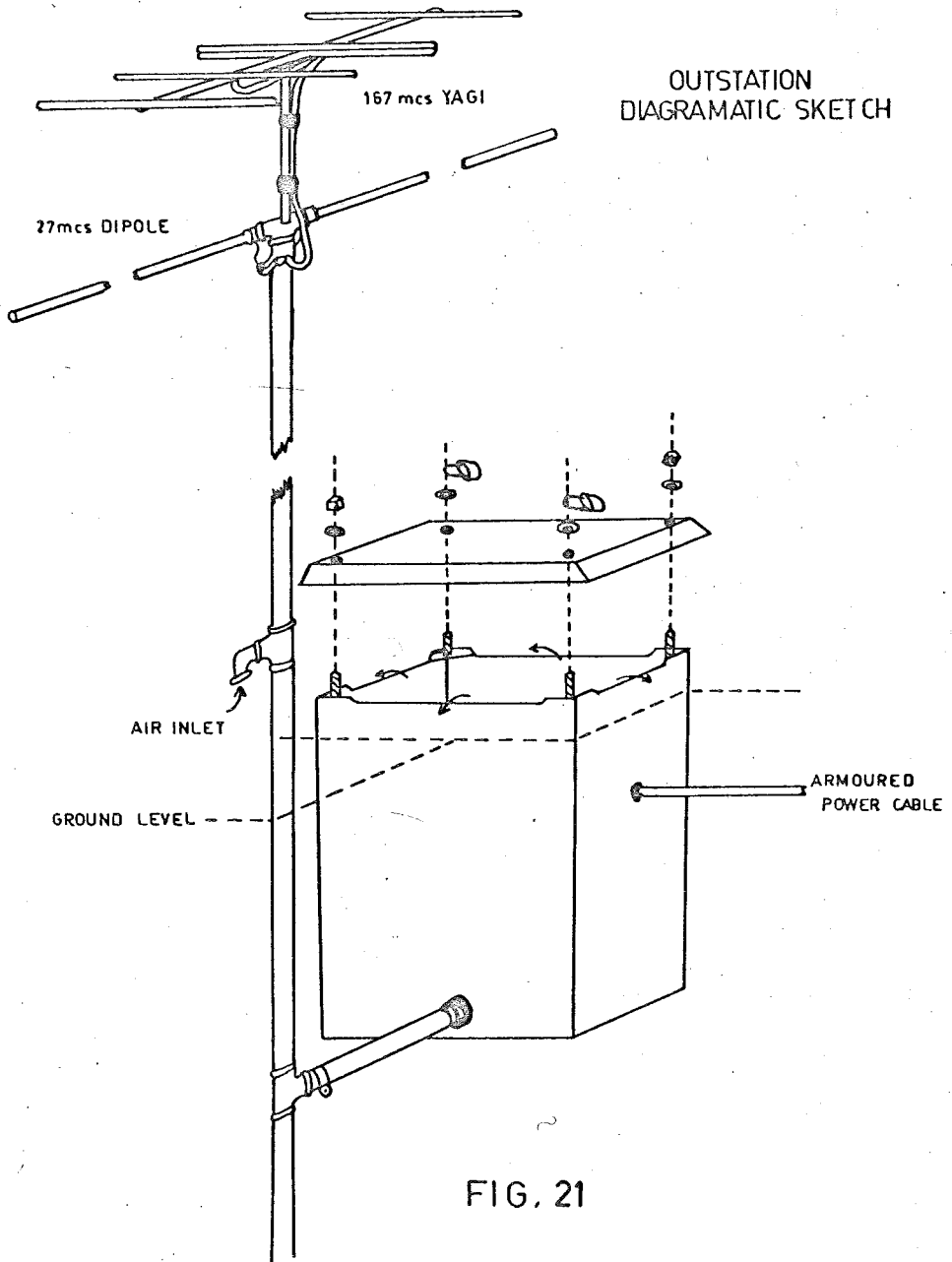
## CHAPTER VIII.

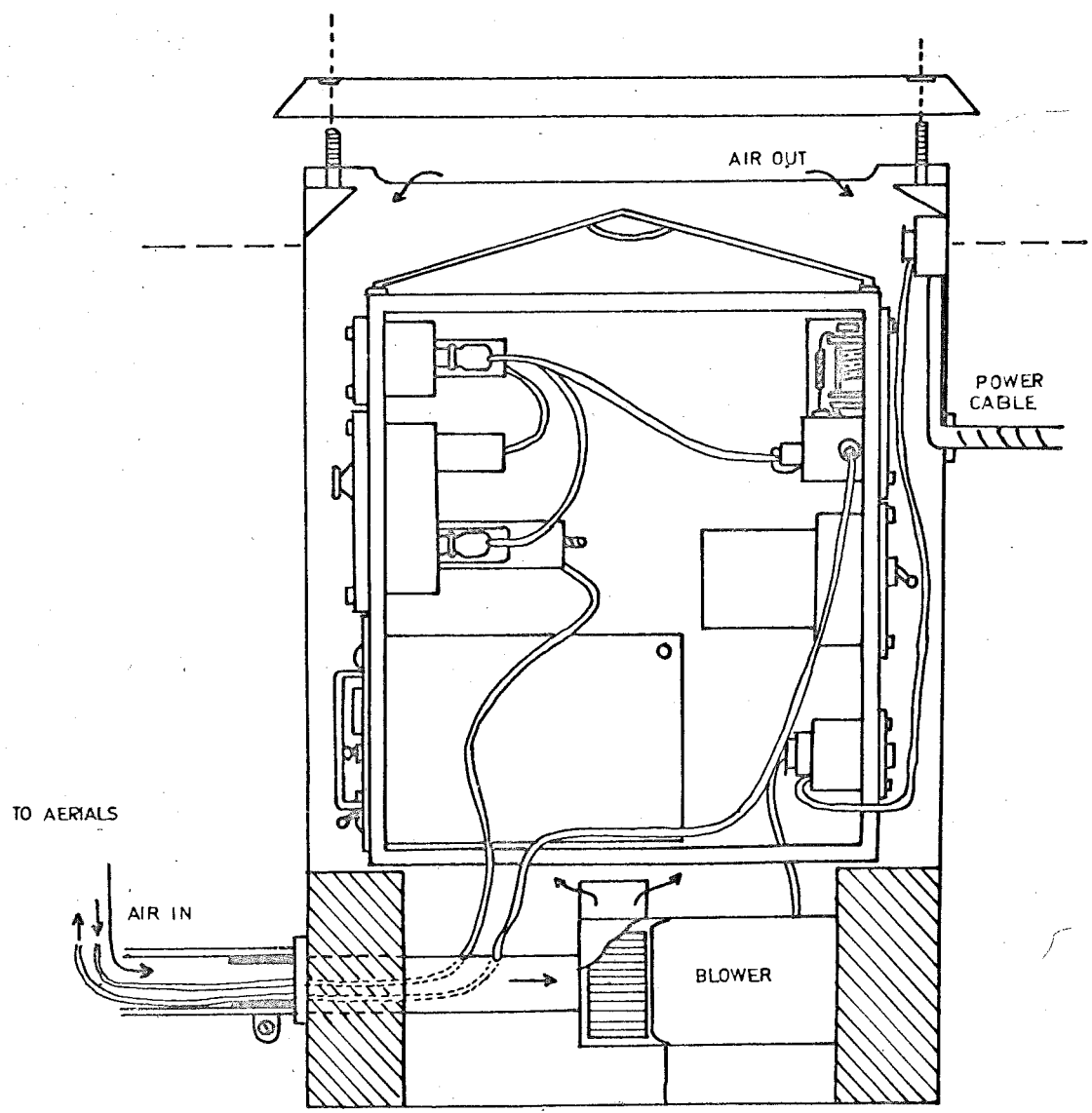
### The Outstation Equipment

#### 8.1 General

Each of the outstations situated at Sheedys' Farm and Direk is completely self contained, and runs unattended except for routine maintenance. For protection from the weather and possible vandalism, the rack of equipment at each is housed in a steel tank buried in the ground with only a padlocked lid visible (Fig. 21). The 27 mcs  $\lambda/2$  receiving dipole is mounted  $\lambda/4$  above the ground on a length of two inch waterpipe, and feeds the receiver via a coax cable which runs down the inside of the pipe and underground into the tank. This pipe also serves as an air intake for the blower mounted in the bottom of the tank. Whereas this blower is not strictly necessary, it does help to remove some of the 300 watts being dissipated by the equipment. The warm air is exhausted through louvres under the lid (see Fig. 22).

A.C. power (240v) is conveyed to the rack by an armoured cable which runs underground from the tank for some 30 yards before being connected to the overhead distribution system. This helps to eliminate interference which might be radiated





OUTSTATION IN SITU

FIG. 22



# OUTSTATION BLOCK DIAGRAM

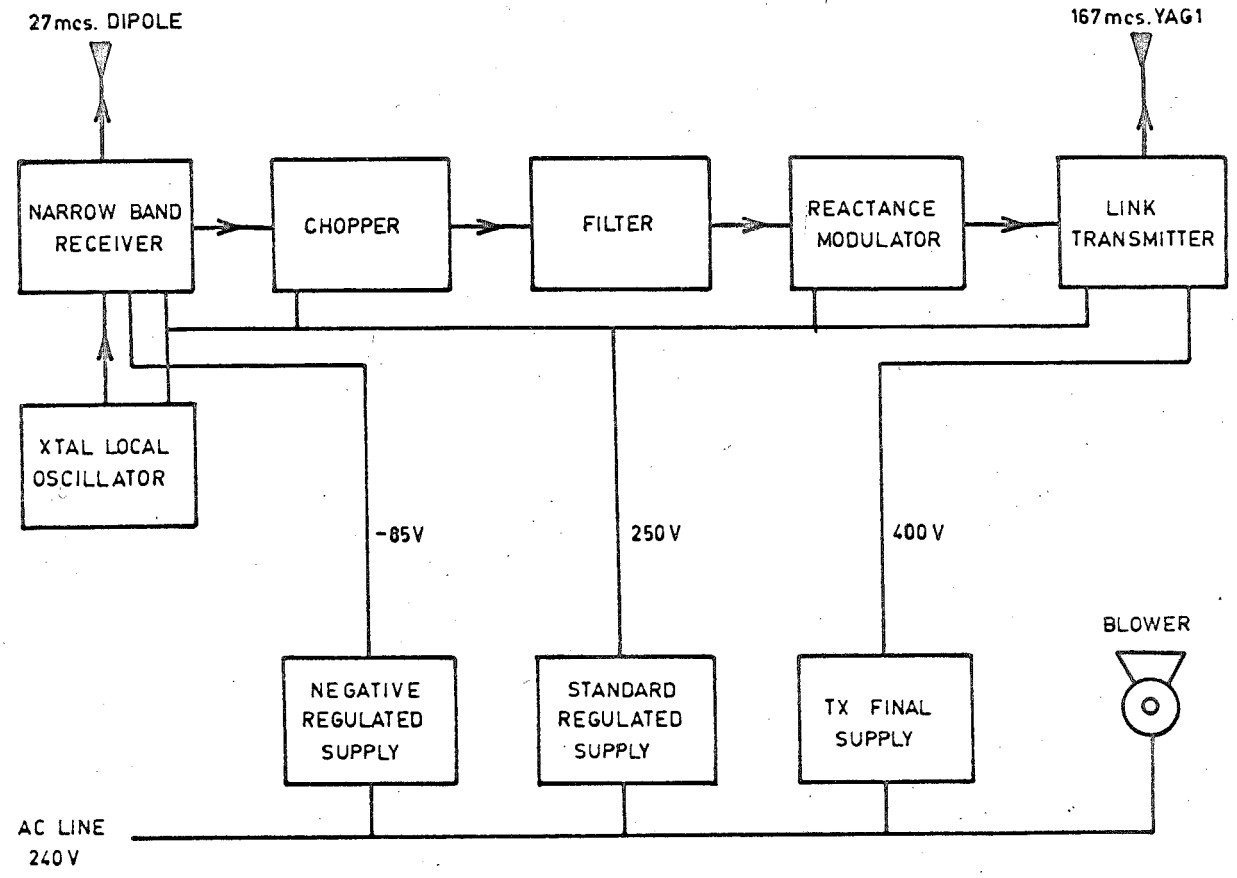


FIG. 23a

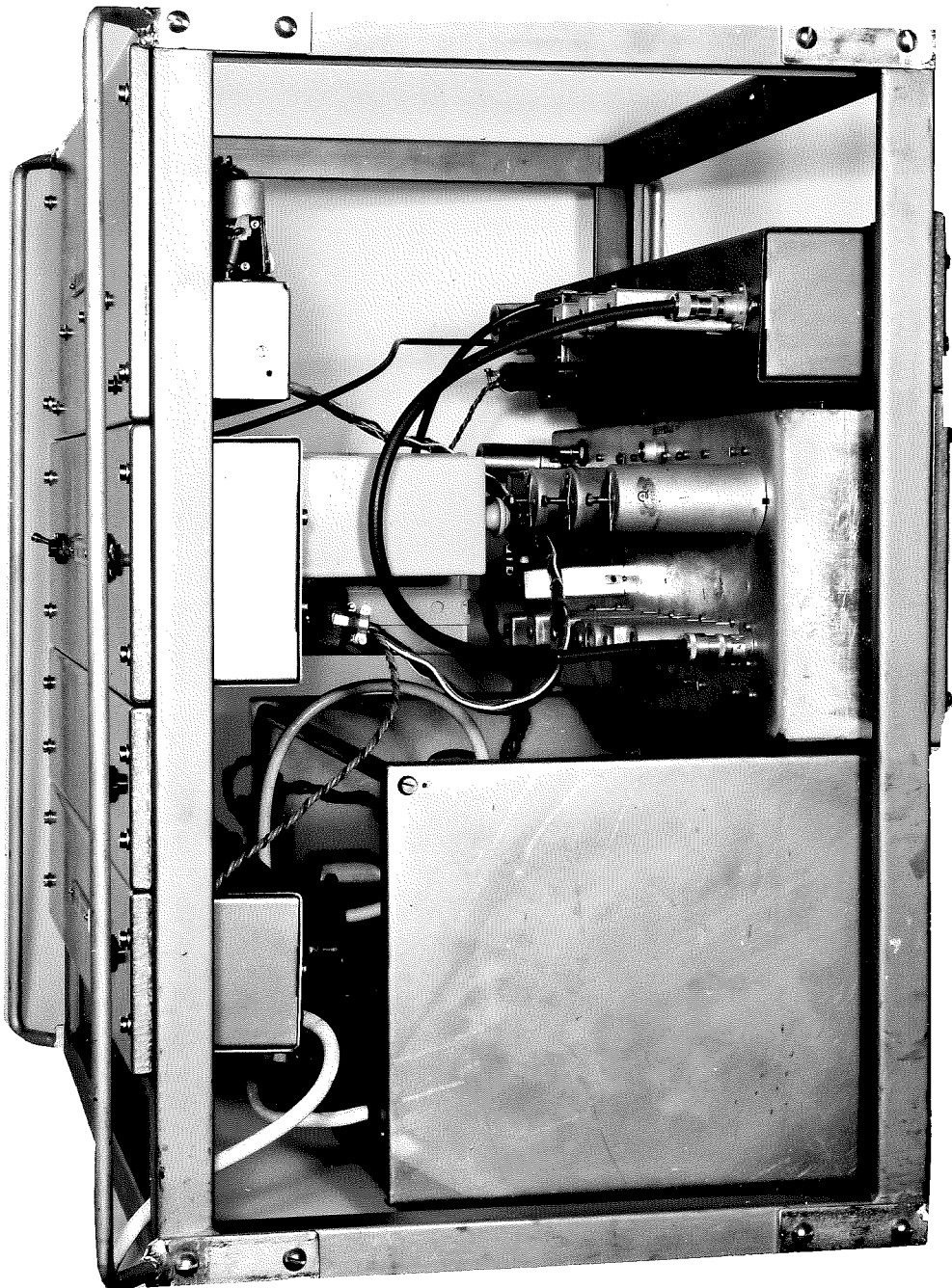


FIG 23b - OUTSTATION RACK

by the power wiring, and to minimise distortion of the aerial polar diagram which could occur due to overhead wires in close proximity to the station.

## 8.2 Detailed Description of the Outstation Equipment.

Each outstation rack (Fig. 23a,b) contains the following:-

- (a) a 27 mcs narrow band superhetrodyne receiver, with crystal controlled local oscillator,
- (b) a chopper and filter unit,
- (c) a frequency modulated 167 mcs transmitter,
- (d) associated power supplies.

### (a) The 27 mcs Receivers.

As mentioned in Chapter V, receivers of similar design are used in each outstation and at the main station. Since both outstations and main station are almost completely free from man-made interference, considerable time was spent on the design and construction of these receivers to meet the requirements of very low noise figure and high gain and stability.

To reduce the noise figure to a minimum, two stages of radio frequency amplification are used. The first is a cascade employing a triode connected 6AM6 and a 6AQ4; the second a 6AM6 as a conventional pentode amplifier. The pentode mixer, another 6AM6, has cathode bias in the interest of

rapid recovery from high amplitude transients (atmospherics). Oscillator injection is crystal controlled. Two I.F. amplifier stages (2 X 6BA6) on 1.9 mcs complete the lineup before detection. Extensive shielding is used between stages; all filament and HT leads to each stage are bypassed by concentric feedthru capacitors. The resultant receiver has a noise figure better than 2 db, image response 72 db down, overall bandwidth  $\pm 4.5$ kc at the 3 db points, and an overall gain of 180db before detector saturation on input circuit noise. Normal gain in operation is 120db, a 10 microvolt input signal producing 10 volts across the detector diode load. Receiver linearity is excellent up to 30 volts output, with a smooth overload up to 40 volts out. Either AVC or manually controlled gain is available. For echo amplitude investigations a knowledge of both absolute groundwave and skywave is desirable, and calibration and operation at fixed gain is necessary. For routine wind and radiant surveys, the constant groundwave output provided by the AVC system simplifies recording.

(b) The Chopper and Filter Unit.

Under normal quiescent conditions the output of the receiver consists of a D.C. signal which is a measure of the groundwave amplitude. In order to transmit this groundwave level back to the main station, techniques similar to those

employed in carrier telephony are used; the carrier wave which is produced by chopping and filtering the receiver output is in a form suitable for transmission. A heterodyne method had been used previously by Dr. J. S. Mainstone to enable tape recordings of the initial whistle accompanying trail formation to be made for subsequent velocity measurement, but is unsatisfactory for the following reasons. Firstly, a very high order of stability is required not only in the receiver and local heterodyning oscillator, but also in the CW transmitter, if the heterodyne carrier frequency (4kc in the original) is to remain within the passband of the 4kc bandpass filter required to realize the full noise figure advantage of the method. This difficulty could have been overcome by the use of some form of automatic frequency control of the heterodyning oscillator to produce a constant carrier frequency. A second difficulty arises because the heterodyne method is phase coherent. The 27 mcs transmitted wave being phase modulated, severe ringing occurred even in the absence of an echo because of the steep sided bandpass characteristic of the original 4kc filter. This problem has no ready solution. A filter with a bandpass over 6kc wide was eventually used. This reduced the ringing but the echo was degraded by noise.

The chopper method suffers from a 3db increase in noise

over the heterodyne method, but since it is used after detection, the signal is no longer phase coherent in that only the genuine phase spikes, whistle and doppler resulting from the groundwave/skywave interaction are present; the 27 mcs receiver AM detector produces no output other than the D.C. level from the phase modulated groundwave alone. This means that the filter bandwidth can be kept down to 1kc (the phase spikes are normally of the order of 1 millisecond long), with obvious noise figure improvement.

The circuit of the chopper unit is reproduced in Fig. 24. Stability has proved excellent, being of the order of  $\pm 100$ cps at an output frequency of 5kc (after an initial 12 hour warmup), this being maintained at one stage over a period of 3 months continuous outstation operation. Switching transients generated in the diode ring modulator are 60db down with respect to the filtered signal output.

The output from the filter appears as an amplitude modulated carrier wave whose frequency is that of the multivibrator driving the switching diodes, and whose amplitude is directly proportional to the receiver detector output.

Different chopping frequencies are used at the two outstations, viz. 2kc at Sheedys' Farm and 5kc at Direk. The choice of suitable chopping frequencies involves consideration of several conflicting requirements. The whistle accomp-

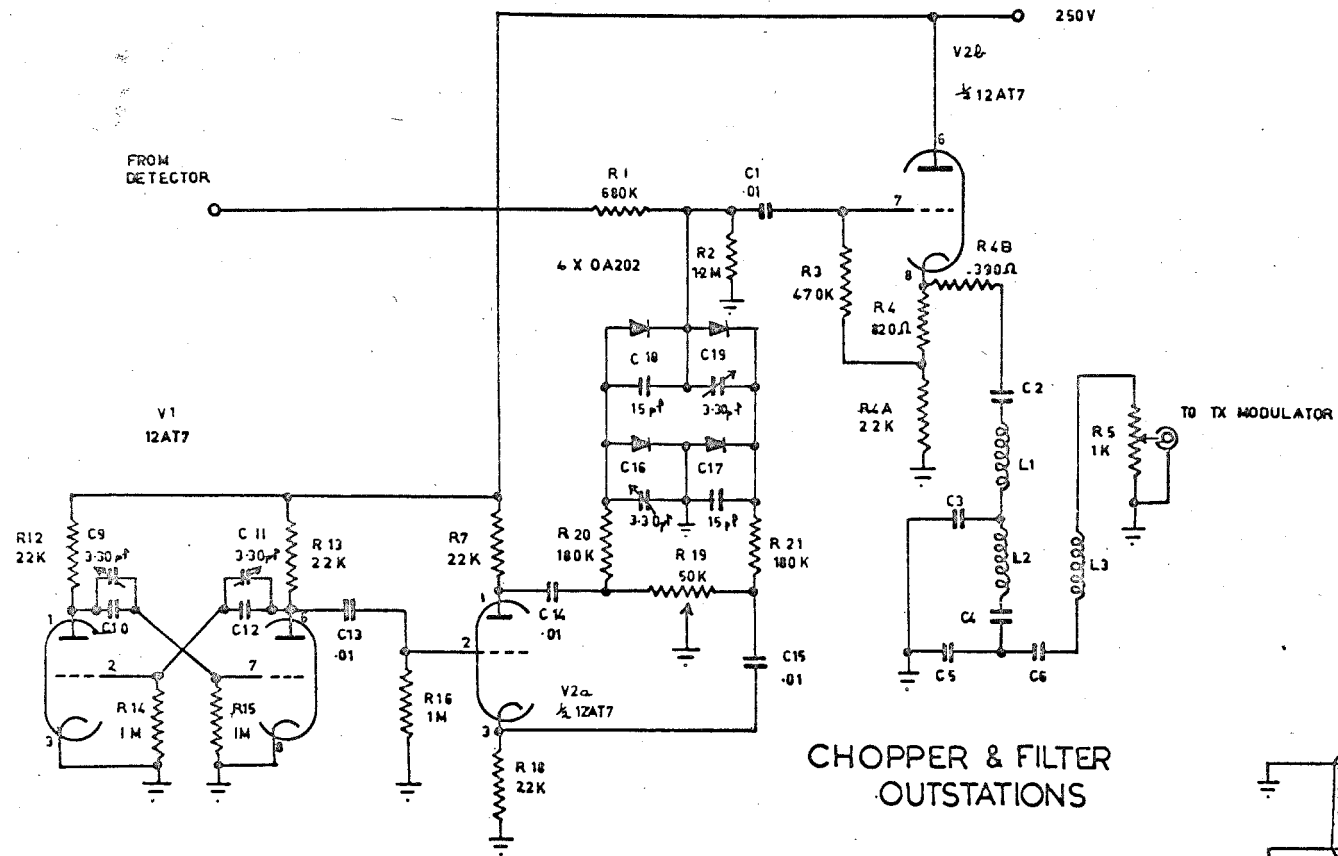


FIG 24

any trail formation contains frequencies as high as 600 cps for Fresnel zones well removed from the  $t_0$  point, and thus the sampling frequency should be several times this value. However, the meteor velocity recording technique involves the storage of information on magnetic tape, which imposes upper frequency limits at conventional tape speeds. During storage on tape all signals are mixed and subsequently separated; if sampling frequencies are harmonically related, interference between channels can occur. The frequencies chosen have proved to be a good compromise.

(c) The F.M. Transmitter.

A commercial frequency modulated crystal controlled 167 mcs transmitter (Philips type 1645/02b, now obsolete) is used to transmit the chopper/filter output back to the main station at St. Kilda. The original transmitter, designed for voice communication, was modified by removal of an input limiting arrangement which would have had a disastrous effect on the overall linearity of the system, and incorporation of overload protection devices, deemed necessary since the outstations run unattended.

(d) The Power Supplies.

In the interests of stability, the high tension to all units except the transmitter final amplifier is provided by an electronically regulated 250v supply, as used at the main



station and described in Chapter V. Negative bias to the receiver manual gain control is also regulated. The transmitter final stage is fed from an unregulated 400v supply; all that is required of this supply is that the output from the transmitter be kept well above the limiting level of the link receivers used at the main station.

Output from the transmitter is conveyed via coax cable inside the antenna pole to a 4 element Yagi array beamed on St. Kilda. The Yagi is mounted 3 feet above the 27 mcs receiving dipole.

### 8.3 Subsidiary Main Station Equipment.

The transmissions from each outstation are received at St. Kilda on 4 element Yagi arrays, and are amplified and discriminated by 13 valve FM receivers (Philips type 1645/01b, now obsolete. These receivers have been superseded by the manufacturers from considerations of size only. Their performance is quite good). Both outstations are transmitting simultaneously on the same frequency of 167.02 mcs, but the 40db front to side ratio of the receiving arrays ensures that each receiver limits on the required signal. Simultaneous transmissions would not be possible with amplitude modulated signals, because of interference problems. Link noise is 57db down on the information carried; this is due in part to the excellent performance of the bandpass filters connected

# RECORDING RACK BLOCK DIAGRAM

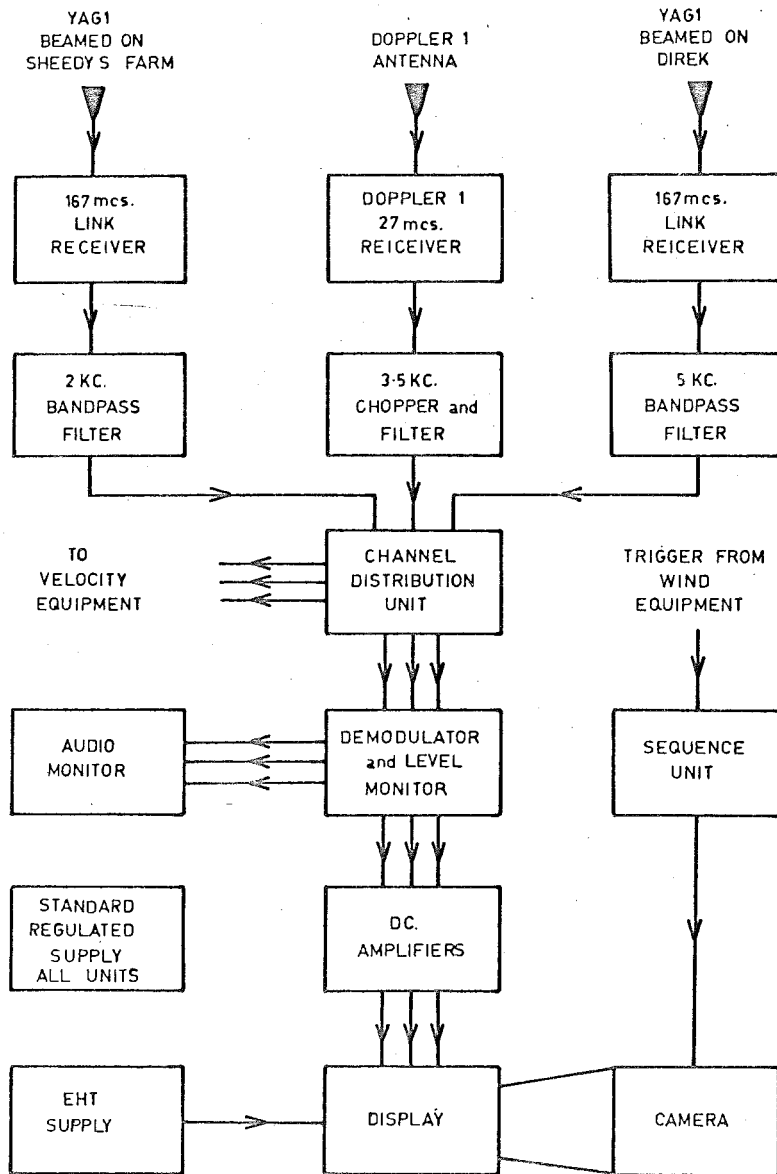


FIG. 25

to the receiver outputs. These filters are the same as those used at the outstations to filter the chopped receiver output. Through the use of Ferroxcube potcore inductances a good bandpass is achieved with only two constant k T-sections, and two m-derived end sections.

The signal to noise ratio at the 27mc outstation aerials averages 26db. Low receiver noise figure means that external noise completely governs even weak signal performance, and because the links introduce no measurable degradation, the signal to noise ratio of the filtered FM receiver output is the same as that at the outstation 27 mcs aerial.

The outputs of the link receivers are fed to the channel distribution unit in the Three-station Wind Recording Rack, Figs. 25 and 26. This unit also chops and filters the output from one of the main station doppler receivers. The three carriers thus derived are

2kc transmitted from Sheedys' Farm,

3.5kc output from one main station receiver,

and 5kc transmitted from Direk.

These are in a form suitable for use in the velocity measurement technique.

#### 8.4 The Multi-channel Tape Delay Unit.

Because the Wind Equipments' only record information received after the trail is formed, some form of memory circuit

is required to enable the initial whistle associated with the formation of each trail to be recorded.

This is provided by the Multi-channel Tape Delay Unit first designed by Mainstone (1960), but which has since undergone considerable modification and improvement. This was undertaken by C.S. Nilsson, who was also responsible for the reduction of the orbit data.

The original recorder incorporated three separate record and playback head pairs, four erase heads and a highly complicated switching mechanism to automatically effect tape reversal, and thus allow for continuous recording. However, because of the tension differential between heads, simultaneity of time between channels was lost; in addition wow was such that time marker pips on each channel would have done little to improve the situation.

Because of the inherent stability and flexibility of the chopper method of carrier derivation, an alternative recording procedure was devised. The three carriers are mixed (linearly added, so as not to beat together), applied to the one recording head, and picked off the tape approximately 1.5 seconds later by a single playback head. Considerable simplification of the reversal switching mechanism results, and, because of the reduction in the number of heads traversed by the tape, wow and flutter are reduced to negligible proport-

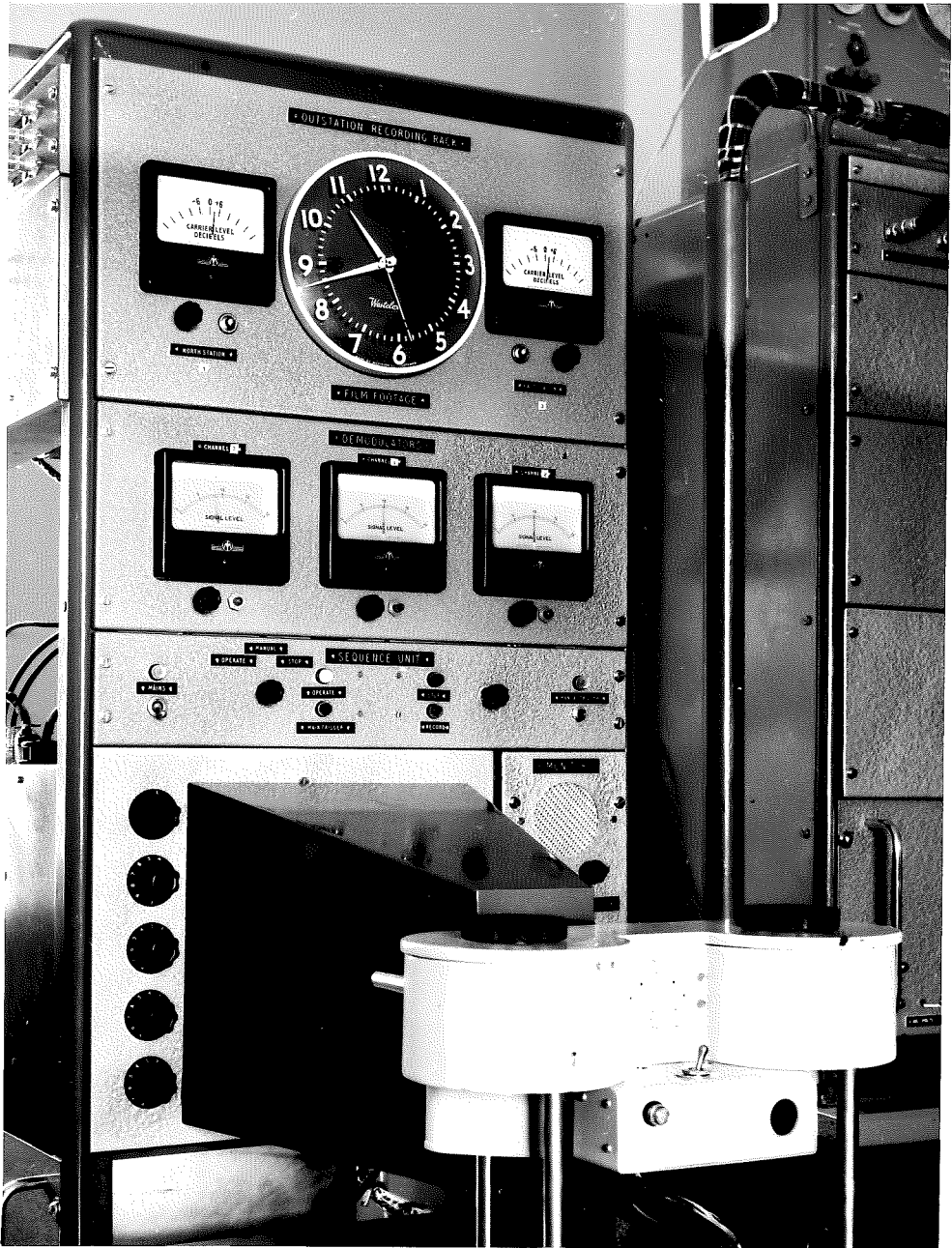


FIG 26-THREE STATION WIND RECORDER

ions. To unscramble the playback signal, three six section filters are used, each 1200 cycles wide and centred on one of the three carrier frequencies. The separate carriers are then amplified, demodulated and applied to three cathode ray tubes in the orbit display. If an echo triggers the main wind equipment, the orbit camera is then set in motion, and runs for a sufficient interval to record the initial echo information which has been delayed 1.5 secs by the tape.

Because the whistle amplitude is normally only a few per cent of the groundwave amplitude, the large body doppler after the whistle is attenuated by a frequency selective network (the doppler frequency is much less than any component of the whistle) so that the maxima and minima of the doppler are recorded without overloading the display amplifiers.

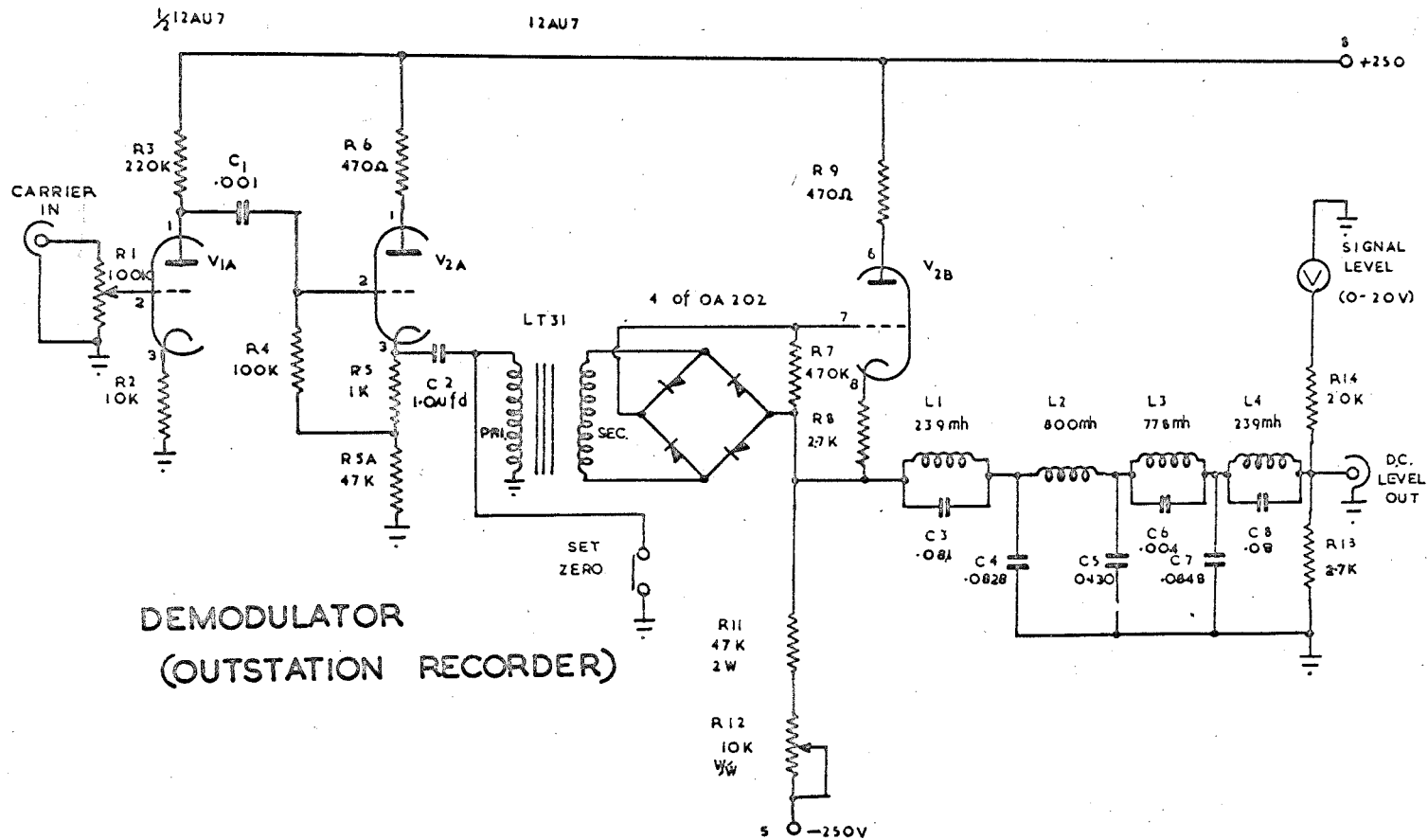
#### 8.5 The Three-station Wind Recorder.

The three outputs from the channel distribution unit are also applied to the Three-station Wind Display. The records obtained from this display make possible the measurement of reflection amplitudes and decay times (the measurement of diffusion coefficient differential with height is contemplated as an extension of the turbulence programme) and give a better overall picture of the post  $t_0$  echo than that available from the orbit display.

The two outstation carriers are amplified and demodulated

by two circuits of the type shown in Fig. 27. The potentiometer in the cathode of  $V_{2b}$  returns to a negative voltage supply; suitable adjustment of this potentiometer produces zero voltage at the output of the lowpass filter  $L_{1-4}C_{3-8}$  for zero carrier input. Presence of carrier input results in rectification of the signal by the diodes (4 X OA202) and application of a positive bias to the grid of  $V_{2b}$ . This causes an increase in current flow through the valve, and a positive voltage is developed across the filter load  $R_{13}$ . The DC voltage output is plotted against peak to peak carrier input in Fig. 28. The linearity in the 0 to 20v output region is excellent. The lowpass filter has a frequency cutoff of 1kc, which is sufficiently high to pass the phase spikes, which are of the order of 1 millisecond duration, but low enough to completely suppress the rectified carrier component. Once again, the use of potcore inductances has produced a filter of high quality. The filter characteristics are presented in Fig. 29.

The output from the main station receiver is applied directly to the appropriate DC amplifier in the cathode ray tube display, without being chopped, filtered, and then demodulated. It is significant that the signal to noise ratio of the film record obtained from this channel is the same as that of the outstation records.



DEMODULATOR  
(OUTSTATION RECORDER)

FIG 27



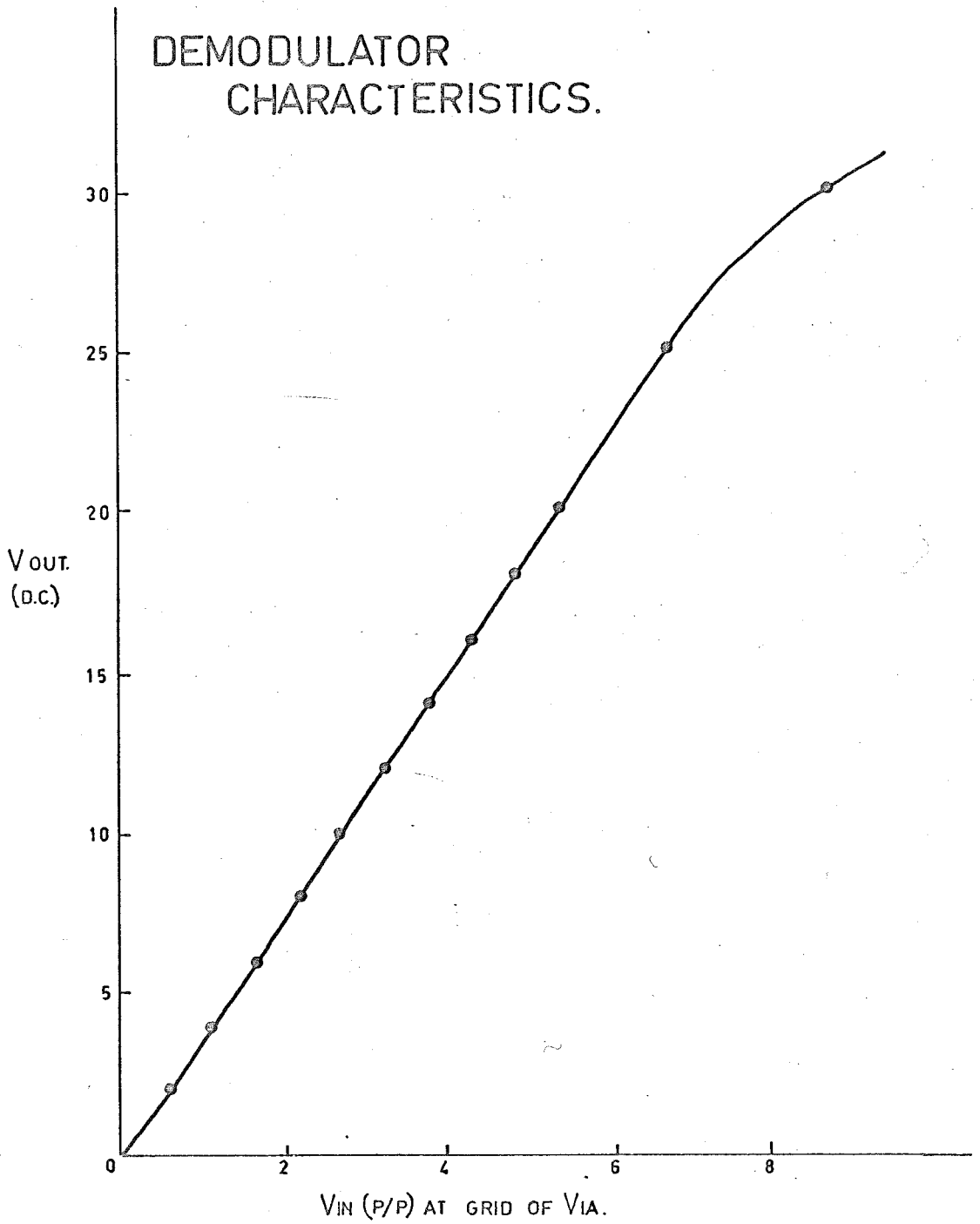
DEMODULATOR  
CHARACTERISTICS.

FIG. 28

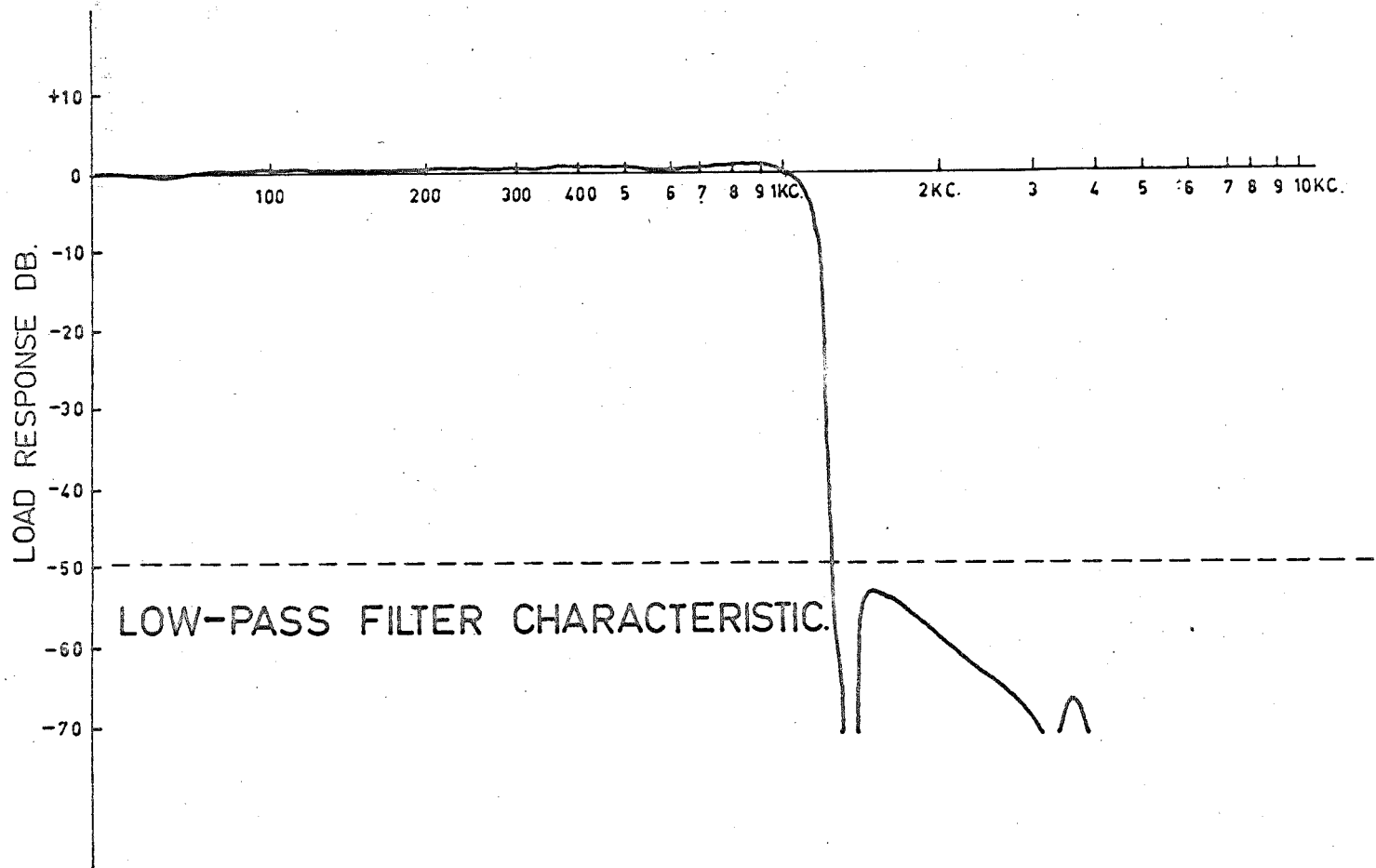


FIG. 29

For monitoring purposes, the signal levels representing the DC outputs of the three station receivers are displayed on three panel mounted meters.

The DC amplifiers in the display were designed for long term stability. A cathode follower drives an anode follower whose plate is connected directly to the appropriate Y plate in the cathode ray tube display. Three such amplifiers are used, two handling the outputs of the outstation carrier demodulators, and the third the output from the main station receiver. These signals are applied to two double beam Cossor 89J C.R.T's. The main station signal deflects the lower trace of the upper cathode ray tube in the display, and also the upper trace of the lower tube. The Sheedys' Farm signal is the very top trace, the Direk signal the very bottom trace (See Fig. 32b). These tubes are certainly not the best for long term photographic recording, but are adequate in this application.

Graphs of the signal level outputs as measured from the film records (normalized to a 10 volt groundwave level) against RF signal microvolts input at the outstation antennae are presented in Fig. 31. The non linearity in the overall characteristic is introduced in the frequency modulation of the 167mc transmitter and subsequent discrimination in the link receiver.

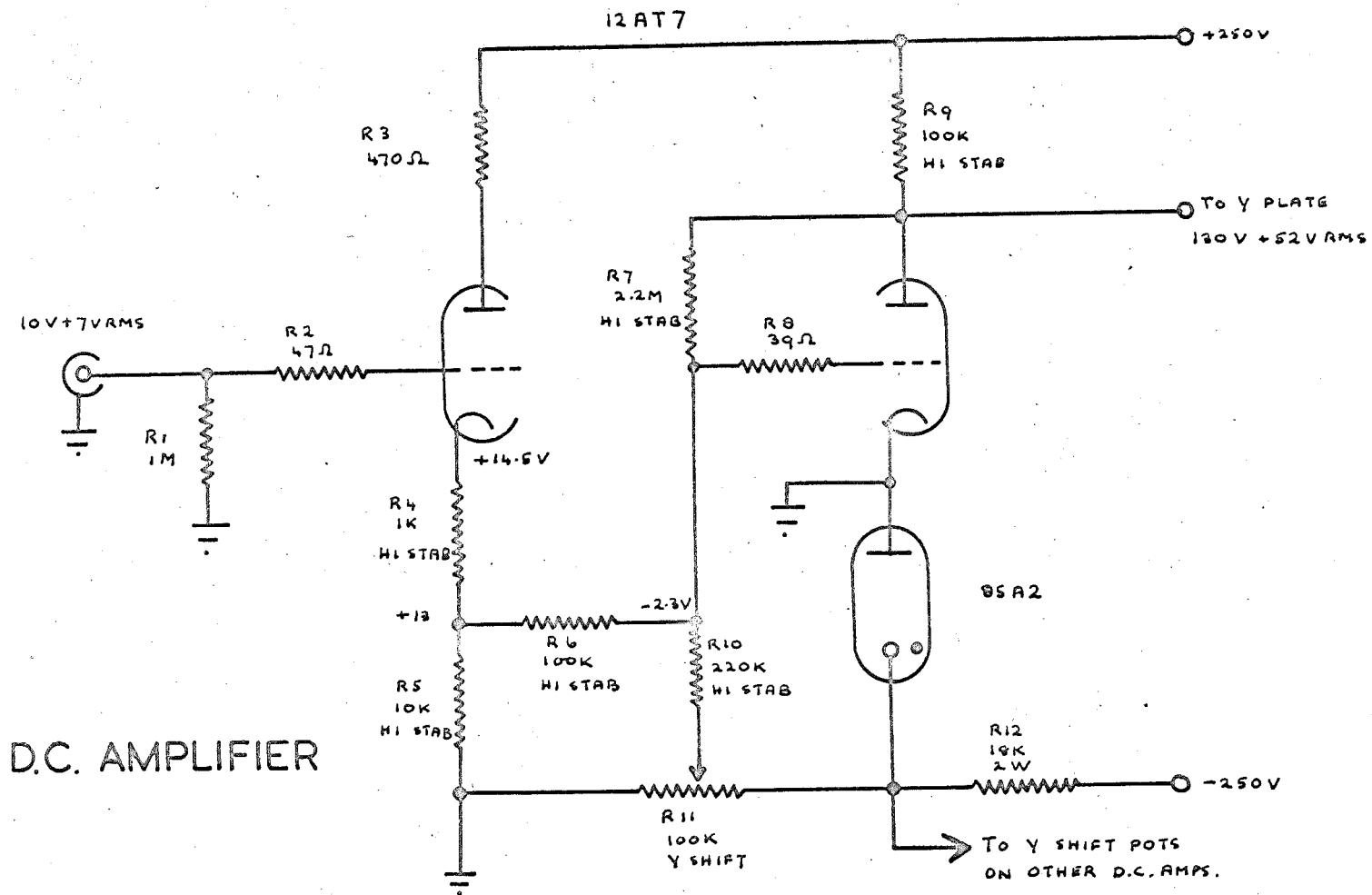


FIG 30

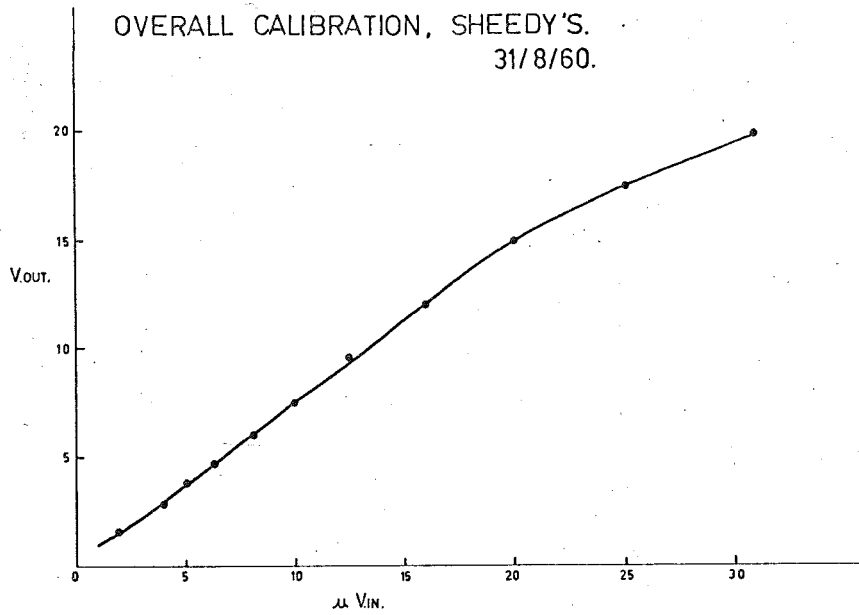


FIG. 31a

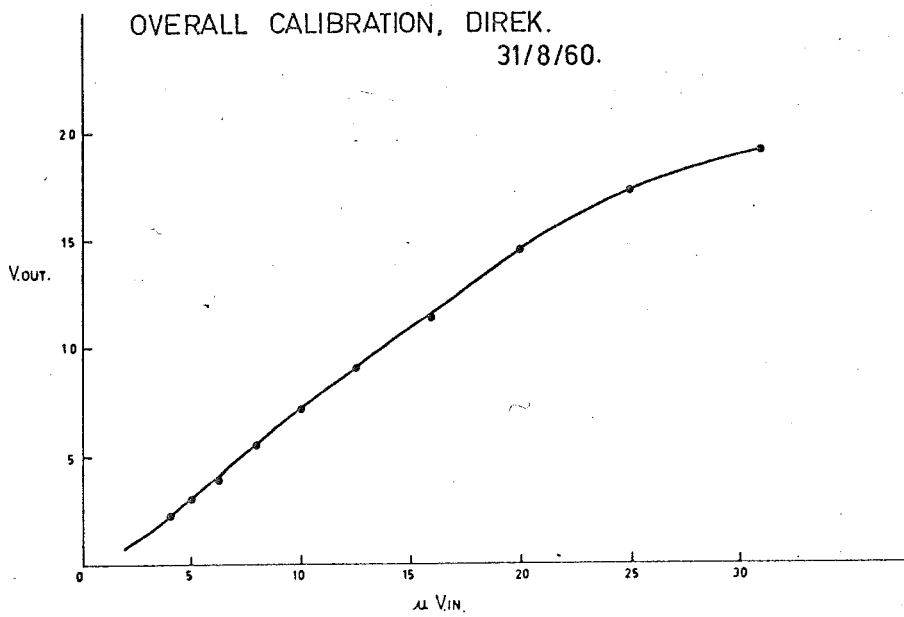


FIG. 31b

The recording camera is a Shackman, which has been modified by including a magnetic clutch between the motor and the film drive. The motor operates continuously and the film is brought into motion by applying a voltage to the clutch from the sequence unit. Gear ratios are such that the 35mm film travels at 0.38 inches/sec when the clutch is actuated. Thus, at the shortest recording duration of 1.4 secs, each echo occupies about half an inch of film. Film is daylight loaded in 100 foot lengths; the sporadic echo rate is such that one film normally lasts 36 hours.

For film correlation purposes, an electronic flash and a six digit counter are mounted in each of the three displays (orbit, mean wind, and three-station wind). The counter and flash are operated during each echo by trigger pulses from the mean wind recording rack sequence unit.

The magnetic clutch actuating the three-station camera is wired in parallel with the mean wind equipment clutch to minimise delay in the commencement of recording, but once the film is in motion control by the wind equipment is removed and the three-station sequence unit takes over. This unit contains several relays and an electronic timer, the duration of the run being set by a switch on the front panel. Recording intervals from 1.4 secs minimum to 30 seconds maximum can be selected. Most of the routine running is done with the

shortest duration in the interests of film economy, but, for observations of long duration echoes, times up to the maximum have been used occasionally. Manual triggering of the clutch and sequence unit is also available for testing purposes.

#### 8.6 The Film Records.

The films from the orbit display (a), three-station wind display (b) and mean wind display (c) are shown in Fig. 32 for echo number 54746. The whistle patterns of 32(a) show that the echo was first received at the Main Station (middle trace), then at the Direk Outstation (bottom trace) and finally at Sheedys' Farm (top trace). The time scale is indicated by the phase spikes which are quite evident on the slow body doppler at the end of each echo. These spikes are 20 milliseconds apart.

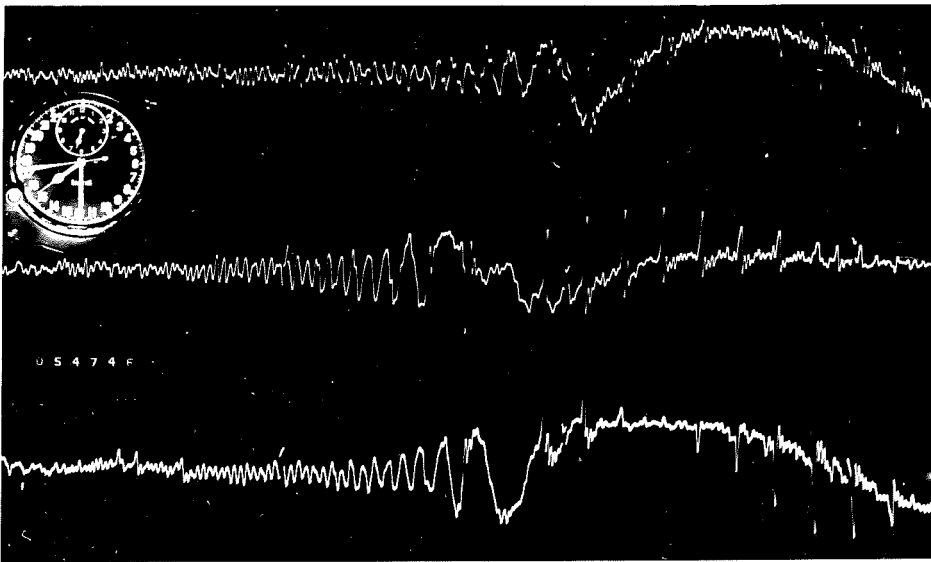
The record of 32(b) shows the presence of wind shear. The initial whistle does not appear, since it precedes the main echo which triggers the camera. Taking the traces in the order of reflection point occurrence as determined from 32(a), the line of sight drifts are

27 metres/sec at the main station (centre two mirror image traces)

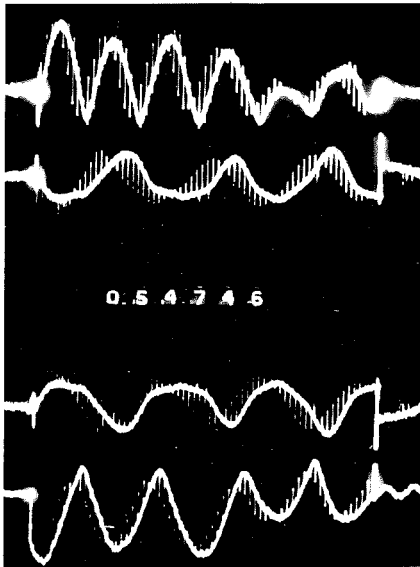
40 metres/sec at Direk (bottom trace)

and 55 metres/sec at Sheedys' (top trace).

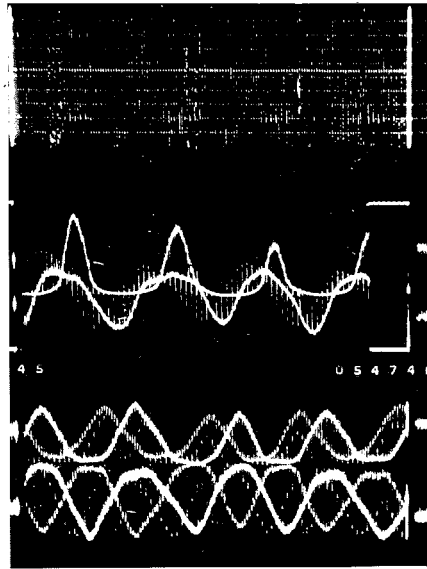
The first reflection point always occurs at the greatest



a ORBIT



b 3-STATION WIND



c MEAN WIND

FIG 32



height; thus this echo shows the presence of a negative wind gradient with height.

Before the meteor velocity and separation distances can be determined from 32(a), a knowledge of the range and direction cosines of at least one reflection point are required. This information is furnished for the main station echo by the film 32(c). The upper trace is a conventional radar scan, the equispaced dotted horizontal lines being 20km range markers. The range at 108km is clearly visible. The next two traces are the outputs of the two doppler receivers. Unfortunately the magnitude of the echo was such that receiver overload occurred. However, the doppler minima are still well defined; because of the nature of the beat waveform ( $\cos^2 t$ ), the minimal cusps are always used for measurement. The spikes on the trace caused by the phase modulation of the transmitter are clearly visible. Since the wave traced out by these spikes leads the doppler beat, the trail drift is towards the observer.

Local time is recorded by the clock in the orbit display. Films are correlated by means of the electronically flashed counters.

CHAPTER IX.The Methods of Data Reduction  
as Applied to the Results For December 19609.1 The Mechanics of the Turbulence Data Reduction.

At the commencement of the orbits and turbulence projects in 1958, it was appreciated that considerable time would be involved in data reduction, since any worthwhile orbit survey and turbulence investigation would necessitate recording for at least twelve months. The design and construction of the completely new St. Kilda receiving station and the two outstations took until mid July, 1960, and the whole system was operating reasonably satisfactorily by December of that year. For the first few months, until the end of April, 1961, equipment reliability was not good, and the rate of echoes suitable for reduction suffered. However, for the rest of the survey, which terminated at the end of December, 1961, very little equipment maintenance was necessary, and reduction of the data already obtained was commenced. This proved to be a most tedious process, six months of graphing, curve fitting and desk machine calculation being required to produce the final results for only one month, viz. December,

1960. Fortunately, early in 1961 the Weapons Research Establishment at Salisbury installed and made available to University users an IBM 7090 digital computer to ease the load on their digital/analogue computer WREDAC. From the research viewpoint, the 7090 is one of the best computers in the world. Not only is it quite fast in operation, but it is relatively easily programmed via the IBM FORTRAN language; programmes are written in an elementary algebraic form, which is translated by the machine itself into the highly complicated binary coded machine language.

Programmes to expedite some aspects of the orbit reduction were first run by C. S. Nilsson early in 1961, but the complete orbit reduction programme was not finalized until the beginning of this year (1962). The first turbulence programmes were run in January, 1962, and underwent considerable addition and modification until the installation, in April, 1962, of an IBM 1620 computer in the University's Computing Centre. Although considerably slower in operation than the 7090, and having a very much smaller storage capacity, the 1620 has the most decided advantages of personal operation by the user, 24 hour a day availability, and no charge for research use. Because of programme complexity and large storage requirements, the orbit reductions were carried out on the 7090 at W.R.E., and the information for each echo relevant

to the turbulence programme was punched onto cards by this machine. All subsequent reduction was then performed on the University's 1620. The FORTRAN programmes appropriate to each portion of the data processing appear in Appendix II. No attempt is made to translate these programmes directly into conventional mathematics, since the method is adequately detailed in subsequent sections of this Chapter. The lists given of variables used in the programmes are not complete in that many of the variables used within the programmes are concerned only with machine manipulation, and the lists are therefore confined mainly to quantities used as input and output.

## 9.2 The Formulation of the Turbulent Velocity Field.

Before any application of the theories of homogeneous turbulence can be attempted, all mean velocities and mean gradients must be eliminated from the total flow field. As detailed in Chapter V, the Adelaide method of mean wind measurement involves the hourly grouping of wind velocities measured over several days. This represents a time average over an interval which is certainly long compared with the time constant of 100 minutes for the energy bearing eddies determined by Greenhow and Neufeld. The annual repetition of the pattern for a typical day of a given month lends weight to the assumption that the motion as determined is, in fact, a

true mean.

The first assumption made in the preliminary data reduction as applied to the turbulent motions is that these are isotropic. Although this radical assumption is not true (other workers have clearly demonstrated the anisotropy of turbulent motion at these heights), it simplifies the initial approach, and enables an eventual estimate of the possible vertical component, and thus a clue to the degree of anisotropy.

The data available from each record for turbulence measurement consists of

- a) the line of sight motion at three points on the meteor trail,
  - b) the spacial separations of these three points,
  - c) the line of sight direction cosines and the height above ground of one of these three points,
  - d) the direction cosines of the meteor trail,
- and e) the time of occurrence of the echo.

Accurate determination of the line of sight drift requires measurement of at least two cycles of the body-doppler beat immediately after trail formation. This corresponds to a line of sight drift of one wavelength at the transmitted frequency of 27 mcs; i.e. 11.2 metres. Even if another of the measured points on the trail was subject to zero wind

drift, the deformation of the trail during the interval of measurement is small since reflection point separations are of the order of hundreds of metres. Thus the trail may be assumed linear.

The sampling method involves the measurement of velocities perpendicular to the separation parameter, and is thus of the form specified for the transverse correlation of Chapter II, Section 4.

To perform the mean wind extraction, the NS and EW mean wind components for each of the three height groups centred about 80, 90 and 100km are tabulated hour by hour. A linear interpolation is used to determine the mean gradient for each hour above and below the 90km level. All subsequent reduction utilizes the velocity components at the 90km level, and the appropriate gradient, depending on whether the reflection point lies above or below this height. The height of the echo received at the main station is computed directly from the wind records (see Chapter V). The height of each outstation echo is calculated relative to this height, from the direction cosines of the trail itself, and the separation of the reflection points. The information for this purpose is obtained from the orbit data. Whereas the main station reflection point determination is subject to errors of the order of  $\pm 3$ km, the height differentials are measured to  $\pm 20$

metres. The mean wind velocity appropriate to the mean solar time of the echo is then calculated for each of the three reflection points. The line of sight components of the mean velocities are then subtracted from the line of sight drifts as measured from the Three-station Recorder film to give the velocities of the turbulent flow field. The RMS value of the turbulent velocities for all echoes in a given month is taken as the velocity characteristic of the energy bearing eddies.

### 9.3 Turbulent Velocities as a Function of Separation.

The two important quantities which can be related to separation are the turbulent velocities and velocity differences.

The transverse correlation function given in 2.4.1 is

$$g(\xi) = \frac{\overline{u(x)u(x+\xi)}}{[\overline{u^2(x)} \overline{u^2(x+\xi)}]^{1/2}}$$

This can be redefined as

$$g(\xi) = \frac{\sum u(x)u(x+\xi)}{[\sum u^2(x) \sum u^2(x+\xi)]^{1/2}}$$

since, in forming the means for a given  $\xi$ , the number of echoes of that separation is common to both numerator and denominator. In practice the summations are made over discreet intervals of  $\xi$  in 200 metre steps, commencing at a separation of 100 metres; all reflection point velocities with separat-

ions  $\xi$  less than 100 metres fall in the first group, all with separations of more than 100 metres but less than 300 metres in the second and so on.

The RMS turbulent velocity differences for a given separation are found for similar intervals from

$$\left[ \frac{\sum (u_i - u_j)^2}{N} \right]^{1/2}$$

where  $u_i, u_j$  are the line of sight turbulent velocities measured at reflection points  $i$  and  $j$  of separation  $\xi$ , and  $N$  is the number of observations with this value of  $\xi$ .

To illustrate the method, the results for December, 1960, the first month of the survey, are presented in detail. The transverse correlation has been graphed (Fig. 33) according to the relation

$$1 - R(\xi) \sim \xi^m \quad (\text{equation 2.11.5})$$

A line of slope  $\frac{2}{3}$  is superimposed.

Very little significance can be attached to the variation of the velocity correlation function with separation. The scatter of Fig. 33 can be explained in terms of the distinct anisotropy of the energy bearing eddies, the time dependent nature of the turbulent flow, and the perturbation of the small scale spectrum by the eddies of the buoyancy sub-



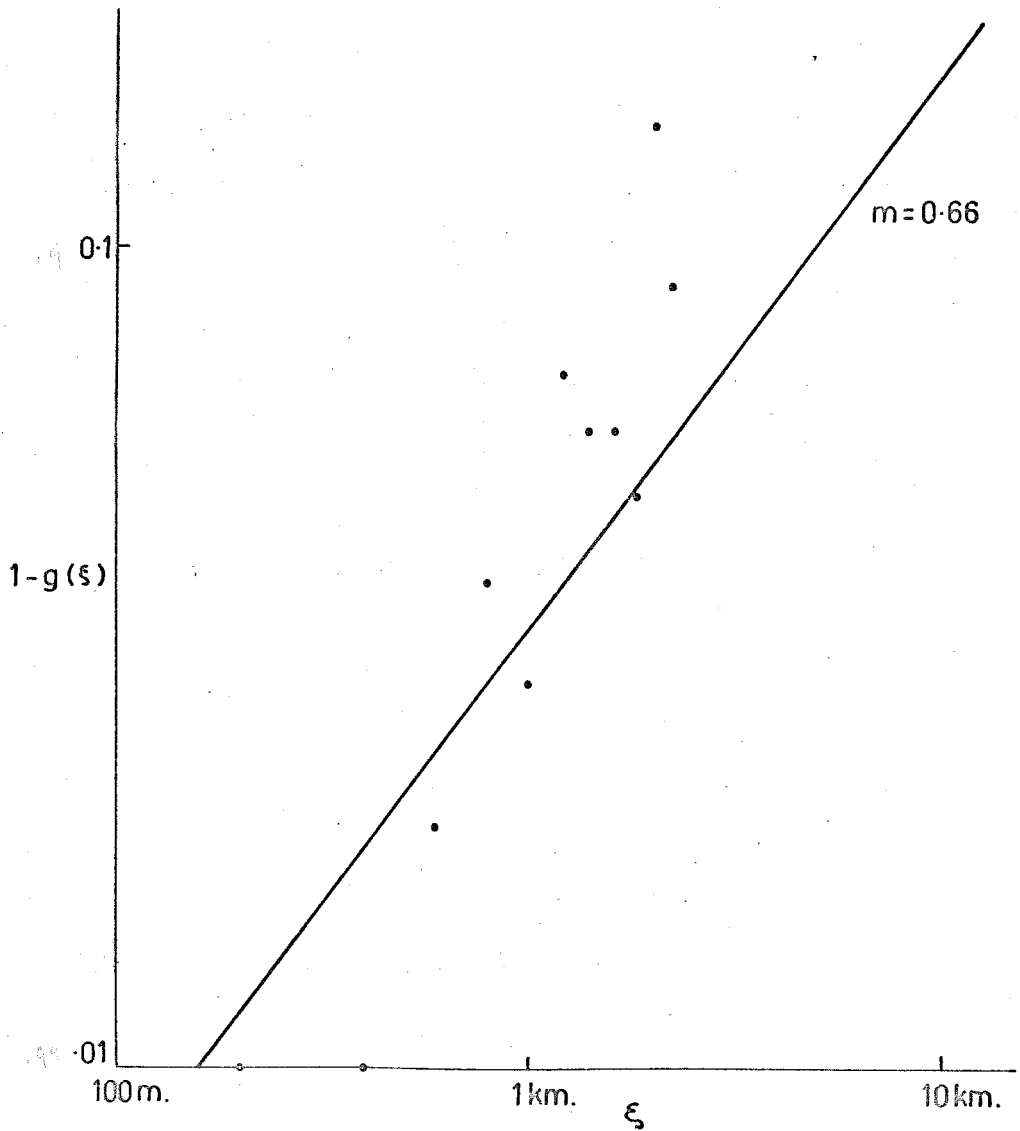


FIG. 33 - The transverse velocity correlation,  
December 1960

range. These factors are considered in detail later in this chapter.

The velocity differences of Fig. 34, however, do prove to be of some consequence. To relate the RMS velocity differences to the Kolmogoroff spectrum, use is made of equation 2.10.3, viz.

$$\overline{\Delta v} = 0.83(\epsilon \xi)^{\frac{1}{3}}$$

where  $\epsilon$  is the rate of dissipation of the turbulent energy.

Various values of  $\epsilon$  were substituted until the  $\Delta v$  curve fitting most of the measured velocity differences was derived. During this curve fitting process, it became quite obvious that there existed a distinct anomaly at the smaller scales, in particular those less than 500 metres. Whereas the overall best fit was obtained for a value of  $\epsilon$  of approximately 300 ergs/gm/sec., the smaller scales were better represented by a rate of dissipation of energy of less than 100 ergs/gm/sec. This is of considerable interest in that optical measurements of the expansion of sodium and meteor trails, which are limited by the visibility of the trail to scales less than 500 metres, have produced  $\epsilon$  values of some 70 ergs/gm/sec (see Chapter III Section 6).

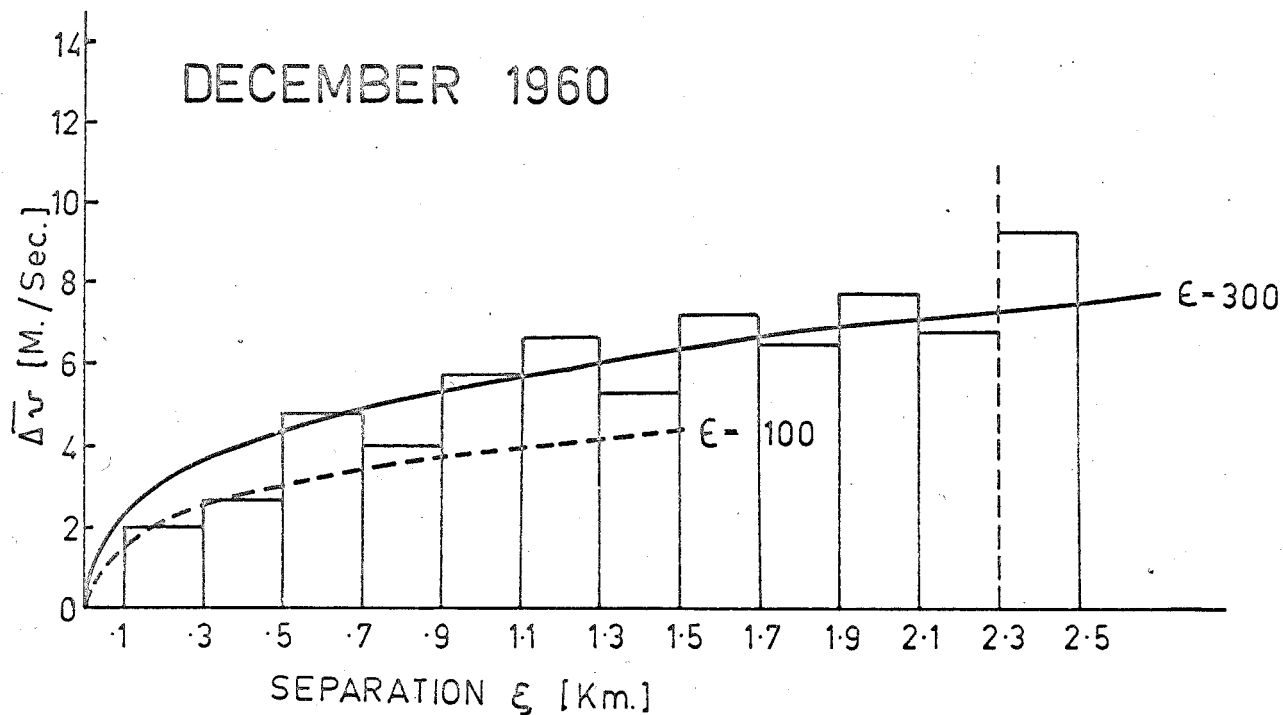


FIG.34-Variation of velocity difference with separation.

#### 9.4 The Anomalous Velocity Differences at Small Separation.

One of the axioms of the Kolmogoroff theory is that all scales of the inertial subrange are characterized by the same value for the turbulent dissipation parameter  $\epsilon$ . The breakdown of 9.3 is explicable in terms of the Kolmogoroff theory only if the scales less than 500 metres are no longer inertial, but comprise the region of viscous dissipation. The size of the eddy characteristic of the viscous dissipation subrange is given in 2.6.4 as

$$\eta = \left( \frac{\nu^3}{\epsilon} \right)^{\frac{1}{4}}$$

where  $\nu$ , the kinematic viscosity, is of the order of  $10^5$  cms /sec at 95km.

For  $\epsilon = 300$  ergs/gm/sec.,

$$\eta = 13 \text{ metres.}$$

Since the characteristics of the inertial subrange are expected to hold down to some five times this scale (vide 2.6.6) the onset of viscous dissipation should not be apparent until separations less than 100 metres are reached. It would appear that some other source of energy extraction is required to account for the anomaly.

Examination of Fig. 34 in greater detail brings out some other features. While the anomaly in the velocity differences is most marked below 500 metres, there does appear to

be an overlying periodic variation at larger separations, with period between minima in the velocity differences of approximately 700 metres. As has been previously mentioned in Chapter III Section 3, the presence of a buoyancy subrange can extract energy from all scales of the turbulent flow field. These buoyancy effects become most important when the buoyancy potential energy per unit mass is of the order of the turbulent translatory kinetic energy per unit mass.

The buoyancy potential energy per unit mass is given by

$$P_{\epsilon} = L_{\beta}^2 \frac{g}{\Theta} \frac{d\Theta}{dh}$$

where  $L_{\beta}$  is the eddy size characteristic of the buoyancy (or Richardson type) instabilities, and  $\Theta$  is the potential temperature of the gas.

From Chapter II Section 12

$$\frac{1}{\Theta} \frac{d\Theta}{dh} = \frac{1}{T} \left[ \frac{dT}{dh} + \Gamma \right]$$

At a height of 95km

$$g = 950 \text{ cm/sec/sec}$$

$$T = 180^{\circ}\text{A}$$

$$\Gamma \approx 10^{-4}$$

and  $\frac{dT}{dh} \approx 2.5^{\circ}\text{A/km}$

$$\therefore P_{\epsilon} = 0.6 \times 10^{-3} L_{\beta}^2$$

By equating this to the total turbulent translatory kinetic energy per unit mass,

$$\frac{1}{2}U_0^2 = 0.6 \times 10^{-3} L_{\beta}^2$$

where  $U_0$  is the velocity characteristic of the energy bearing eddies.

i.e.  $L_{\beta} \approx 25U_0$

9.4.1

For the December 1960 results,

$$U_0 = 25 \text{ metres/sec}$$

$$L_{\beta} = 625 \text{ metres.}$$

In all probability, the motion observed at scales less than 600 metres is not that of the small scale regions of a Kolmogoroff spectrum but is due to the vertical displacement of gas brought about by the turbulent shears. Since this motion has a preferred direction, and involves potential as well as kinetic energy, the mode of energy dissipation would be expected to be modified, and not to proceed as for the Kolmogoroff spectrum.

Although the possible presence of these buoyancy scales is considered to provide the best explanation of the small scale anomaly, two other factors inherent in the meteor method of turbulent velocity measurement should not be overlooked.

The first factor is involved in the idealized notion of discreet reflection points. In actual fact theory shows that the majority of the radio frequency energy reflected from the trail comes from the first half Fresnel zone about the specular reflection point. The length of this half zone is given by

$$l = \frac{1}{2} \sqrt{R_0 \lambda} \quad (\text{Kaiser})$$

where  $R_0$  is the midpoint range of the echo, and  $\lambda$  is the radio frequency wavelength of the transmitted energy. For an average range of 150km, and a  $\lambda$  of 11.2 metres,

$$l = 700 \text{ metres.}$$

If the trail remains straight in terms of line of sight distortions of less than  $\lambda/4$ , the relative motions of points separated by 700 metres will not be resolved. However, once distortion exceeds this rather small figure, lengths of trail of only a few wavelengths will give rise to echoes of amplitude very little less than that from the whole half zone. This effect has been demonstrated most conclusively by Greenhow and Neufeld (1959b) using a precision radar with a resolution of the order of better than 10 metres at a range of 200km. The effect places a lower limit on "discreet" echo separation of some 50 - 60 metres.

The other factor, that of reflection point motion along the trail under the influence of wind shears, has been treated

in detail in Chapter VI Section 3. This effect has been shown to place a lower limit of some 200 metres on the measurement of reflection point separations.

Since both these limits are associated with scales much less than those affected by the observed anomaly, the presence of the buoyancy subrange with a scale of a few hundred metres seems reasonably well substantiated.

### 9.5 The Large Scale Irregularities.

In any attempt to extend the turbulent flow field to scales outside the range of measurement, considerable caution must be exercised. However, the good agreement obtained between theory and experiment over the 1km to 2.5km region, coupled with the experience of meteorologists obtained at lower heights, does inspire some confidence in an extrapolation to the size of the energy bearing eddies.

The relation between the scales of the inertial subrange is given in terms of the turbulent dissipation energy  $\epsilon$  by

$$\epsilon = A \frac{U^3}{L} \quad 9.5.1$$

(Batchelor, 1953)

where  $U$  is the velocity of the eddies of scale  $L$ .  $A$  is a constant (of order unity) whose value depends on the definition of  $L$ .

If the scale of an eddy is defined as the distance over which the velocity of that eddy falls from its characteristic



value to zero, the velocity difference relation of 2.10.3 (which has been shown to apply to measurements made in the meteor region at separations up to 2.5km) may be used to determine  $L$ .

Rearrangement of 2.10.3 yields

$$\epsilon = \left[ \frac{1}{5.7} \right] \frac{U^3}{L} \quad 9.5.2$$

with  $L$  replacing the separation length  $\xi$ .

From the December 1960 results, the velocity of the energy bearing eddies was 25m/sec, and the turbulent dissipation rate 300 ergs/gm/sec. Substitution in 9.5.2 gives the scale of the energy bearing eddies as

$$= 95\text{km.}$$

This is comparable with the scales of 60 to 250km for the energy bearing eddies as measured from their time correlation by Greenhow and Neufeld (see Chapter III Section 5).

For eddies of velocity 25 m/sec and scale 150km, as measured at Jodrell Bank, 9.5.2 gives a turbulent dissipation of 200 ergs/gm/sec, which is of the same order as the Adelaide results.

The reason that the expression 9.5.2 seems to be more consistent than 9.5.1 with  $A$  unity, which latter is well substantiated in measurements of laboratory turbulence, probably

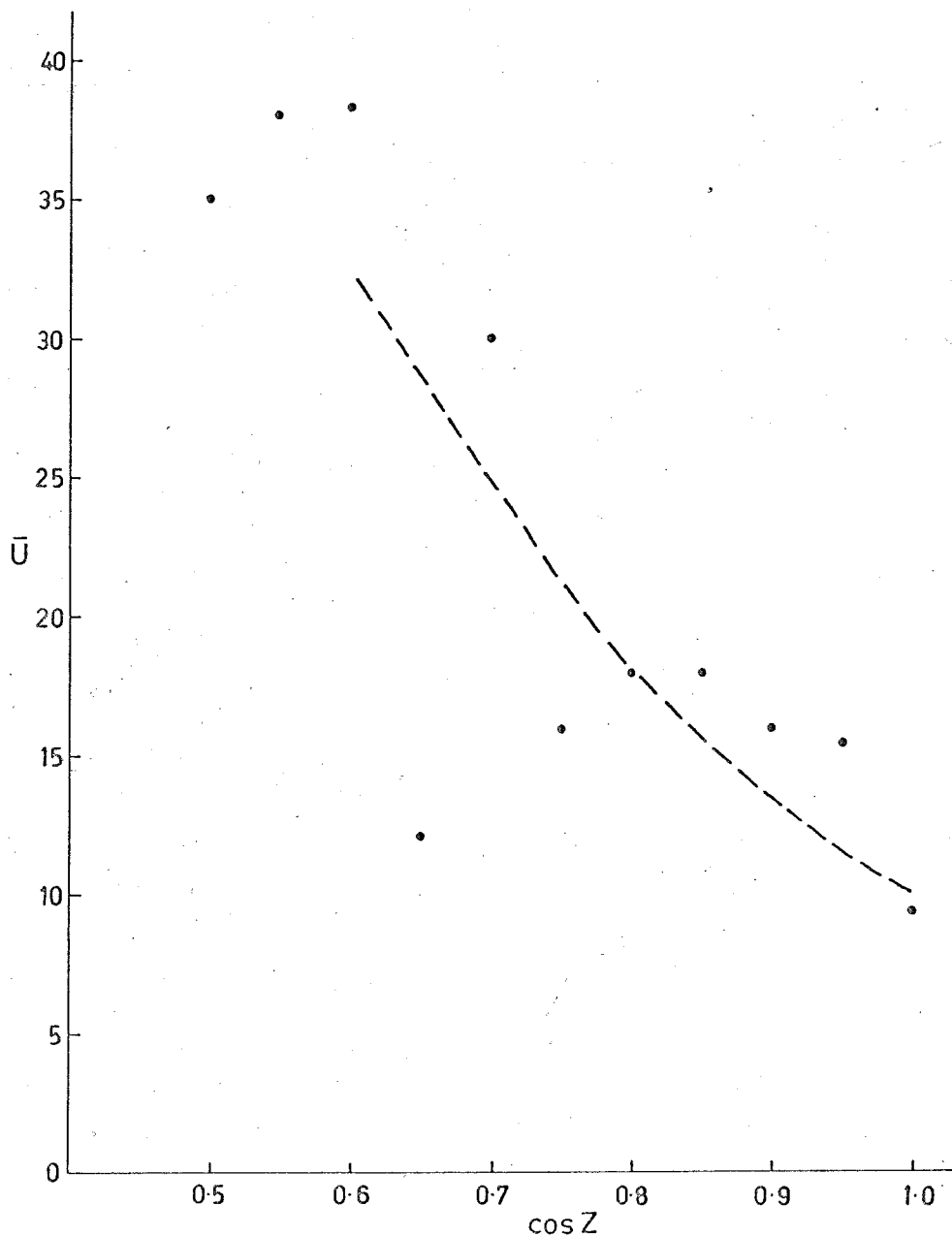
arises from the anisotropy of the turbulent flow field under investigation and the redefinition of the characteristic scale involved in its application. For the purposes of this thesis the relation 9.5.2 is used as an empirical tool; further work on its derivation is contemplated at a later date.

#### 9.6 Further on the Degree of Anisotropy.

Three values of line of sight turbulent velocity are obtained from each meteor trail observed. These values of turbulent velocity have been grouped for all echoes in the month according to the zenith angle  $Z$  of the main station reflection point; because the separation of reflection points is less than 3km at an average range of 150km, there is only a small variation in the zenith angles of arrival of the signals associated with each echo as received at each of the three stations.

The RMS turbulent velocities for intervals of  $\cos Z$  of 0.05 are plotted in Fig. 35. The velocities decrease with decreasing zenith angle, until, for echoes nearly overhead, a value of approximately 10 m/sec is observed.

If the height correlation interval of 6.5KM observed by other workers in this field is taken as the characteristic eddy length in the vertical direction in the equation 9.5.2, then, for a turbulent dissipation of 300 ergs/gm/sec., the



Variation of RMS Turbulent Velocity with  
Zenith Angle.

FIG.35

characteristic velocity of such an eddy is 10 m/sec. If 9.5.2 is considered, as originally derived, as the velocity difference/separation relation for an isotropic, inertial eddy subrange, the scales to 6.5km could belong to such a subrange.

The characteristic velocity of the turbulent flow field (considered as isotropic) has been determined as 25 m/sec; the RMS vertical velocity is 10 m/sec; thus, regardless of whether or not a Kolmogoroff spectrum is assumed, anisotropy of the large eddies with the maximum dimension horizontal is certainly well established.

#### 9.7 The Variation of the Velocity Difference with Height Difference.

To investigate the relation between turbulent velocity difference and height difference, the analysis for spacial separation was repeated, using height differences as the length parameter, regardless of the actual spacial separation. To obtain a measure of the variation of the spectrum function  $E(\Delta h)$ , the logarithm of the squares of the velocity differences has been plotted against the logarithm of the height differences  $\Delta h$ . The result is shown in Fig. 36. The best fit line of slope  $4/3$  is also plotted. While there is considerable scatter, agreement is generally good. The height differentials would appear to be indicative of a tur-

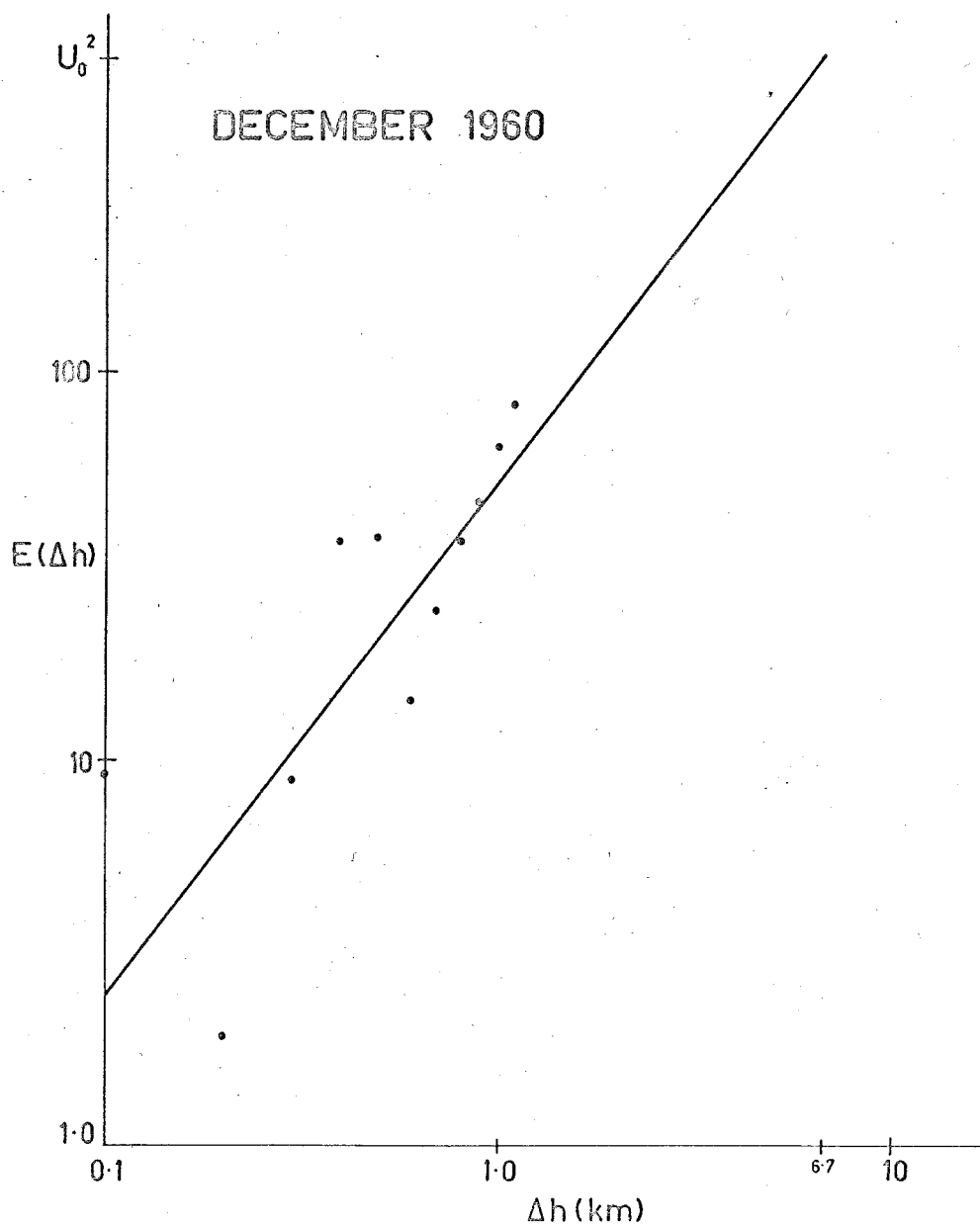


FIG. 36

bulence subjected to shear.

Extrapolation of the best fit line to a value of 625 (corresponding to the characteristic turbulent velocity  $U_0$  of 25 m/sec.) yields a scale of 6.7km. Since the spacial separations follow the isotropic Kolmogoroff spectrum, the  $4/3$  power law for the height differentials pertains to a region of small shear, which, if the extrapolation to the characteristic velocity is valid, should hold up to 6km.

#### 9.8 The Variation with Time of the Characteristic Velocity of the Energy Bearing Eddies.

To determine the variation of the magnitude of the characteristic velocity of the turbulent flow field throughout a typical day, the turbulent velocities for each echo are subjected to an analysis similar to that used to derive the mean wind. There is, however, the added complication of the vertical motion which must be considered in any attempt to resolve the turbulent velocities in the NS and EW directions. This problem does not arise in the mean wind analysis since mean vertical motion is very small relative to the horizontal velocity.

The effect of the vertical motion (taken as 10 m/sec) was removed by subtracting from the magnitude of each turbulent velocity measurement the component of the vertical motion

in the line of sight direction. The resultant was assumed to be the line of sight component of a horizontal turbulent motion. The radical assumption is now made that the azimuth of the line of sight is as likely to be the true direction of horizontal turbulent motion as any other direction. Provided that a sufficient number of echoes randomly disposed in azimuth is observed over a sufficiently long period of time, the RMS values of the turbulent velocity components in the NS and EW directions determined by the foregoing method will be a good approximation to the true values.

The hour by hour turbulent velocity components for December, 1960, are presented in Fig. 37, together with the mean wind for the height range 85 - 94 km. The curves are the result of a Fourier analysis truncated at the second harmonic, i.e. mean plus 24 hour and 12 hour components.

The RMS turbulent velocity has its maximum value in the early morning, and is a minimum during the late evening. If the turbulent and mean wind velocities are compared on the basis of magnitude alone, there is no correlation, as was found by Greenhow (see Chapter 3 Section 4). If, however, relative phases are considered also, then some dependence is evident. The maximum turbulent velocity is associated with the change in mean wind from meridional to zonal, and the minimum turbulence with a mean wind change from zonal to

DECEMBER 1960

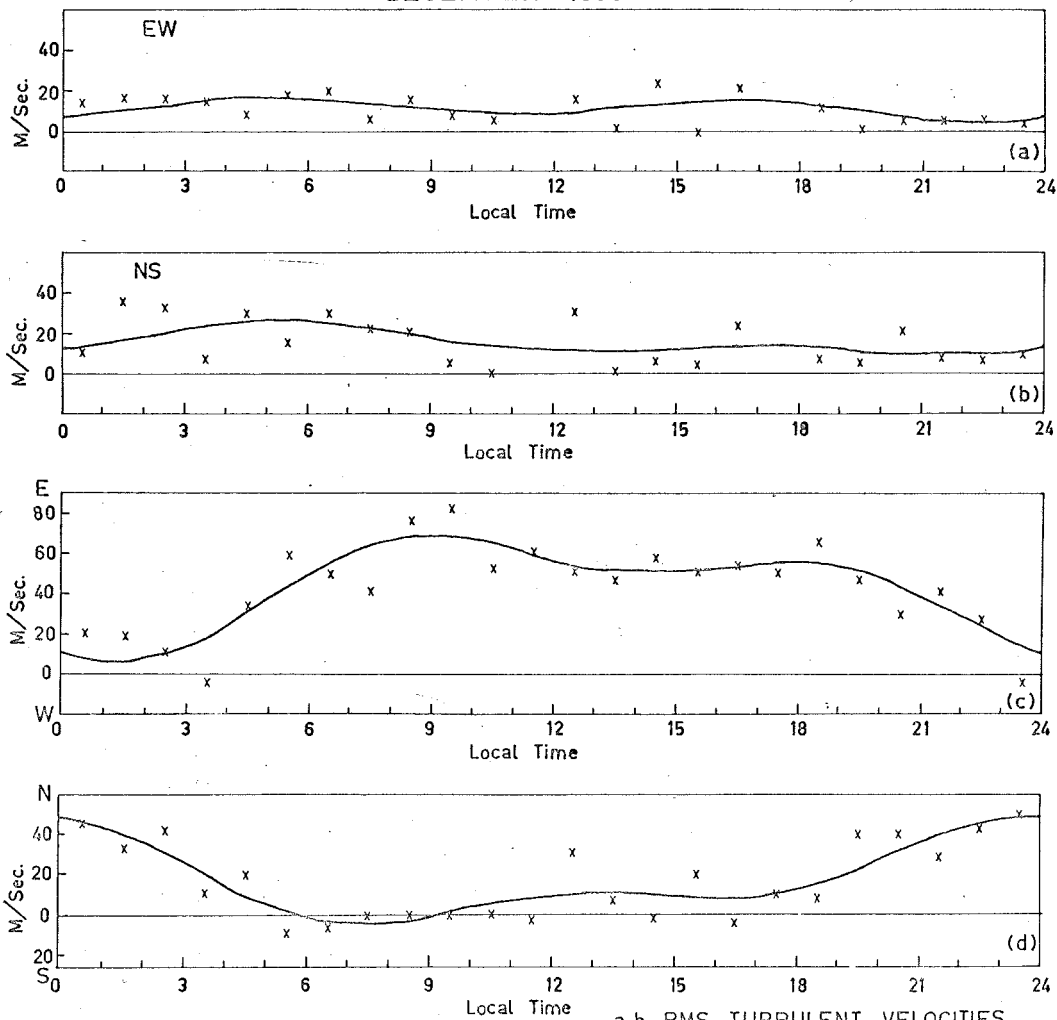


FIG. 37 a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94 KM.



meridional.

Further comments are made on the time dependent nature of the turbulent flow field in Chapter XI, after consideration of the 1961 results for January to December inclusive.

### 9.9 The Characteristic Dissipation Length $\lambda$ .

The characteristic dissipation length  $\lambda$  can be estimated by solving the equation

$$\epsilon = - \frac{15\nu U_0^2}{\lambda^2}. \quad (2.7.1)$$

For  $\epsilon = 300$  ergs/gm/sec

$U_0 = 25$  metres/sec

and  $\nu = 10^5$  sq. cm/sec

$\lambda = 11.7$  km

A similar value was obtained from Greenhow's velocity correlation function in Chapter III Section 3. This is somewhat surprising, since Greenhow's correlation function does not describe the total turbulent flow field, but only the scales to 6km. Because of this, and the time dependent nature of the turbulence, it is felt that no great significance should be attached to this agreement.

### 9.10 The Reynolds Number.

The value of  $\lambda$  determined in Section 9 is probably accurate to within a factor of 2, and it is of interest to calculate the value of the turbulence Reynolds Number defined in

Chapter II Section 8 as

$$R_{\lambda} = \frac{U_0 \lambda}{\nu}$$

Substitution of the appropriate value yields

$$R_{\lambda} = 4 \times 10^3$$

which is of the same order as the critical value encountered in laboratory turbulence.

## CHAPTER X.

### The Results for 1961.

This Chapter is devoted to a discussion of the seasonal variations of turbulence parameters as deduced from the data for each month of 1961, January to December inclusive.

#### 10.1 The Velocity Differences.

The mean wind analysis has not as yet been completed for all months of the year. Whereas the total turbulence survey involves some 1800 three station echoes spread over the thirteen months, the mean wind reduction requires at least 1000 single station echoes per month. Once available, the processing of the data on the computer involves only a few hours for each month; the time consuming portions of the reduction are the initial film reading and the subsequent card punching.

Before any mean wind information was available, the velocity differences with separation for several months were computed on the basis of zero mean wind. When the mean wind extraction was subsequently performed and the data reprocessed, the velocity difference/separation relation for each month was found to be almost identical to that obtained by assuming zero mean wind. This is really not surprising when

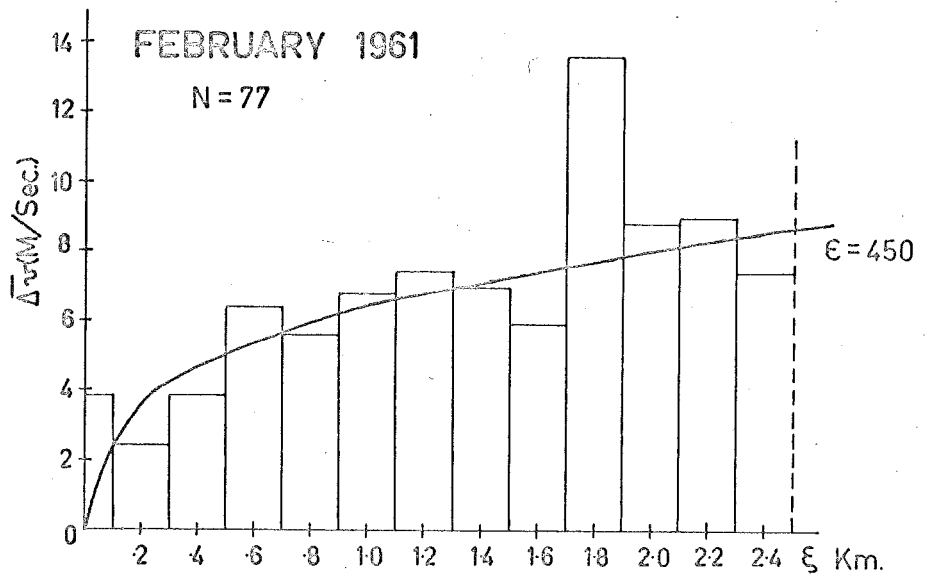
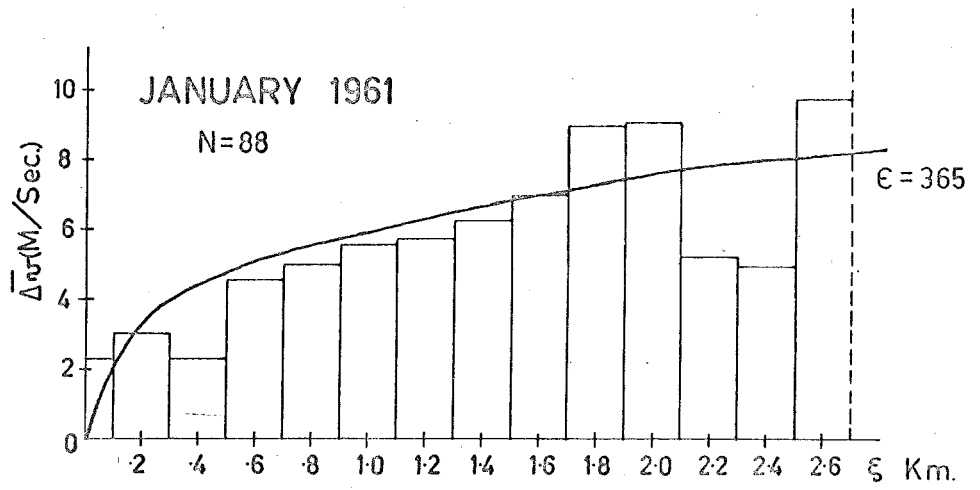


FIG. 38

one considers the factors which can influence this relation. The mean height gradient is always less than the turbulent gradient, and thus has only a small effect on the measured velocity differentials.

The value of the turbulent dissipation rates calculated from the velocity differences for January, February, April, May, October and November, for which the mean wind has not been taken into account, should be regarded as tentative, although no appreciable modification is expected in the final analysis.

The histograms of velocity differences against separations for the months January to December 1961 inclusive are presented in Figs. 38 to 44, together with the best fit theoretical curves determined as in Section 3 of Chapter IX. In comparing theory and experiment four factors must be taken into consideration:-

1. The size of the sample. At least 10 velocity differences are required for each separation to obtain a significant RMS value. This is usually achieved for values of separation up to 2.6km. In general, the total number of echoes in each sample has to exceed 80 (yielding 240 velocity differences) in order to obtain significant velocity differentials for a given month. For the early months of the year, the echo rate was reduced because of equipment troubles. This

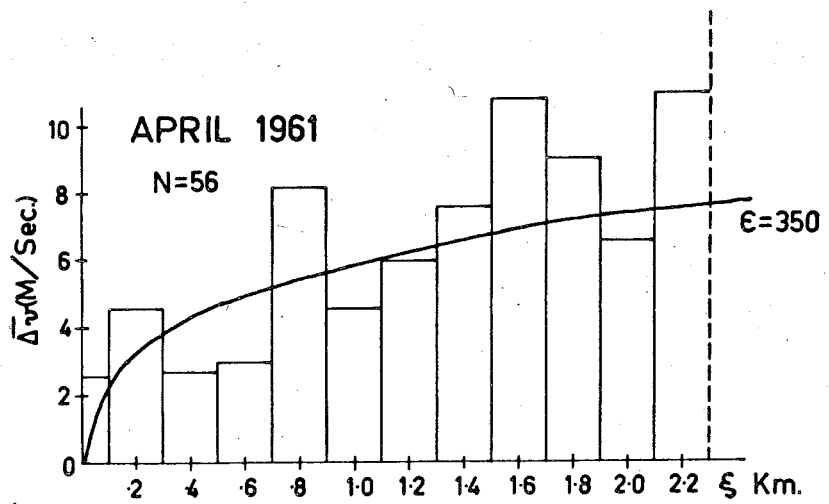
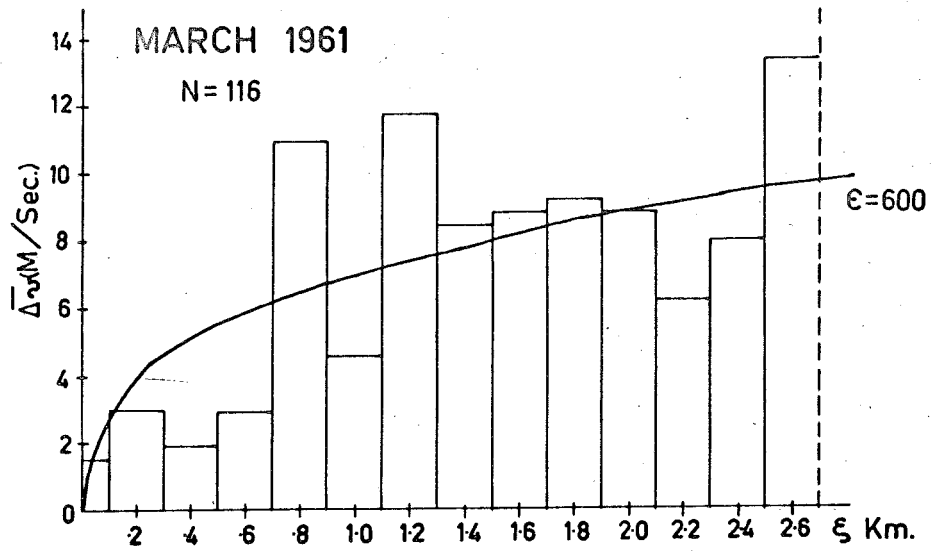


FIG. 39

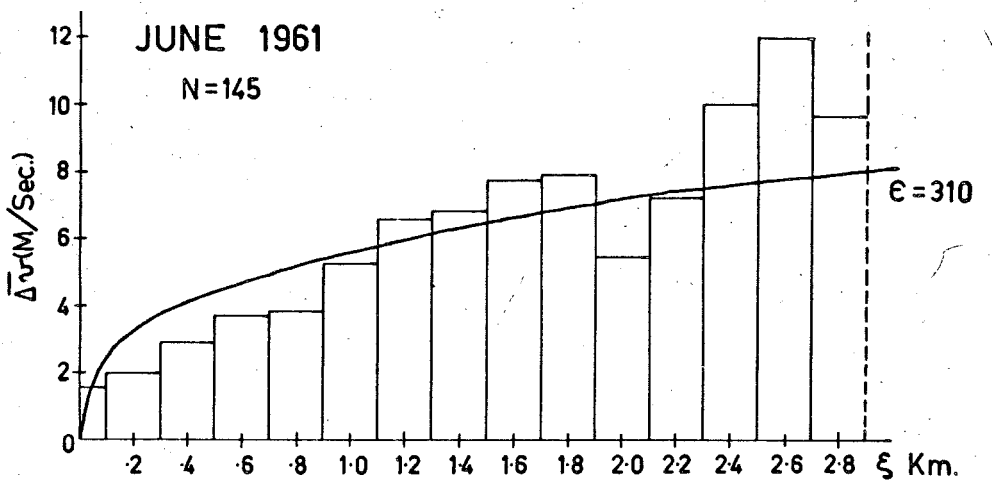
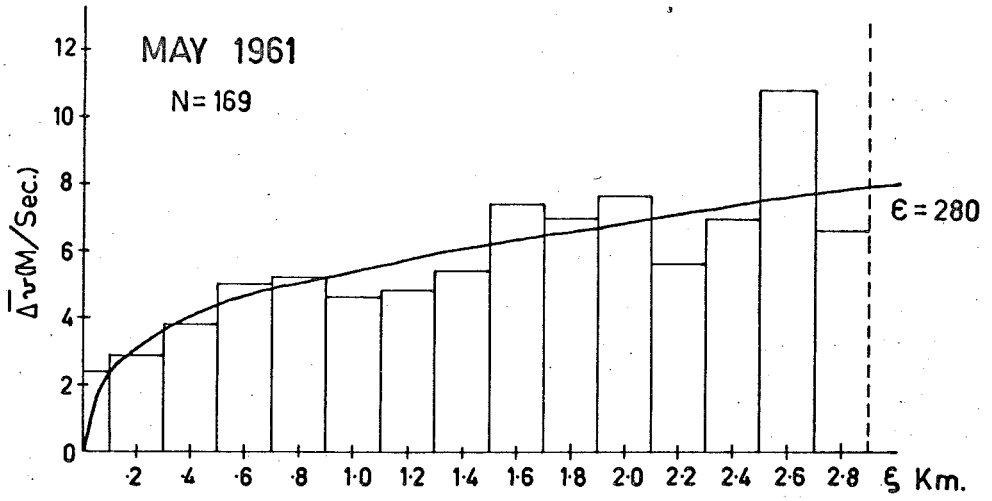


FIG. 40

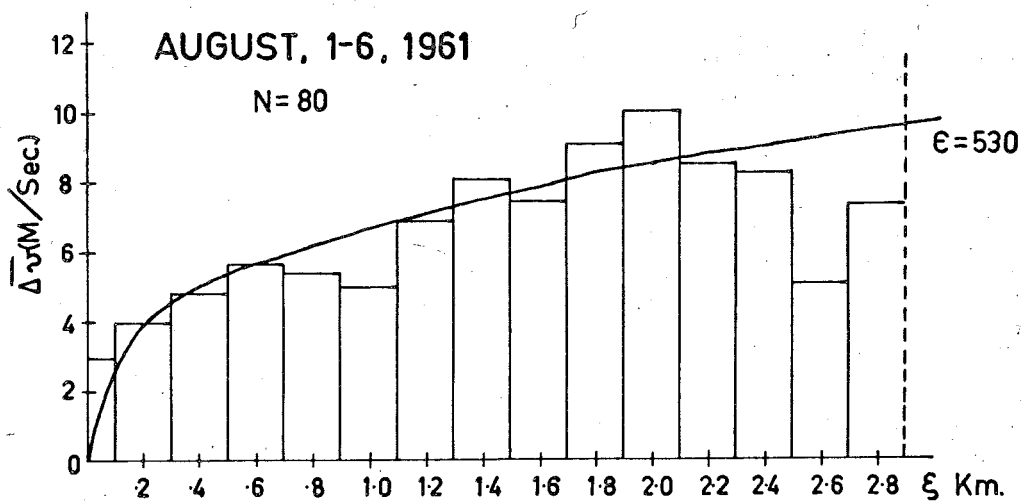
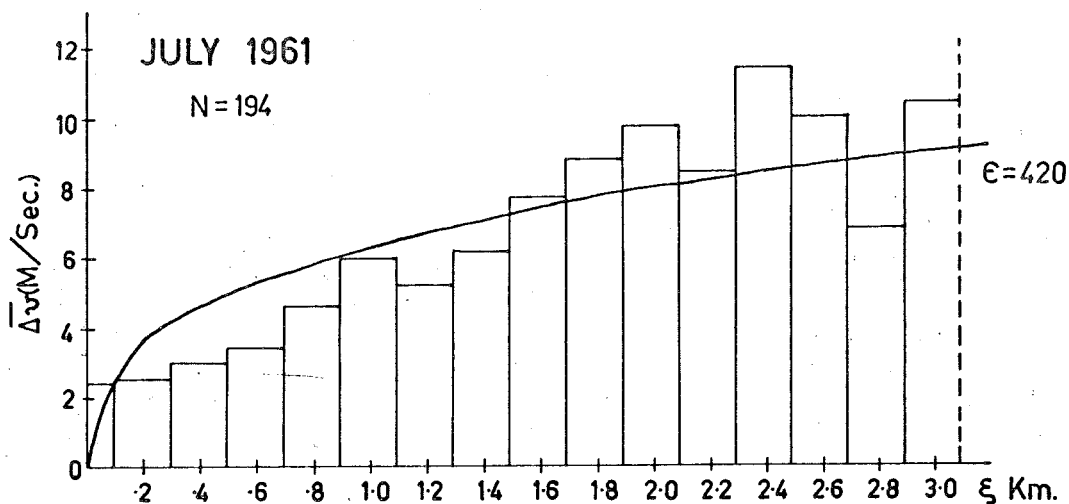


FIG. 41



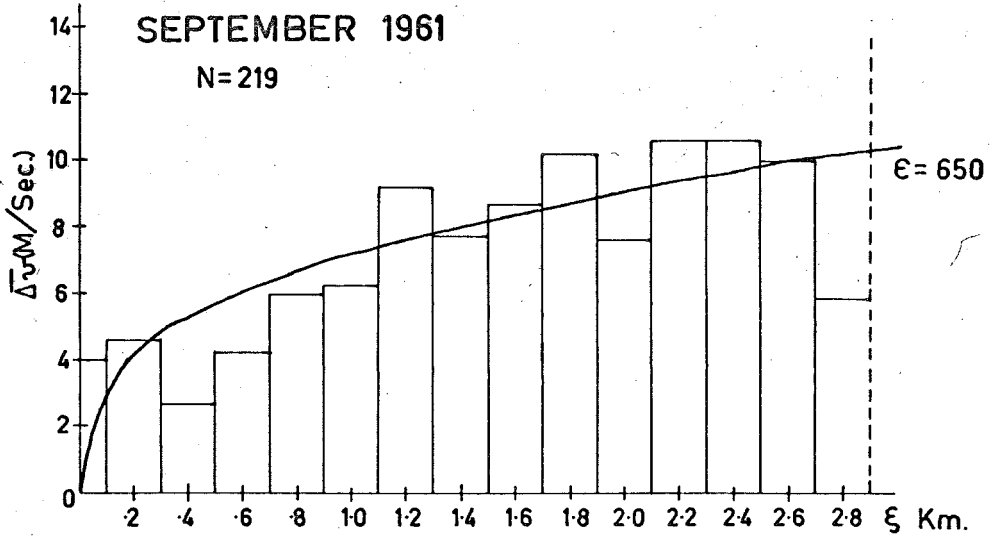
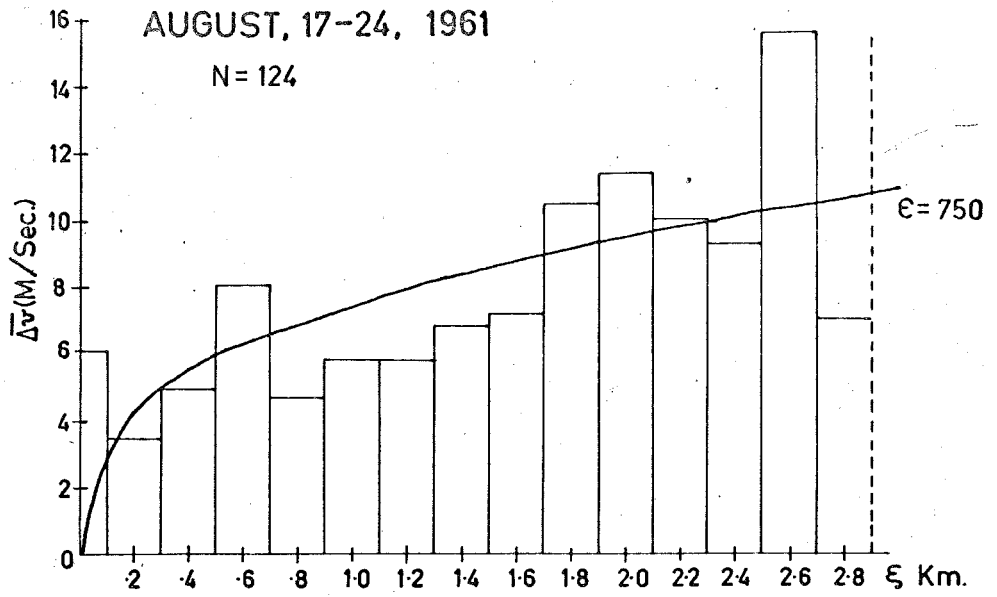


FIG. 42

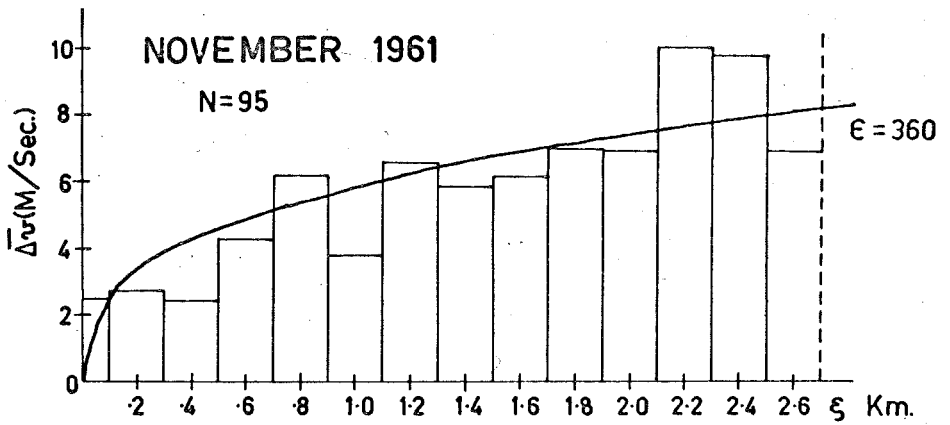
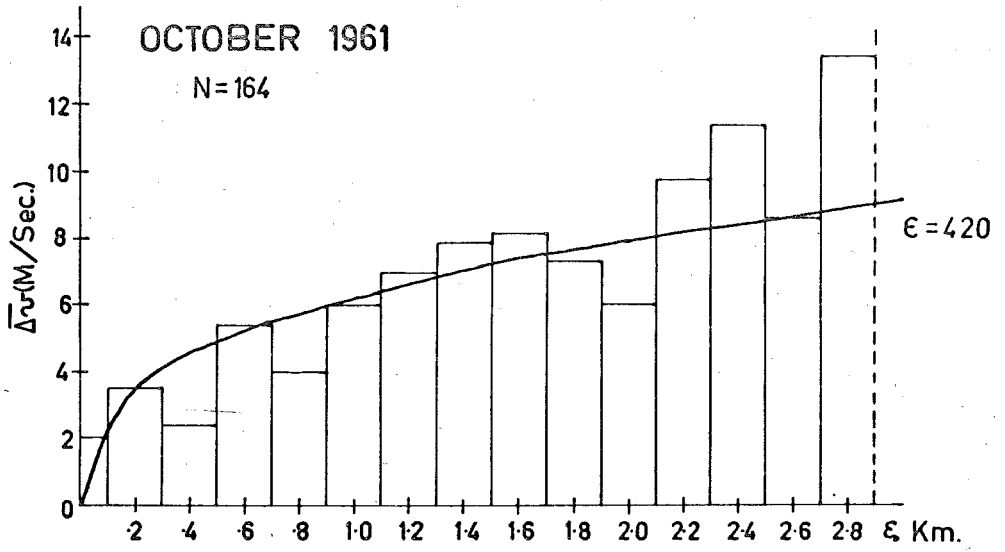


FIG. 43

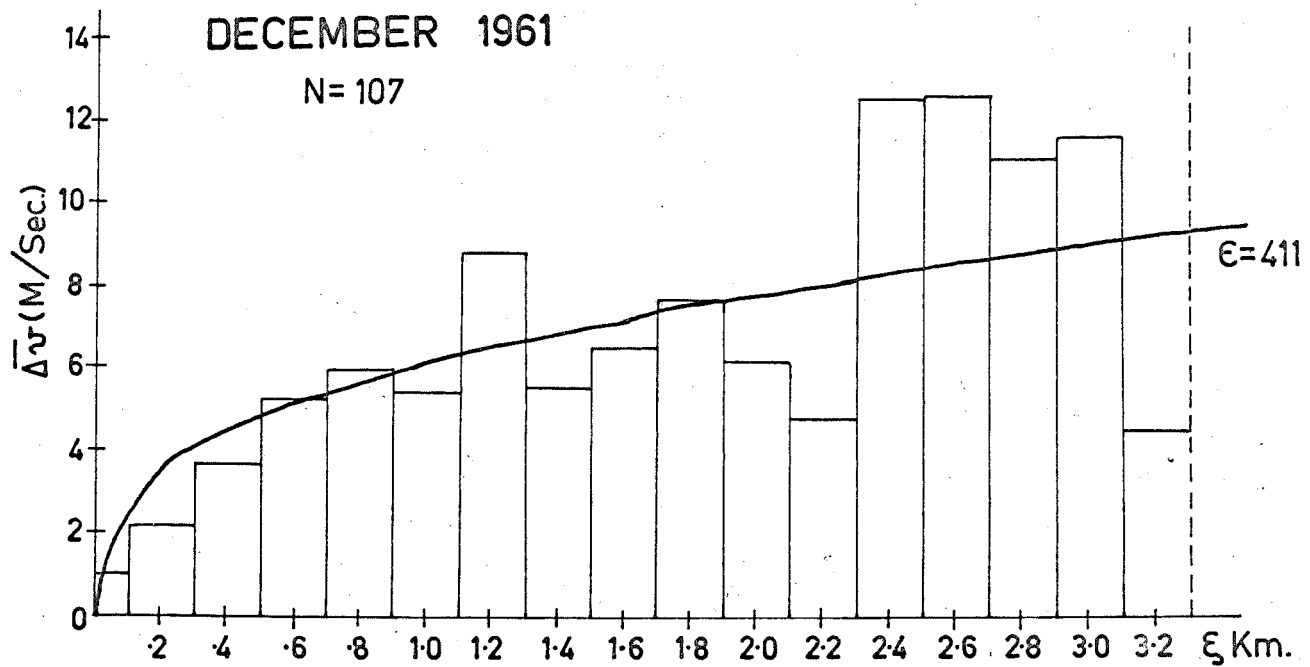


FIG. 44

is reflected in poor fits to the theoretical curves for these months.

2. At scales less than 200 metres anomalous velocity differentials are prevalent, due to the inaccuracy associated with the separation measurement brought about by the phenomenon of reflection point motion (see Chapter 6 Section 3).

3. The observations show a marked deviation from the theoretical curves at separations less than approximately 1km, and, to a lesser extent, an overall periodic fluctuation. This has already been explained in terms of the action of buoyancy forces. In general, these effects are greatest for the more disturbed months in which the turbulent dissipation rate exceeds 450 ergs/gm/sec. In fact, for August 17th to 24th it is almost impossible to decide on the best fit curve for the estimation of  $\epsilon$ , and the actual dissipation rate for this month is probably somewhat less than that estimated.

4. The fourth factor involves the time dependence of the turbulent flow field, which has already been demonstrated for December, 1960, and which is confirmed by the other months analysed in Section 4 of this Chapter. The turbulent dissipation rate is expected to depend upon  $U_0$ , the velocity characteristic of the energy bearing eddies. The diurnal variation in  $U_0$  is of the order of 2:1. Thus the turbulence

spectrum will be time dependent, and the histograms of Figs. 38 to 44 representative of an average over the characteristically different turbulent flow fields which exist throughout the day. If, for any reason, the sampling rate is not uniform, the spectrum determined will be biased towards the conditions pertaining during the high rate hours. In general, these will be the early morning hours, since the earth "sweeps up" more meteors at 0600 hours than it does at 1800 hours.

#### 10.2 The Seasonal Variation of the Turbulent Dissipation Rate $\epsilon$ .

The  $\epsilon$  values of the theoretical curves fitted to the results of Section 1 are considered to be the best estimates yet determined of the turbulent dissipation in the 80 to 100km region of the upper atmosphere.

In determining these values, the value of  $\alpha$ , the constant of proportionality in the expression

$$E(\kappa) = \alpha \epsilon^{2/3} \kappa^{-5/3}$$

has been taken as 2/3 (see Chapter II Section 10).

The value of  $\epsilon$  determined from the velocity difference/separation relation

$$\overline{\Delta v^2} = 4.82\alpha(\epsilon\xi)^{2/3}$$

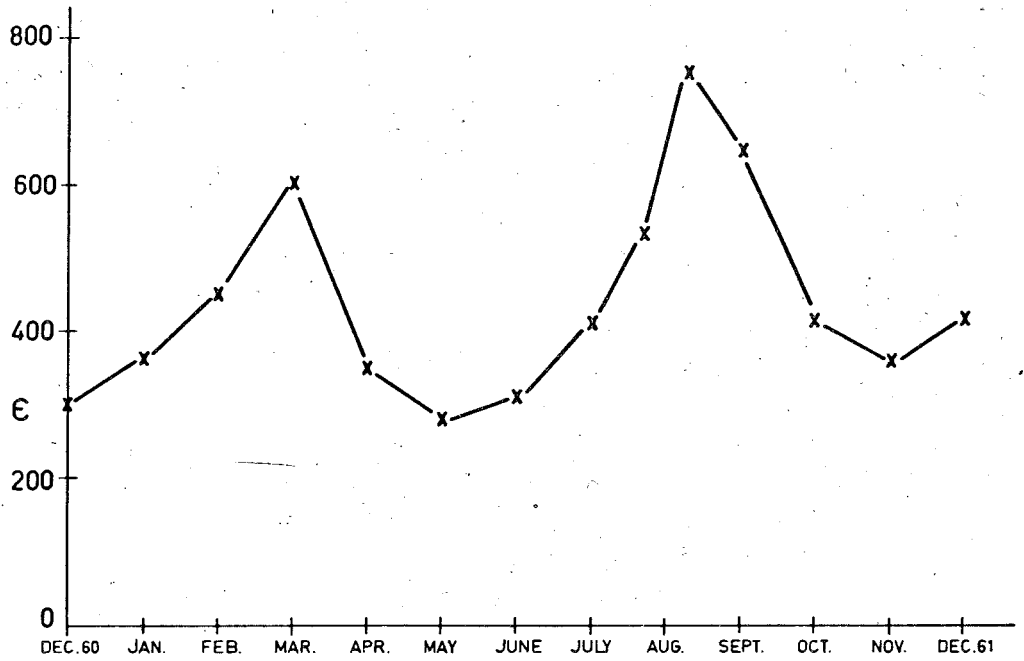


FIG. 45a - SEASONAL VARIATION OF THE TURBULENT DISSIPATION RATE.

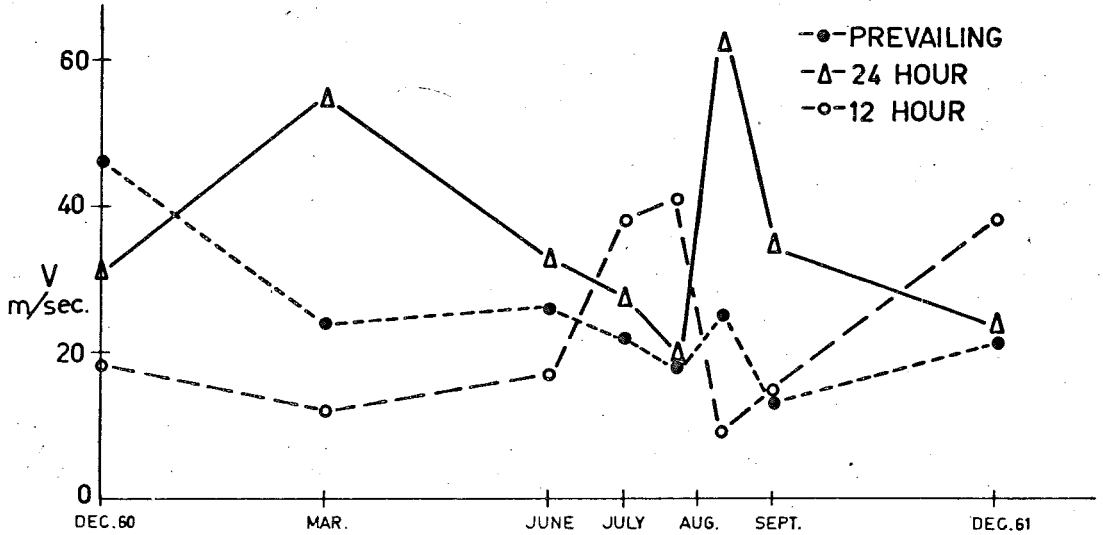


FIG. 45b - SEASONAL VARIATION OF THE MEAN WIND COMPONENTS.

depends on  $\alpha^{\frac{1}{2}} \epsilon$ .

If  $\alpha$  is taken as 1.0, then the resultant turbulent dissipation rates will be approximately half those given in Fig. 45a. ( See Chapter XI, Section 7. )

Unfortunately, the upper limit of  $\epsilon$  is not well established, since increased turbulence is accompanied by increased activity of the buoyancy subrange, and the velocity differences no longer follow the Kolmogoroff spectrum. Nevertheless, the relatively smooth rise in  $\epsilon$  establishes the equinoxial maxima, and the maximum value of 750 ergs/gm/sec is probably a reasonable, if somewhat high, estimate.

It is of interest to compare the seasonal variations of  $\epsilon$  and the three components of the mean wind as shown in Fig. 45. The prevailing and semidiurnal components do not appear to be in any way related to the turbulent dissipation, but the behaviour of the magnitude of the 24 hour component throughout the year is very similar. This suggests that the turbulent flow field may have its origin in the diurnal variation of the mean wind.

### 10.3 The Dependence of the Velocity Differentials on Height Differences.

The velocity differentials as a function of height difference have been calculated for those months for which the mean wind is available. The results are presented in Figs.

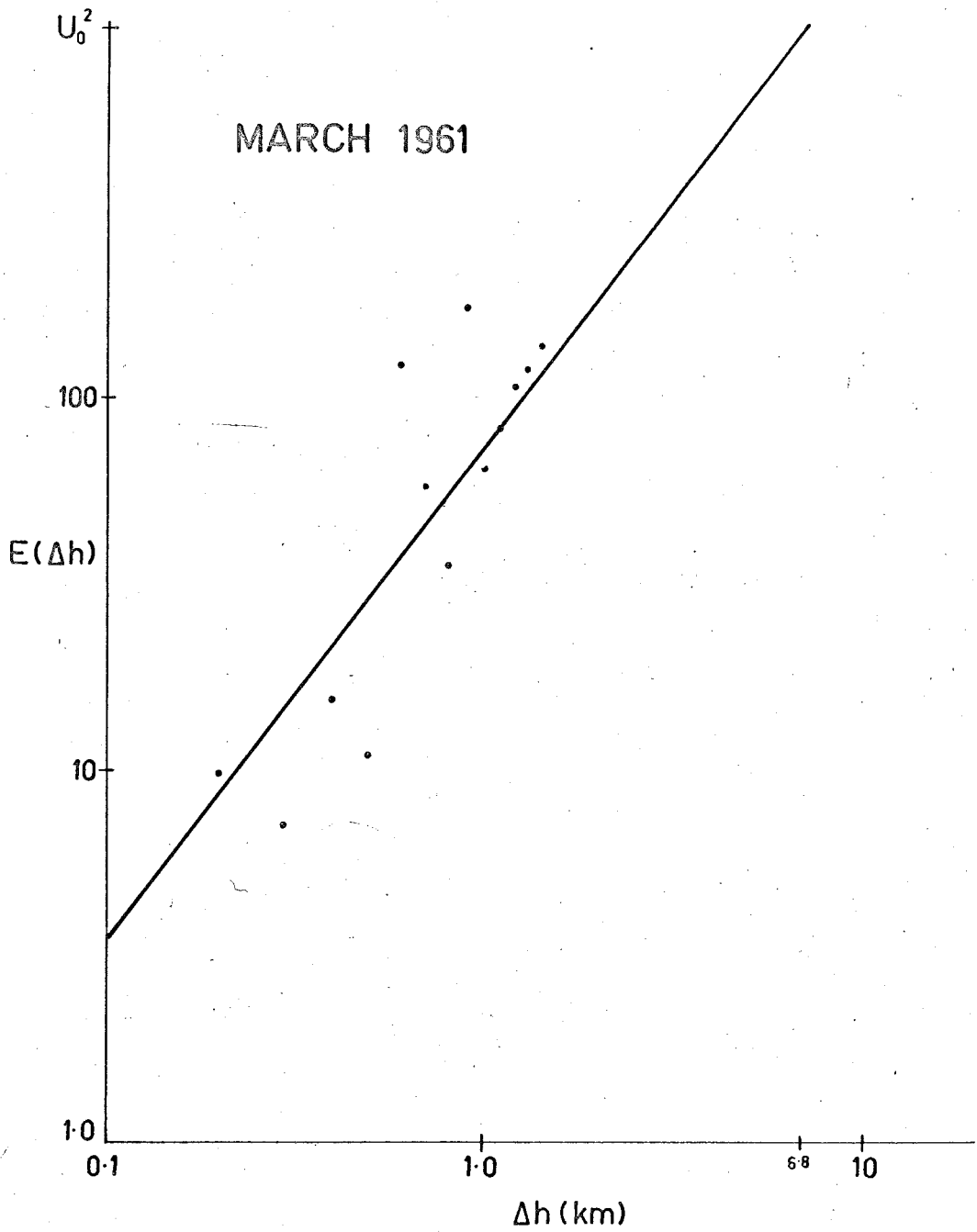


FIG. 46



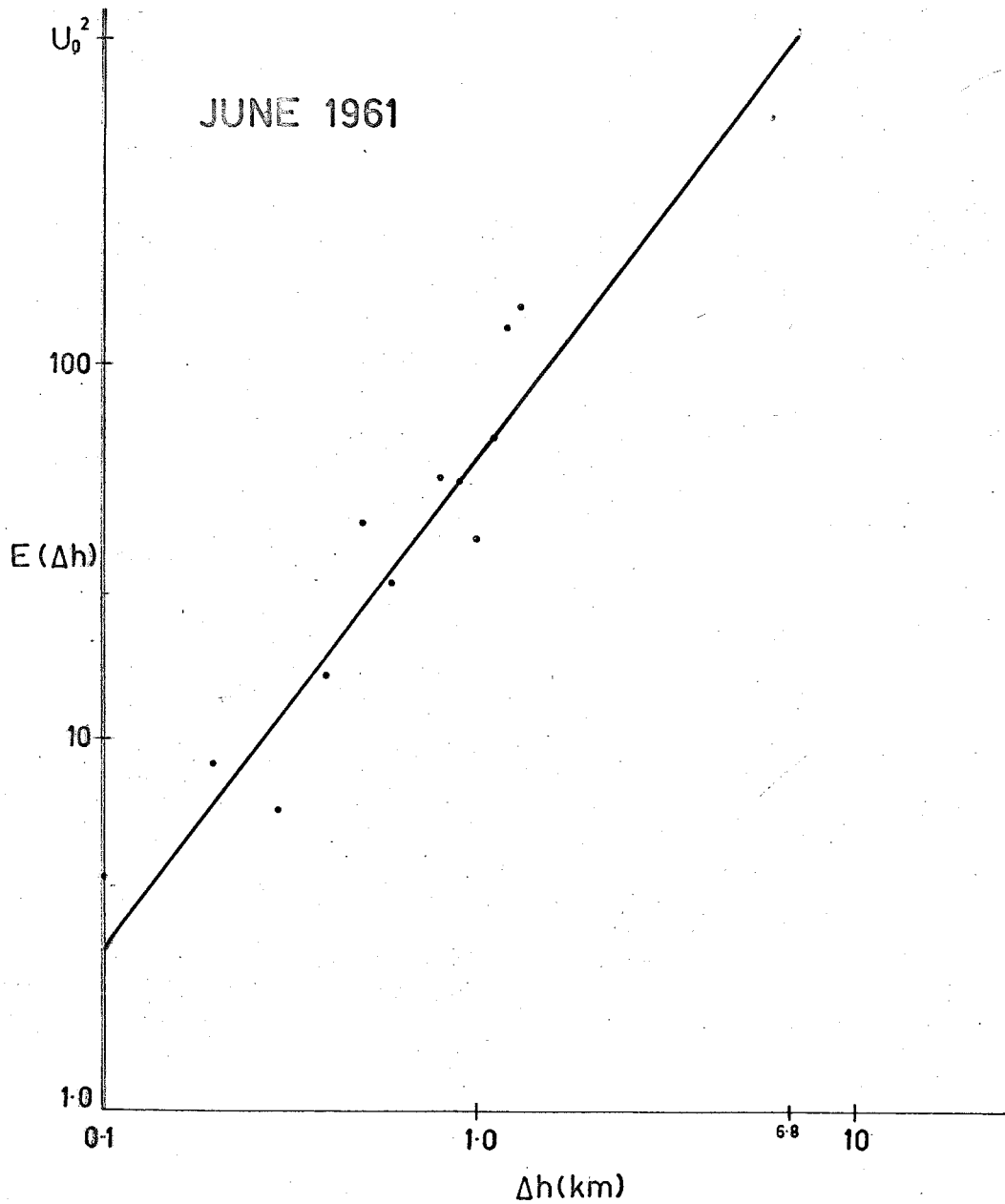


FIG. 47

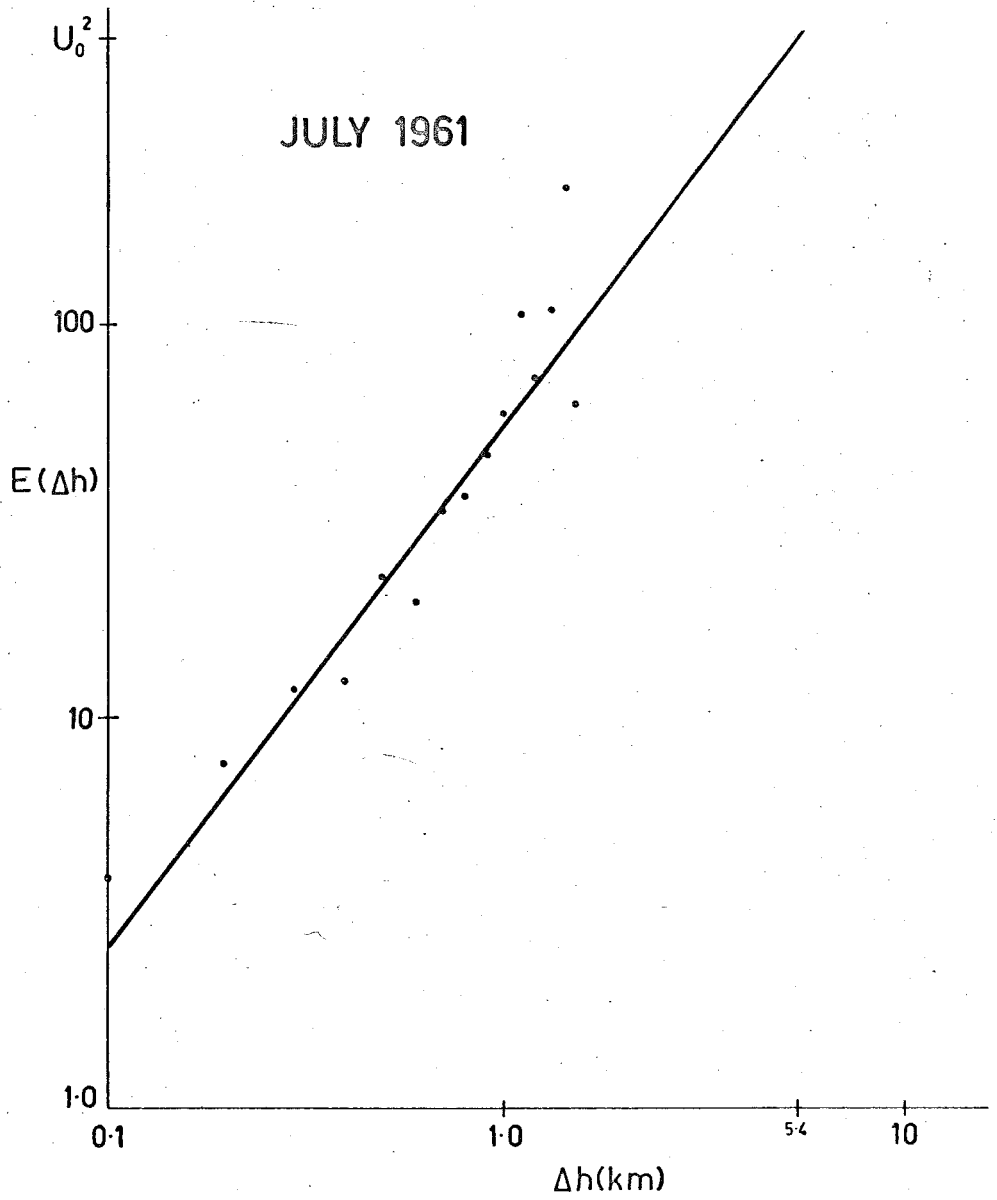


FIG. 48

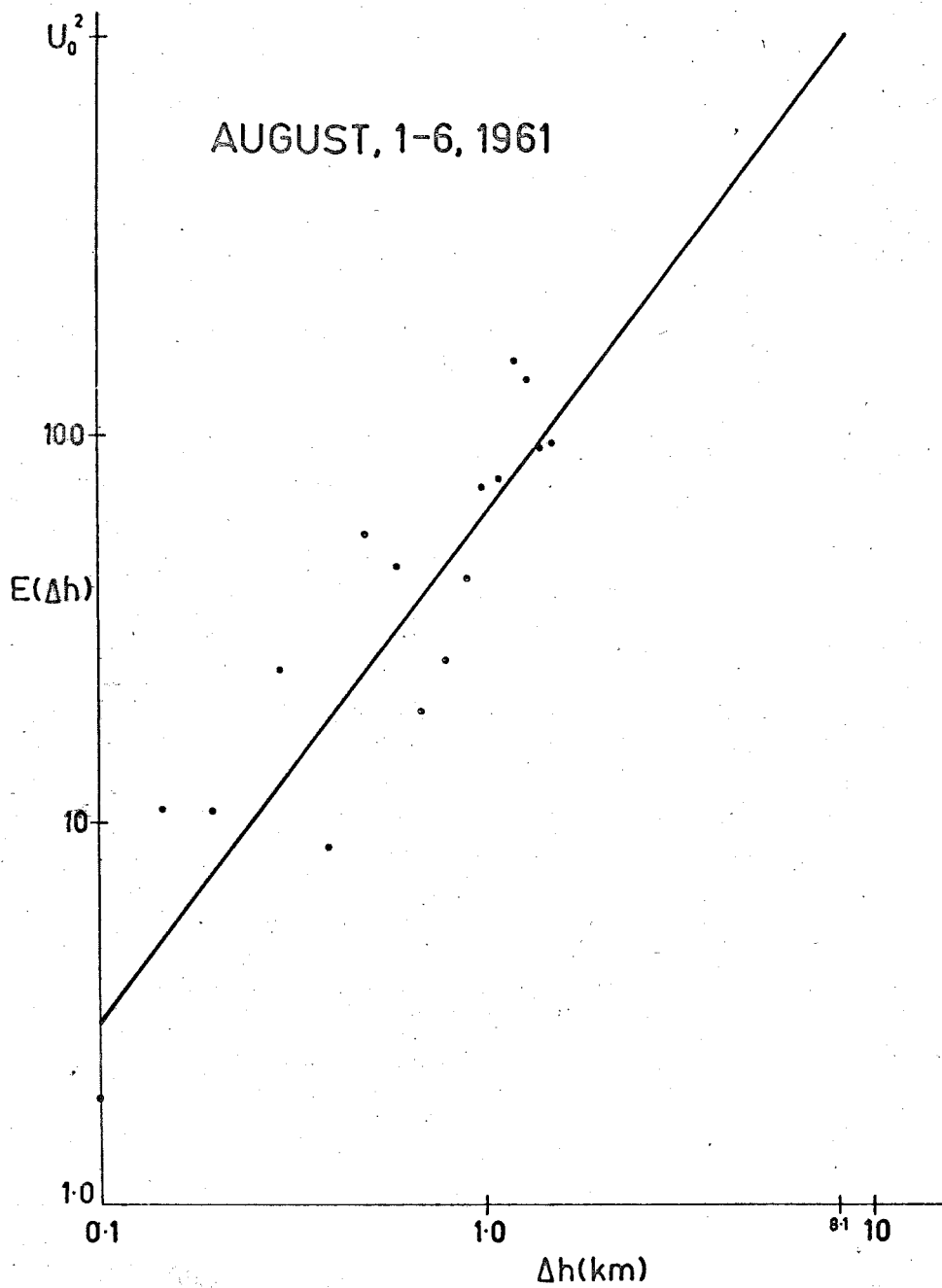


FIG. 49

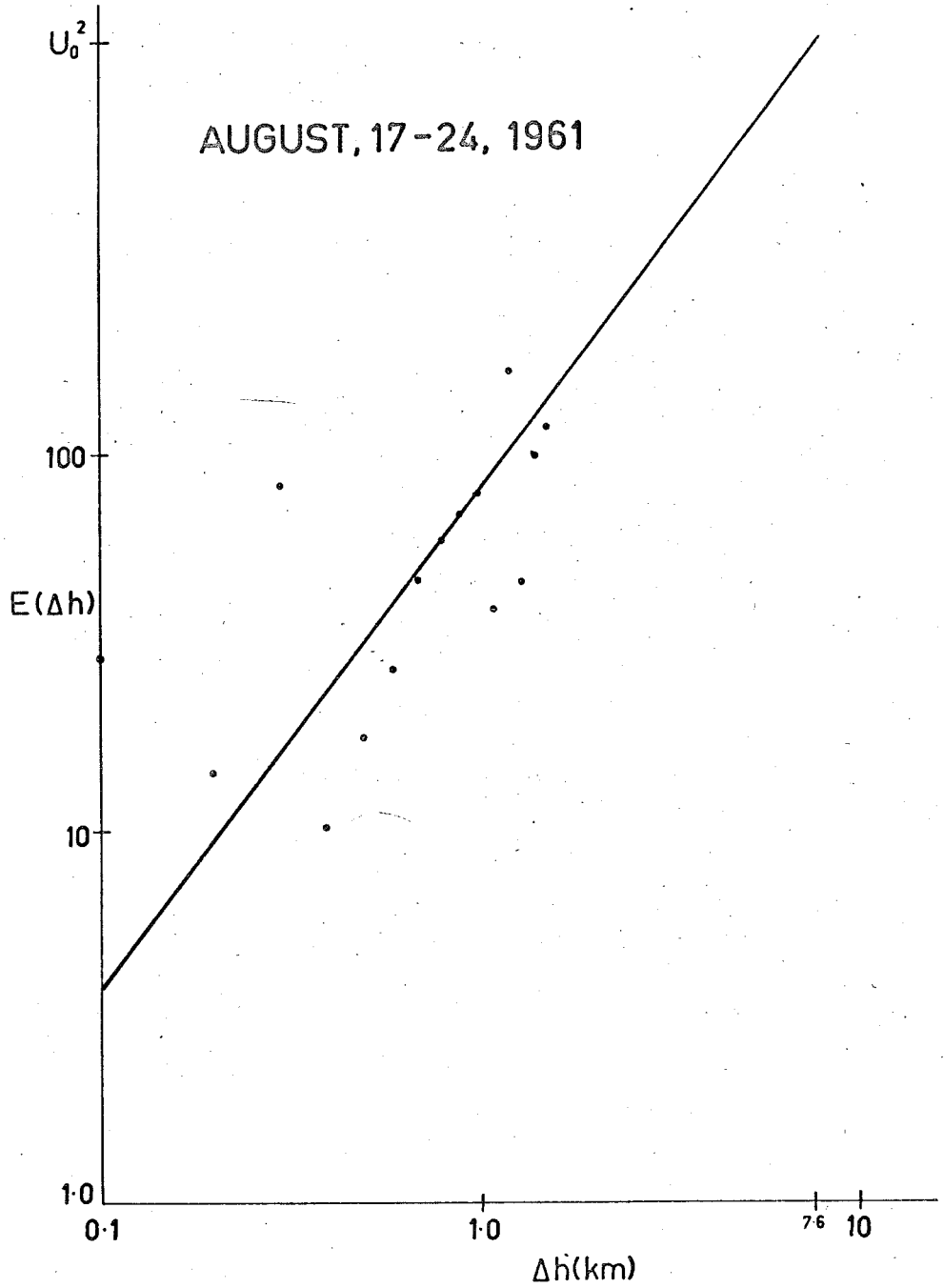


FIG. 50

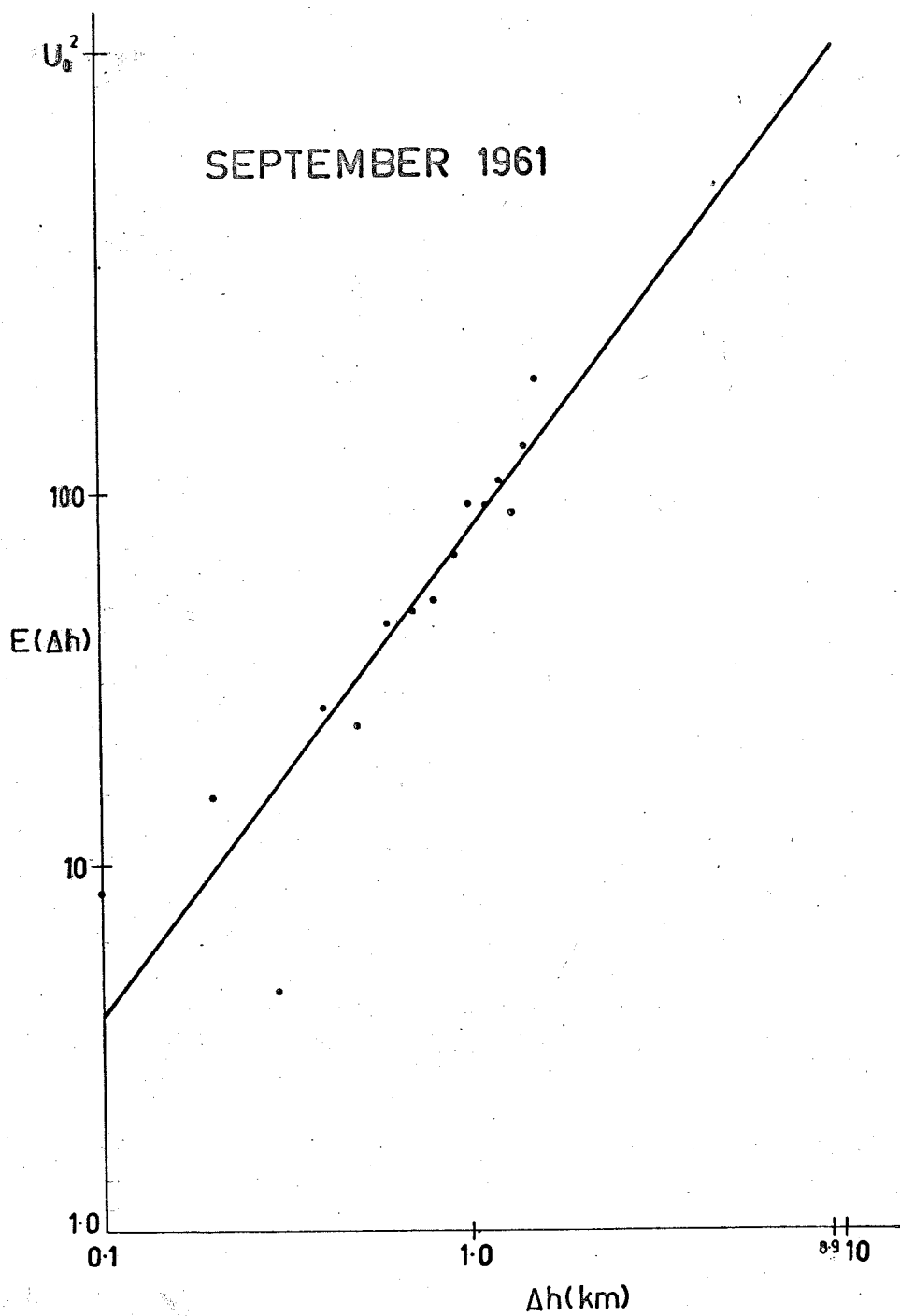


FIG. 51

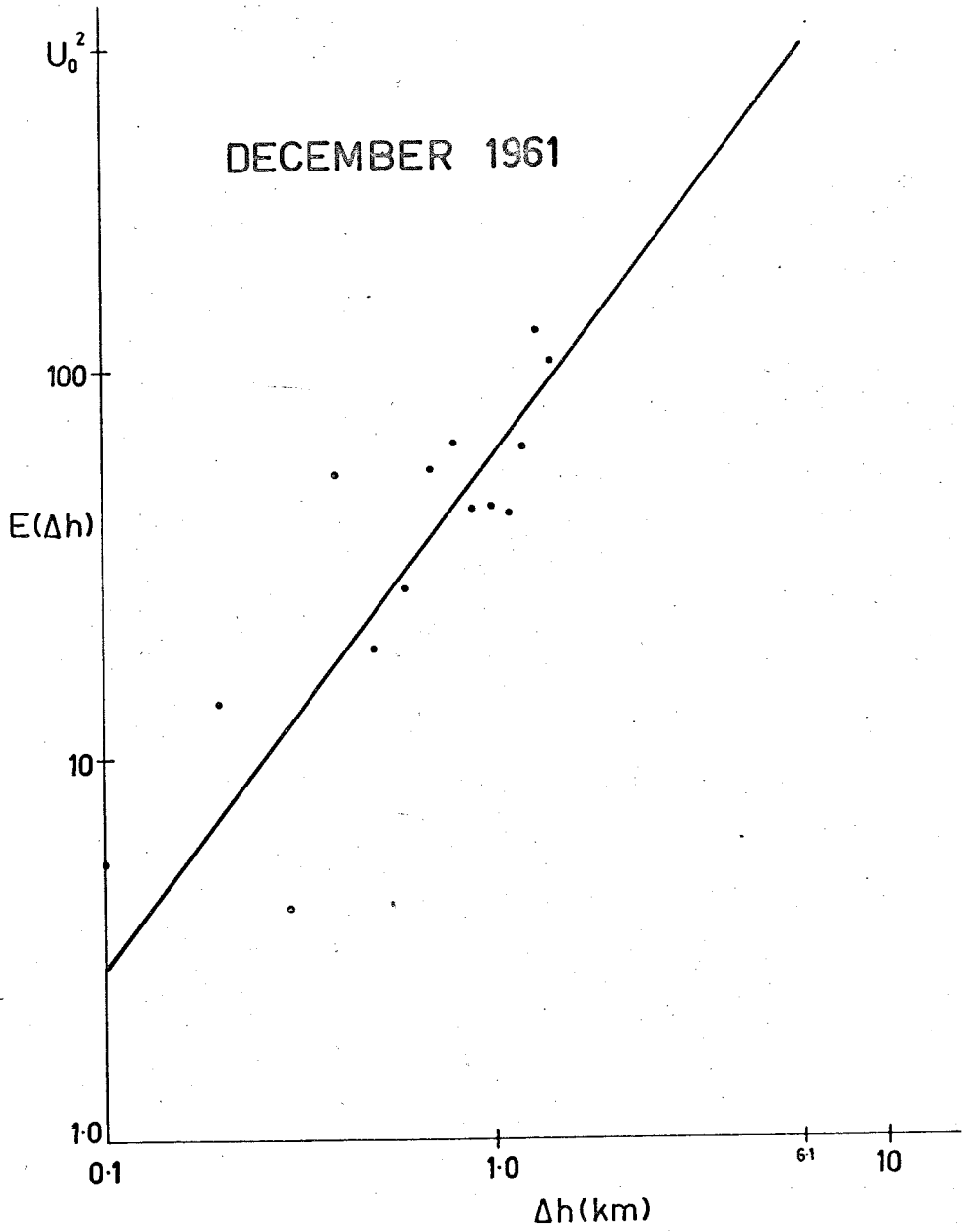


FIG. 52

46 to 52 inclusive. Most of the scatter in velocity difference for each month appears to be periodic and most prevalent below height separations of 1km, due, at least in part, to the action of the buoyancy subrange. Results for all months tend to follow the  $4/3$  power law indicative of shear with height. There is only a small variation throughout the year in the turbulent velocity gradient as given by the RMS velocity difference measured over a height difference of 1km. The best fit lines of slope  $4/3$  correspond to gradients ranging from 7.5 metres/sec/km (June) to 9 metres/sec/km (September).

In an attempt to determine the vertical correlation distance, the plots of Figs. 46 - 52 have been extrapolated to the values of  $E(\Delta h)$  representative of the characteristic velocities of the energy bearing eddies. Because of the anisotropic nature of the large eddies, the characteristic velocity  $U_0$  determined as the RMS over all the line of sight turbulent velocities measured in a given month (see Chapter IX Section 2) is not the actual velocity characteristic of the predominantly horizontal large scale turbulent flow. However, the velocity  $U_0$  does characterize the flow as determined by line of sight observations.

The vertical correlation distances thus determined range from 5.5km (July) to 8.9km (September). This variation in

scale may represent a seasonal change, but since the statistical significance of these values has not been examined, they are included only as substantiating evidence for the validity of the  $4/3$  power shear law at the scales measured, since extrapolation from these scales yields a vertical correlation distance similar to the value of 6.5km observed by other workers in this field.

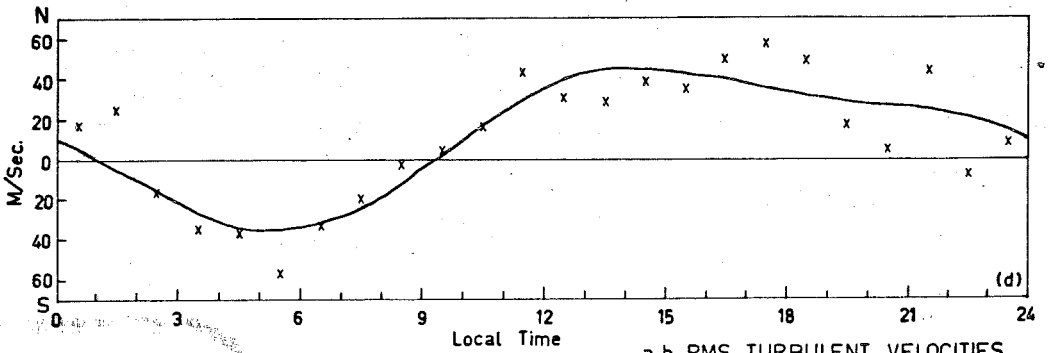
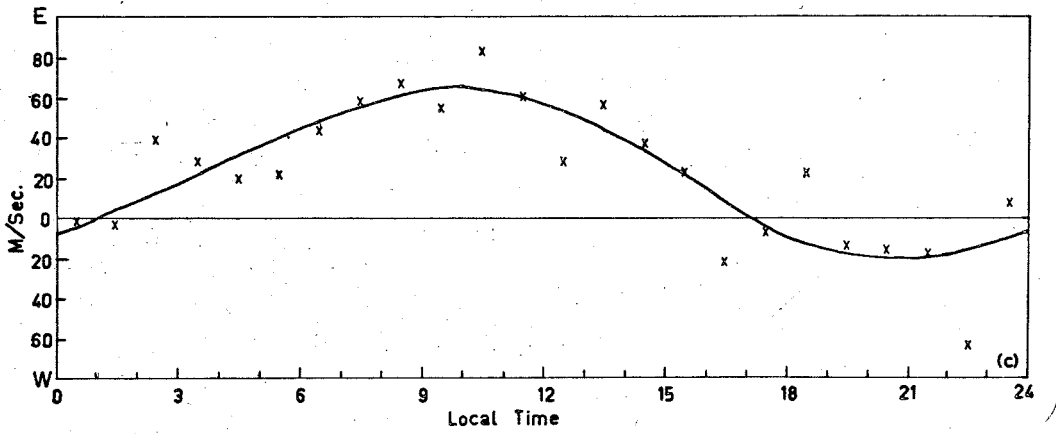
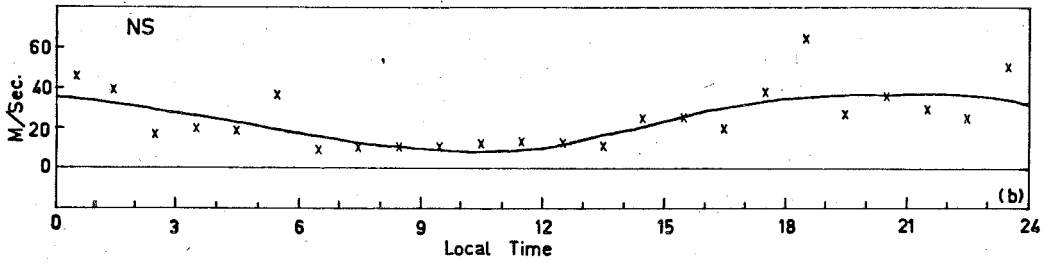
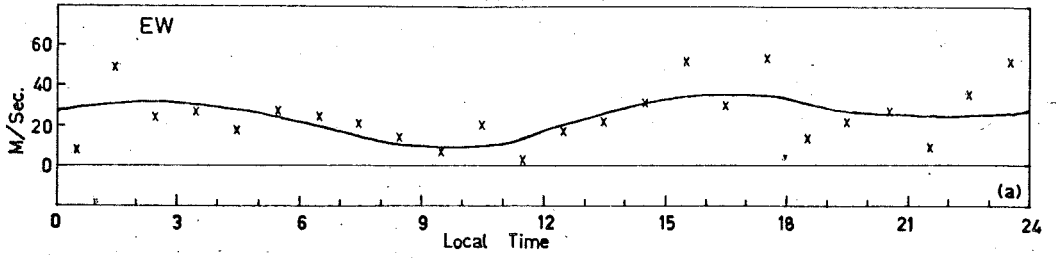
#### 10.4 The Seasonal Variation of the Turbulent Velocity.

The hour by hour turbulent velocity components, together with the mean wind for the height range 85 to 94km for the months of March, June, July, August, September and December, 1961, are presented in Figs. 53 to 59 inclusive.

The maximum turbulent velocity is, on the average, less than half the mean wind maximum. Hour by hour variation in the turbulent velocity is evident for all months; the maximum turbulent velocity occurs in the early morning in the summer, and in the late afternoon in the winter; in general the minimum value is attained some 8 hours before the maximum. It is encouraging, in considering these seasonal variations, to note that the results for December 1961 show all the general features of those for December 1960. The overall increase in the December 1961 turbulent velocity is presumably due to the increased complexity of variation of the mean flow.



MARCH 1961



a,b RMS TURBULENT VELOCITIES  
c,d MEAN WINDS, 85-94 KM.

FIG. 53

JUNE 1961

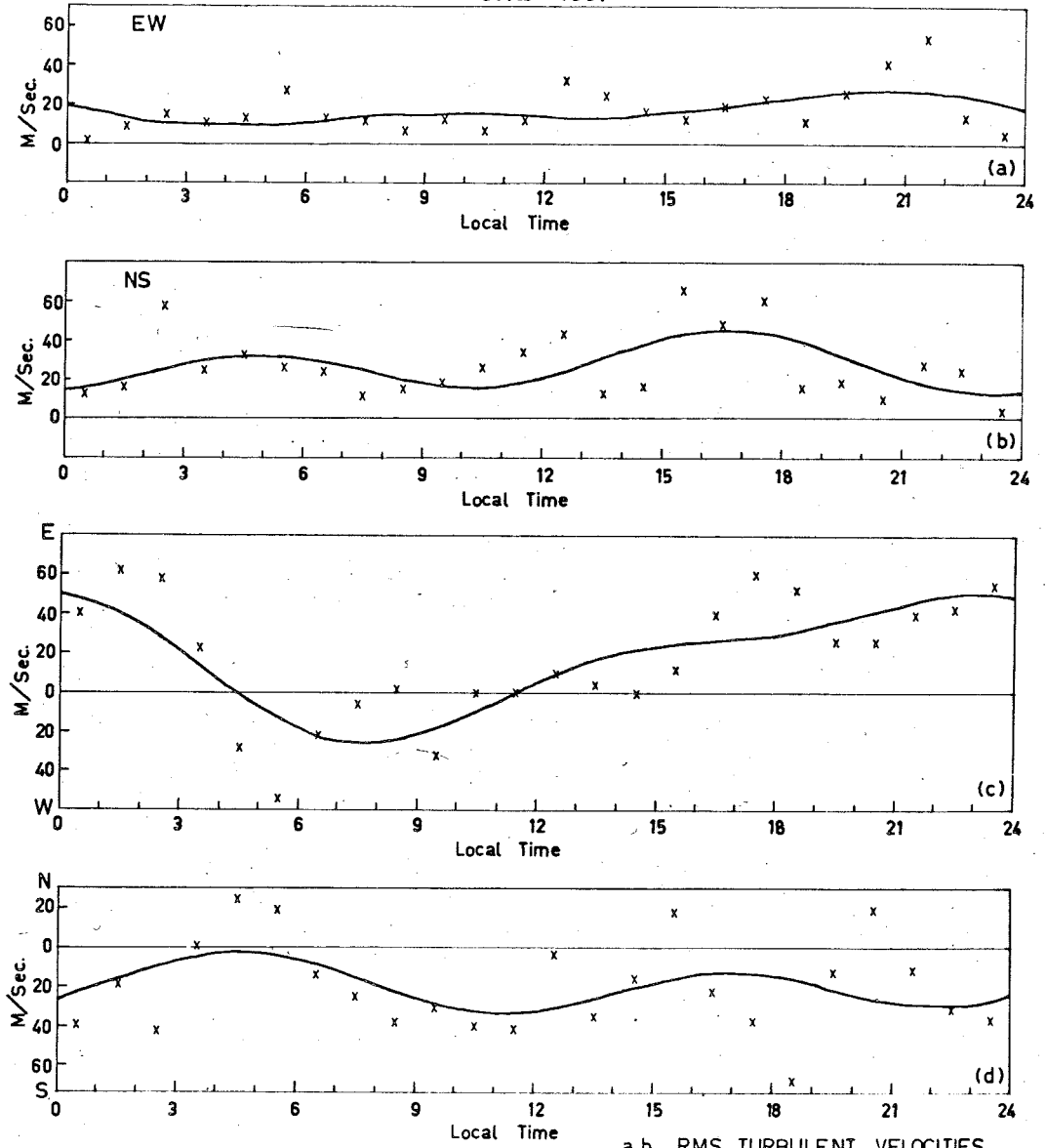
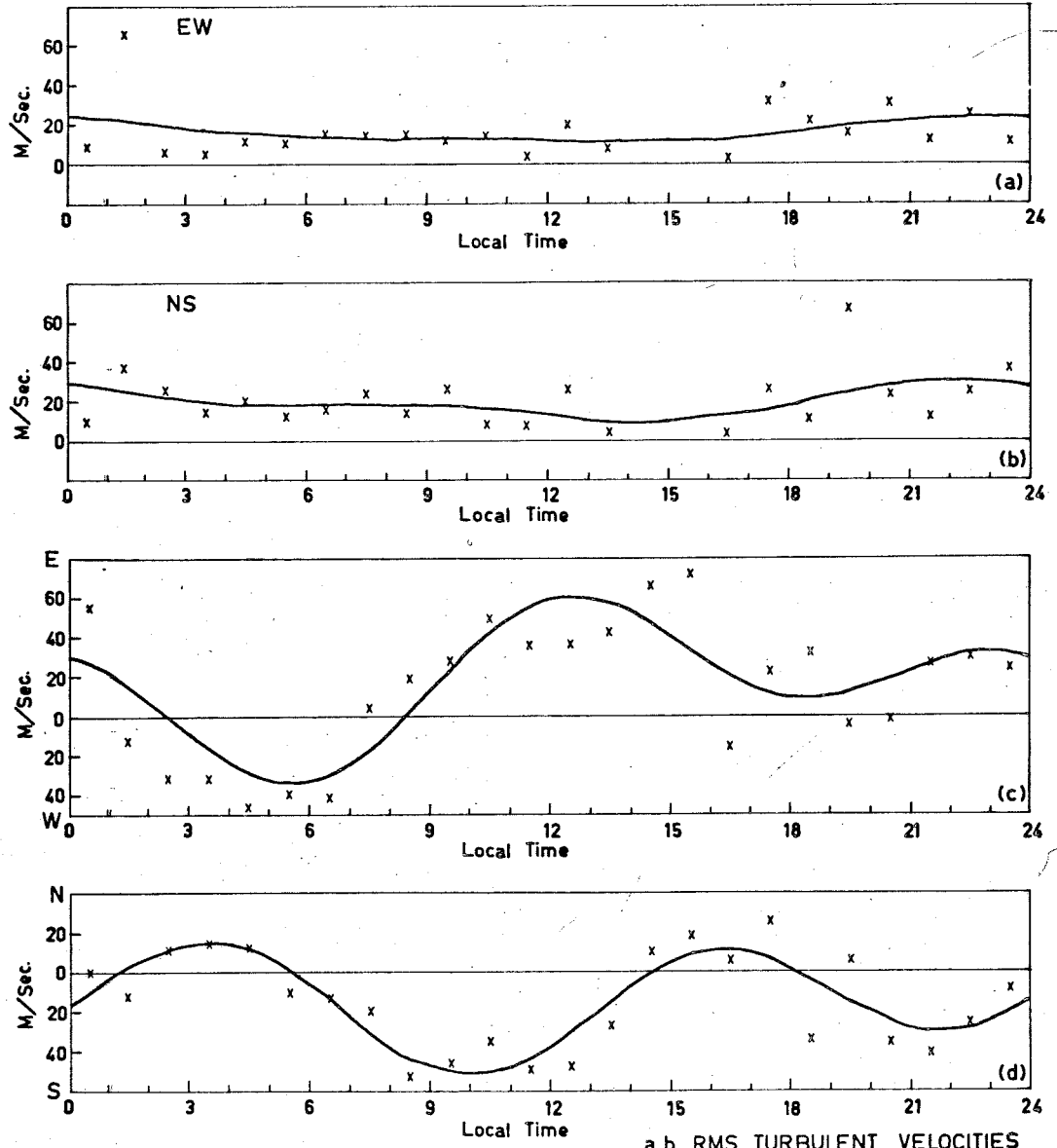


FIG. 54

a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94 KM.

JULY 1961



a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94 KM.

FIG. 55

AUGUST, 1-6, 1961

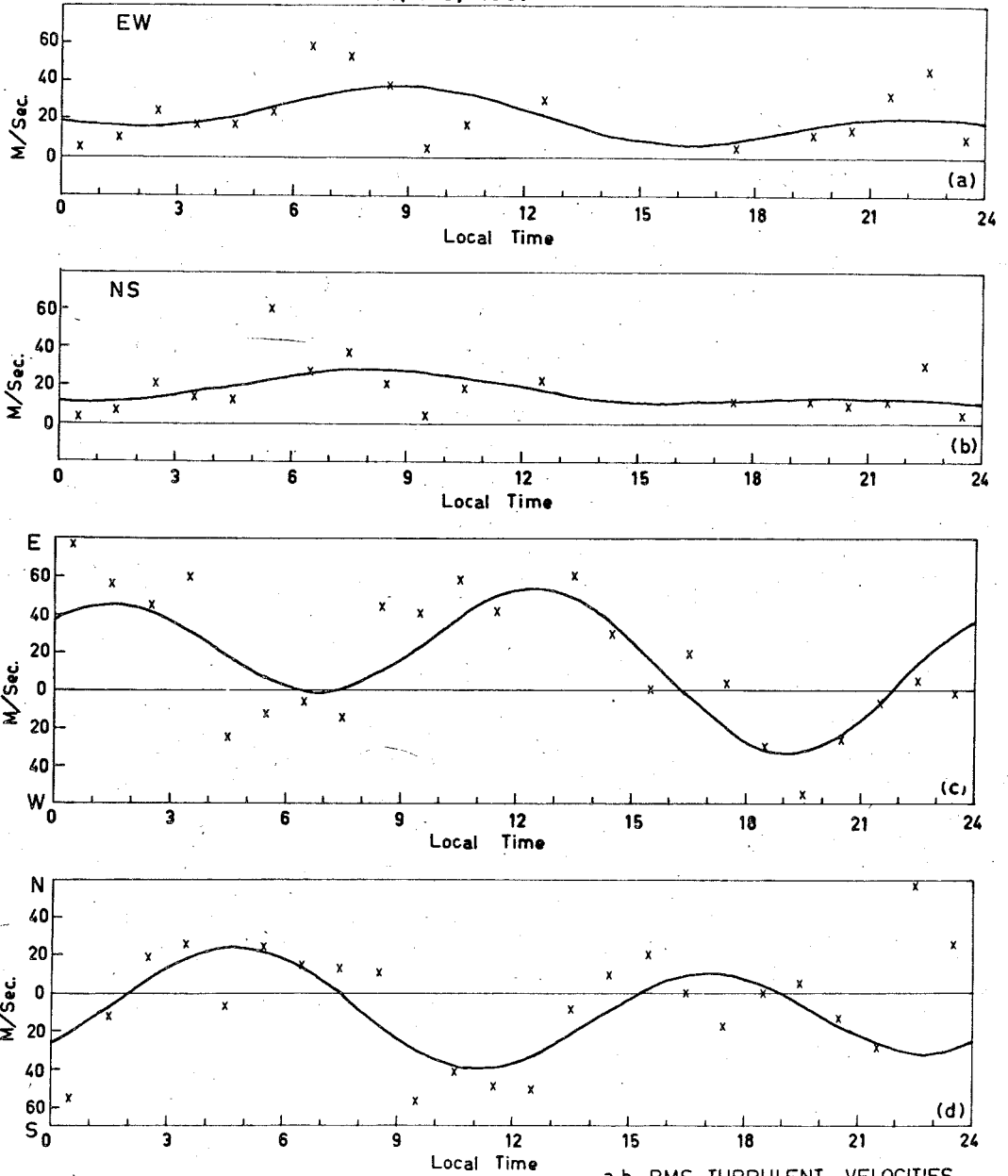
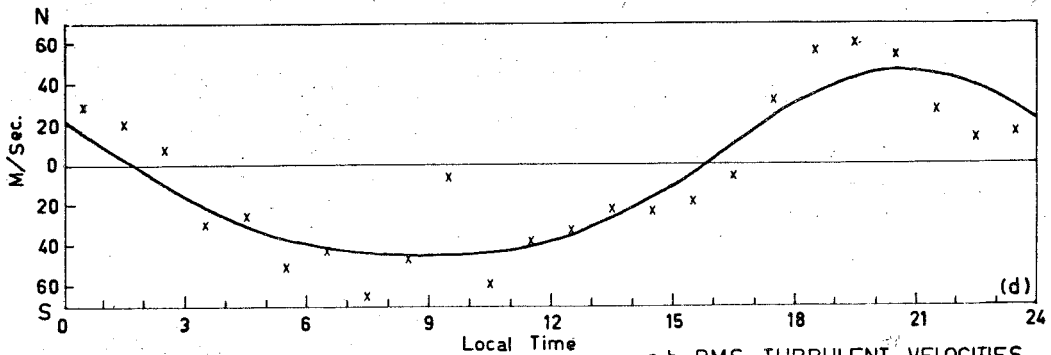
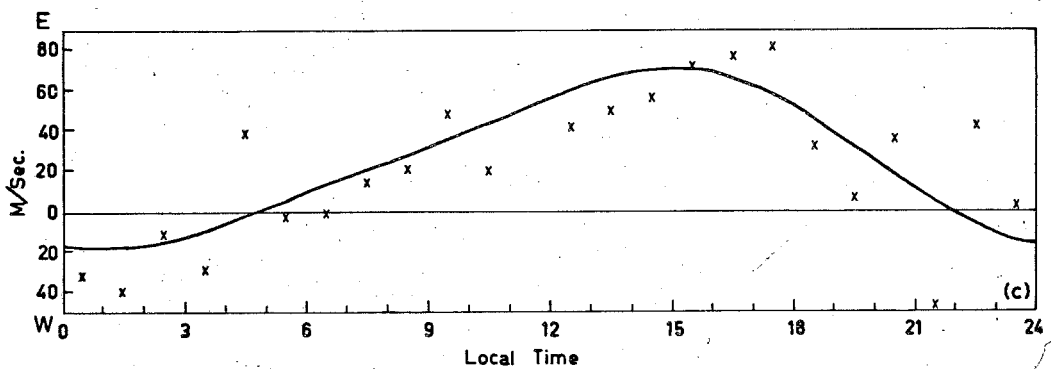
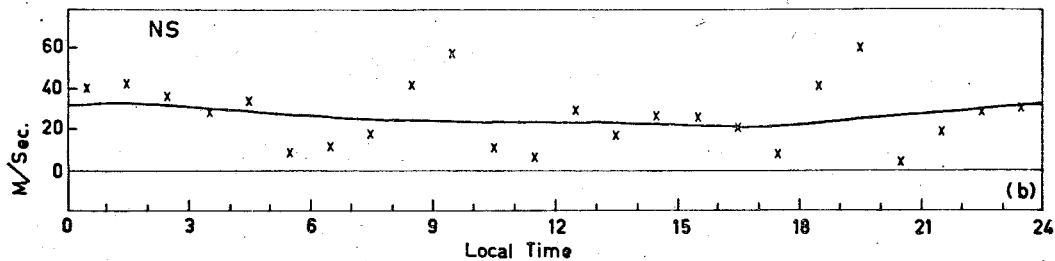
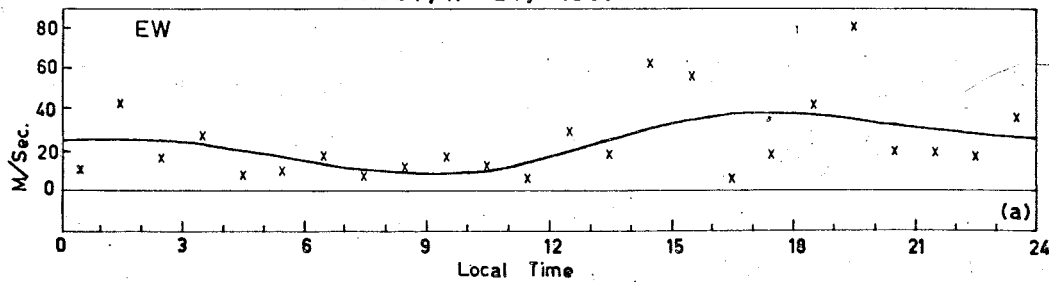


FIG. 56

a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94 KM.

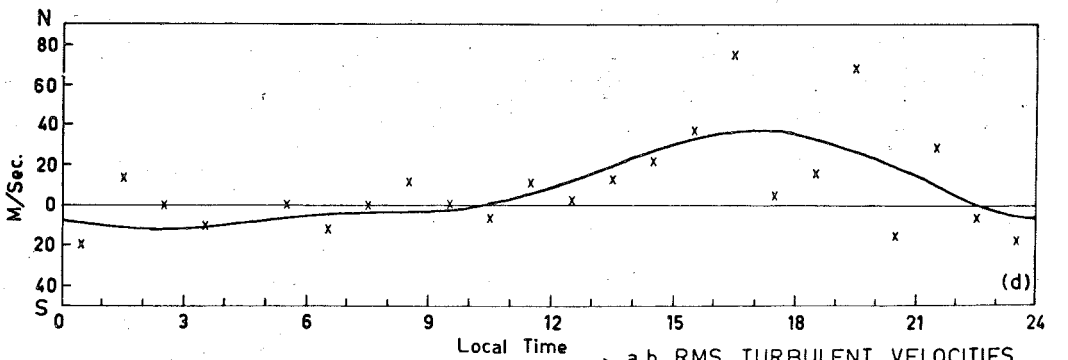
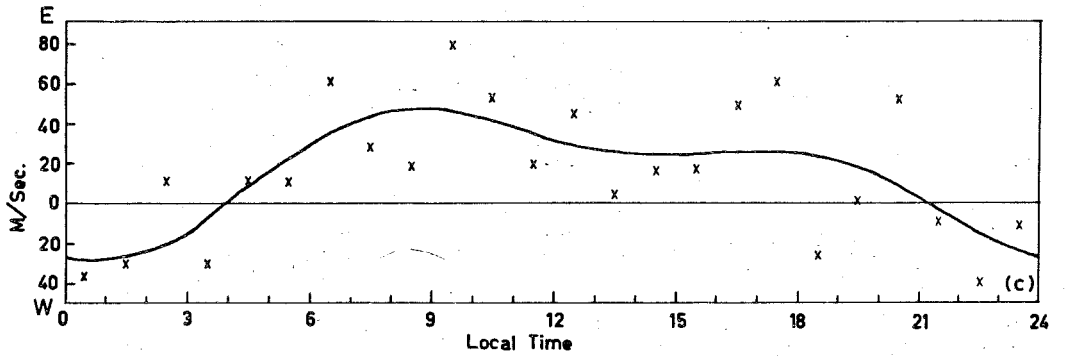
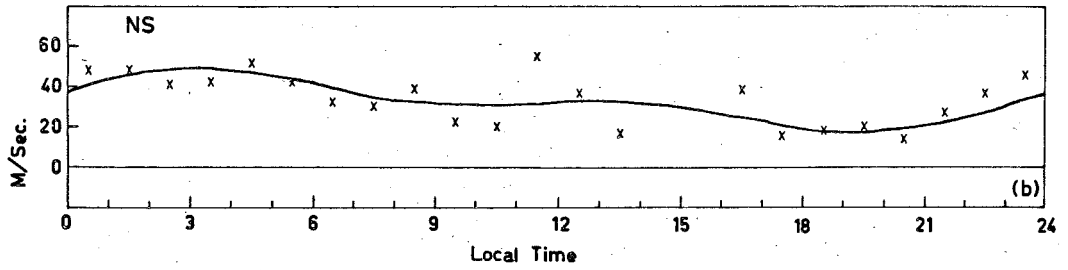
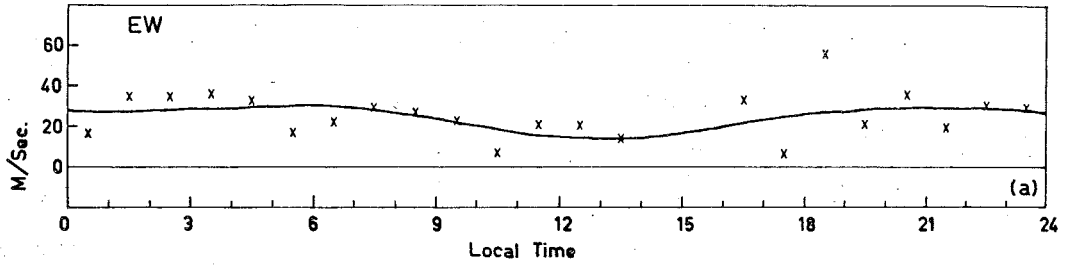
AUGUST, 17 - 24, 1961



a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94KM.

FIG. 57

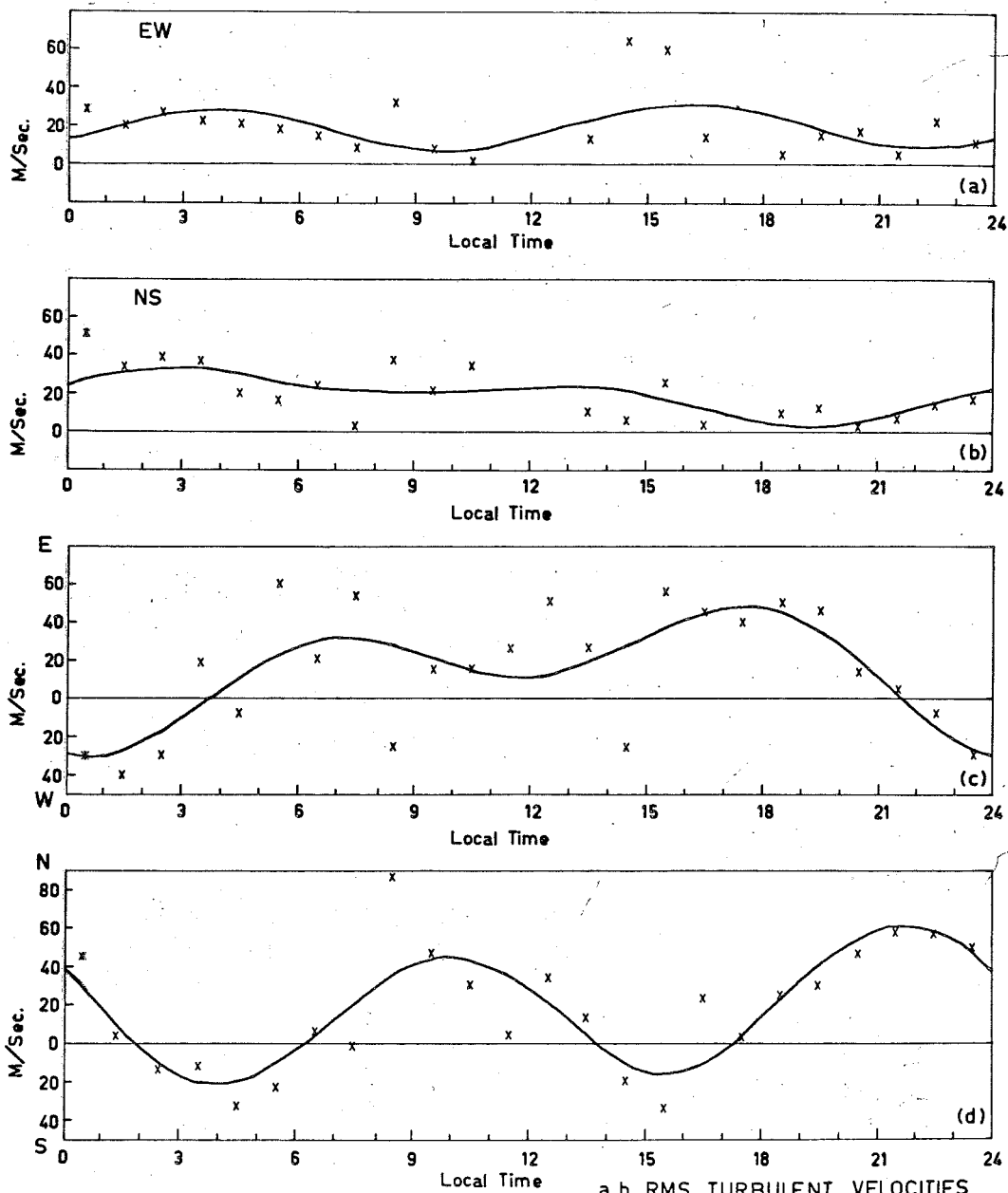
SEPTEMBER 1961



a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94 KM.

FIG. 58

DECEMBER 1961



a,b RMS TURBULENT VELOCITIES  
c,d MEAN WIND, 85-94 KM.

FIG. 59

10.5 Summary of the Seasonal Variations of the Parameters Specifying the Turbulent Flow Field.

A summary of the parameters of the turbulent flow field as determined for each month of the survey is given in Table I, under the following headings;

- $\epsilon$  Turbulent dissipation rate, ergs/gm/sec.
- $U_0$  Characteristic velocity of the energy bearing eddies, metres/sec.
- $L_0$  Characteristic scale of the energy bearing eddies, km.
- $U_z$  RMS vertical velocity, metres/sec.
- $L_z$  Vertical correlation distance, km.
- $L_\beta$  Characteristic scale of the buoyancy subrange, metres.
- $T_{\max}$  Maximum observed rate of reflection point motion along meteor trail, km/sec.
- $T_{\text{rms}}$  RMS value of
- $T_{\text{mean}}$  Arithmetic mean of
- $\xi_{\text{grad}}$  RMS velocity difference across 1km of trail, m/sec/km.
- $\Delta h_{\text{grad}}$  RMS vel. difference across 1km height diff., m/sec/km.
- $\lambda$  Characteristic dissipation length associated with the energy bearing eddies, km.
- $R_\lambda$  Turbulence Reynolds number.
- $N$  No. of meteors observed during each month.



TABLE I

	$\epsilon$	$U_0$	$L_0$	$U_z$	$L_z$	$L_\beta$	$T_{\max}$	$T_{\text{rms}}$	$T_{\text{mean}}$	$S_{\text{grad}}$	$\Delta h_{\text{grad}}$	$\lambda$	$R_\lambda$	N
DEC 60	300	25	95	10	6.7	600	-3.2	0.70	0.05	5.6	7.1	1.6	$4.2 \times 10^3$	104
JAN 61	365						-2.3	0.66	-0.04	5.8				88
FEB 61	450						-6.9	0.98	-0.00	6.4				77
MAR 61	600	30	66	13	6.8	900	-7.1	0.81	-0.09	7.0	8.6	1.5	$4.6 \times 10^3$	116
APR 61	350						4.6	0.71	0.13	5.8				56
MAY 61	276						5.0	0.68	-0.00	5.4				169
JN 61	309	27	111	10	6.8	700	2.6	0.61	0.00	5.6	7.5	1.9	$5.1 \times 10^3$	145
JY 61	416	23	50	12	5.4	600	4.3	0.66	0.00	6.2	7.4	1.4	$3.2 \times 10^3$	194
A(1-6)	530	33	121	12	8.1	800	3.4	0.74	0.00	6.7	7.8	1.9	$6.3 \times 10^3$	80
(17-24)	750	35	95	14	7.6	900	-7.3	0.85	0.04	7.5	9.0	1.5	$5.1 \times 10^3$	124
SEP 61	642	40	177	13	8.9	1km	-4.6	0.83	-0.06	7.2	9.0	1.9	$7.6 \times 10^3$	219
OCT 61	422						-3.1	0.73	-0.06	6.2				164
NOV 61	362						-2.1	0.63	-0.08	5.8				95
DEC 61	411	26	76	11	6.1	650	-4.0	0.74	-0.08	6.1	7.7	1.5	$3.9 \times 10^3$	107

## CHAPTER XI.

### Atmospheric Turbulence at Meteor Heights

#### 11.1 The Energy Bearing Eddies.

It has been shown in Chapters IX and X that the magnitude of the atmospheric turbulence encountered in the meteor region is time dependent; diurnal as well as seasonal variations are quite marked. While no direct observations have been made of the large scale turbulent motions, other than the determination of their characteristic velocities, their properties as predicted by extrapolation from the smaller scales are in good agreement with the observations made at Jodrell Bank. The observed monthly averages of the characteristic velocities of these eddies range from 20 m/sec. to 40 m/sec., and their predicted horizontal scales from some 60km to 200km; during individual months characteristic velocities as low as 15 m/sec. and as high as 60 m/sec. have been measured.

The vertical scale of these eddies has been predicted by extrapolation. The vertical correlation distance of 7km thus determined is the same as that found from other radio and

photographic observations of meteor trails, and also from the results of rocket experiments.

Whereas the large scale eddies have been shown to be a possible extension of the inertial subrange, their relationship to the mean flow at meteor heights has not been clearly established. These large scale motions may have as their origin some type of gravity wave, as has been proposed by Hines (1959) and Long (1959). Gossard (1962) has established that there is a mechanism whereby an upward escape of energy from the turbulent motions in the troposphere is possible, and that the gravity wave thus propagated will produce in the meteor region anisotropic "eddies" of the scale and velocity observed. However, there is some doubt that this energy loss from the troposphere is of a continuous nature, as would be required to sustain the always observable turbulence in the meteor region.

#### 11.2 The Small Scale Structure.

The turbulence investigation at Adelaide has concentrated particularly on scales less than 3km, since the eddies in this range have, in the past, been the subject of considerable controversy. It has been shown that the theories of homogeneous turbulence can be successfully applied to the data obtained from the radio echo observations of meteors, and that these results can be reconciled with the photographic

meteor and rocket results by assuming the presence of what has been termed a buoyancy subrange of eddy scales. The existence of such a subrange was first proposed by Bolgiano in 1959; his theory as applied by Dougherty (1961) to Greenhow's results, however, produced a range of scales less than 100 metres and did nothing to resolve the difficulties associated with the interpretation and reconciliation of radio and photographic observations. The approach adopted in Chapter IX Section 4 predicts that the characteristic eddy of the buoyancy subrange at 95km is simply given by

$$L_B = 25 U_0 \quad (9.4.1)$$

where  $U_0$  is the characteristic velocity of the energy bearing eddies of the Kolmogoroff turbulence spectrum. Since  $U_0$  varies from some 15 m/sec. to 60 m/sec., then

$$400 \text{ metres} < L_B < 1.5 \text{ km},$$

which is of the right scale to qualitatively explain the small scale anomaly.

An interesting comparison can be made with the results obtained by Rofe from rocket borne grenades (see Chapter III Section 6). The size of the "cells" responsible for the vertical fluctuations observed has been given by Rofe as 500 metres at 90km and 1km at 100km. If Rofe's "cells" are the characteristic eddies of the buoyancy subrange, then according

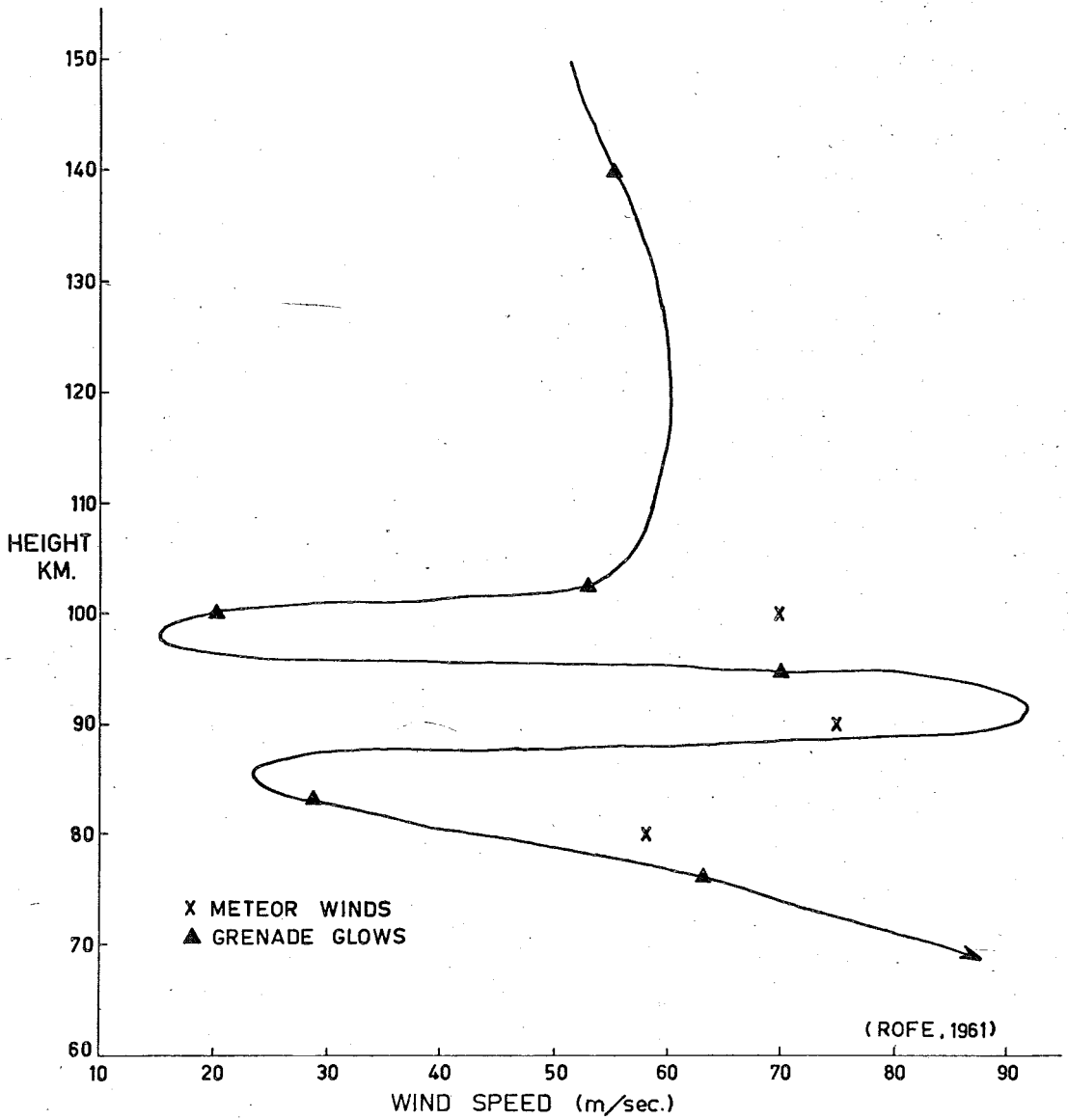


FIG. 60 - WIND SPEED PROFILE. WOOMERA, 17.11.60

to 9.4.1, they would be expected to be associated with turbulent winds of 20 m/sec. and 40 m/sec. respectively. The random wind components at the time of the firing from which Rofe's values are derived can be roughly estimated from Fig. 60 as the deviation of the measured windspeed profile from the mean wind determined by the meteor method. The relevant turbulent velocities are 15 m/sec. and 50 m/sec. The agreement with the predicted values is surprisingly good.

### 11.3 The Isotropic Inertial Region.

Once allowance is made for the buoyancy subrange, the velocity differentials closely follow the Kolmogoroff isotropic inertial spectrum up to the 3km separation limit imposed by the three station geometry. Since the extrapolation of this spectrum to 6km gives the observed RMS vertical velocity (Chapter IX Section 6) one would expect the isotropic inertial region to extend out to this scale. This is confirmed by the behaviour of the height correlation of Greenhow and Neufeld as treated in Chapter III Section 2, and by the results of Blamont and De Jager as revised by Zimmerman (Chapter III Section 7).

### 11.4 The Height Shear.

The velocity shear with height is one of the most interesting properties of the meteor region. The vertical correlation distance has been established almost conclusively as

being 6 to 7km. The most unusual feature of this correlation distance is the fact that it appears to define an actual, stable stratification of the region. Evidence of this comes from the high power pulse soundings of the lower ionosphere by Gregory (1961), who has shown that the maximum fluctuations in the electron density profile in the 55km to 95km region occur throughout the year at preferred heights approximately 10km apart, with the greatest fluctuation located at 80km. The reason for the stability of this stratification is as yet unknown.

The rocket investigations carried out by Rofe and others show a turbulence cutoff at just over 100km, which is probably due to the rapid increase in the kinematic viscosity above this level. Whitehead (1961) has explained the stratified nature of sporadic E ionization in terms of the occasional extension into the E region of the turbulent motions observed at meteor heights. Unfortunately, because of the rapid diffusion of meteor trails above 100km, no data pertaining to atmospheric motions at these heights is available from the Adelaide turbulence survey of 1961.

#### 11.5 The Turbulent Dissipation Energy $\epsilon$ .

In the dimensionally derived expression for the turbulent energy spectrum of the inertial subrange, viz.

$$E(\kappa) = \alpha \epsilon^{\frac{2}{3}} \kappa^{-\frac{5}{3}} \quad ( 2.6.5 )$$

the value of  $\alpha$  has been taken as  $\frac{2}{3}$  ( see Chapter II, Section 6). This leads via the velocity difference/separation relation of 2.10.3 to values for the turbulent dissipation rate  $\epsilon$  of from 300 to 600 ergs/gm/sec, with an average for the thirteen months December 1960 to December 1961 of 430 ergs/gm/sec ( Chapter X Section 2 ).

Since the completion of the main body of this work, an alternative expression for  $\epsilon$  has been derived from the two equations defining the characteristic eddy of the buoyancy subrange contained in Chapter II ( Section 12 ) and Chapter IX ( Section 4 ), viz.

$$Ri^* = \left[ \frac{L\beta}{\lambda} \right]^2 \frac{g}{T} \left[ \frac{dT}{dh} + \Gamma \right] \left[ \frac{d\bar{v}}{dh} \right]^{-2} \quad ( 2.12.2 )$$

and

$$\frac{1}{2}U_0^2 = L\beta^2 \frac{g}{T} \left[ \frac{dT}{dh} + \Gamma \right] \quad 11.5.1$$

The expression 2.12.2 differs from that proposed by Blamont and De Jager in that the characteristic dissipation length  $\lambda$  has been substituted for the momentum mixing length.

Batchelor gives

$$\epsilon = \frac{15\nu U_0^2}{\lambda^2} \quad ( 2.7.1 )$$



which yields

$$\lambda = \sqrt{\frac{15\nu}{\epsilon}} U_0 \quad 11.5.2$$

For the characteristic eddy of the buoyancy subrange  $Ri^* = 1$ , and on substitution of 11.5.2 for  $\lambda$  in 2.12.2

$$L_\beta = \sqrt{\frac{15\nu}{\epsilon}} U_0 \left[ \frac{g}{T} \left( \frac{dT}{dh} + \Gamma \right) \right]^{\frac{1}{2}} \frac{d\bar{v}}{dh} \quad 11.5.3$$

Rearrangement of 11.5.1 gives

$$L_\beta = \frac{U_0}{\sqrt{2}} \left[ \frac{g}{T} \left( \frac{dT}{dh} + \Gamma \right) \right]^{-\frac{1}{2}} \quad 11.5.4$$

Equating 11.5.3 and 11.5.4 we have

$$\sqrt{\frac{15\nu}{\epsilon}} \frac{d\bar{v}}{dh} = \frac{1}{\sqrt{2}}$$

i.e.

$$\epsilon_z = 30\nu \left( \frac{d\bar{v}}{dh} \right)^2 \quad 11.5.5$$

Since the height gradient is not linear, but follows the  $4/3$  power law of a turbulence subjected to shear ( Chapter X Section 3 ) the expression 11.5.5 cannot be regarded as providing an absolute value for  $\epsilon$ . The equation also relies on the existence of a buoyancy subrange. However, substitution of actual values does lead to some interesting results. Since the scale of the buoyancy subrange

averages approximately 700 metres, the gradient is taken as the velocity differential across this height difference. The average gradient measured across a height difference of 700 metres for the 13 months Dec. 1960 to Dec. 1961 is 9 metres/sec/km. The average echo height for the survey was 95km, at which height the kinematic viscosity is  $10^5$  cm<sup>2</sup>/sec.

Thus

$$\begin{aligned}\epsilon_z(\text{av}) &= 30 \times 10^5 \times (9 \times 10^{-3})^2 \\ &= 240 \text{ ergs/gm/sec.}\end{aligned}$$

If the  $\alpha$  of 2.6.5 is taken as 1.0 instead of  $\frac{2}{3}$ , substitution of the measured data in equation 2.10.3 leads to an average dissipation rate for the year of

$$\epsilon(\text{av}) = 270 \text{ ergs/gm/sec.}$$

The revised seasonal variation of the turbulent dissipation rate for the thirteen months of the Adelaide survey is presented in Fig. 61a. This plot has been produced by a more refined weighted mean velocity difference programme using the value  $\alpha = 1.0$ . The characteristics of the seasonal variation presented in Fig. 45 remain, but the turbulent dissipation rates for the low echo rate months have undergone some modification.

Also plotted are the values of  $\epsilon_z$  determined from the height shear across the scale characteristic of the buoyancy subrange for each month. With the exception of the values of

turbulent dissipation calculated for March and August 1st to 6th the seasonal variation follows a similar, although somewhat smoother, pattern. The comparison with the mean wind components of Fig. 61b serves to further substantiate the correlation between the turbulent dissipation rate and the 24 hour component of the mean wind at 90km.

A programme is being developed to determine an average mean wind speed weighted towards the high rate hours for the typical day of each month. Comparison of the seasonal variation of this wind speed with that of the turbulent dissipation rate should settle the question of whether or not the turbulent energy has its source in the mean flow.

Table II presents the revised parameters associated with the turbulent flow fields for each month.

#### 11.6 The Reynolds Number $R_\lambda$ :

The investigation of the significance of the variation in Reynolds Number is proceeding. However, the median value of  $7 \times 10^3$  is considered to be as good an estimate as any yet proposed for this region of the atmosphere.

---

Note re Fig. 61b :- The magnitudes of the 24 and 12 hour components have been taken as the average of the respective hour by hour values,\* not as the RMS of the maximum and minimum values, as was done in Fig. 45b.

\*magnitudes

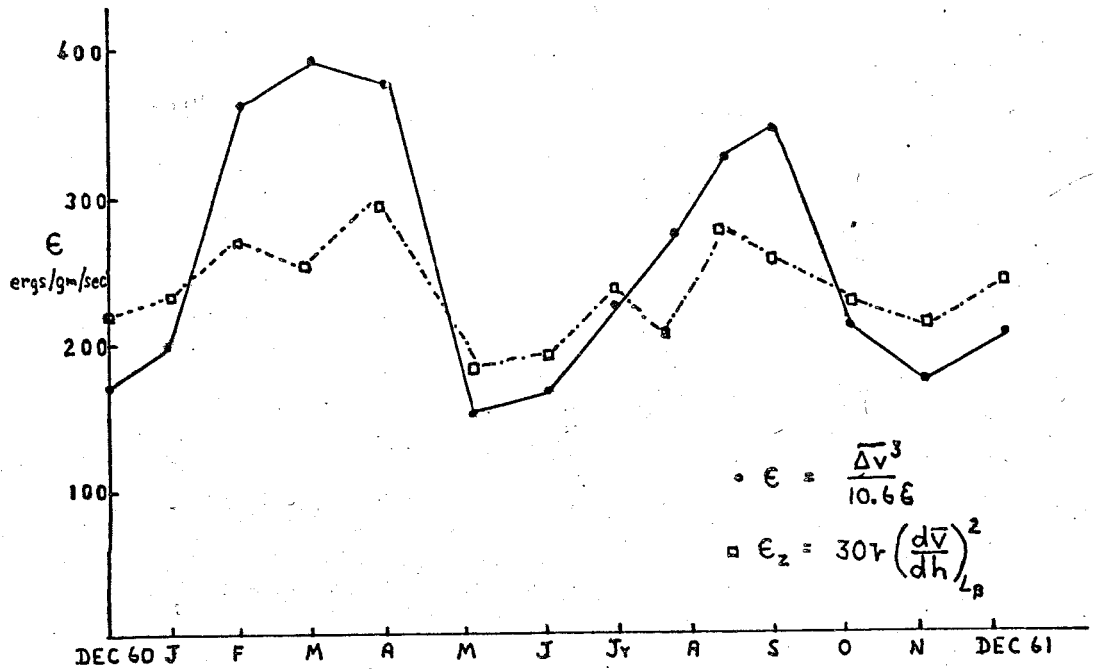


FIG. 61a SEASONAL VARIATION OF THE TURBULENT DISSIPATION RATE AT 90 KM

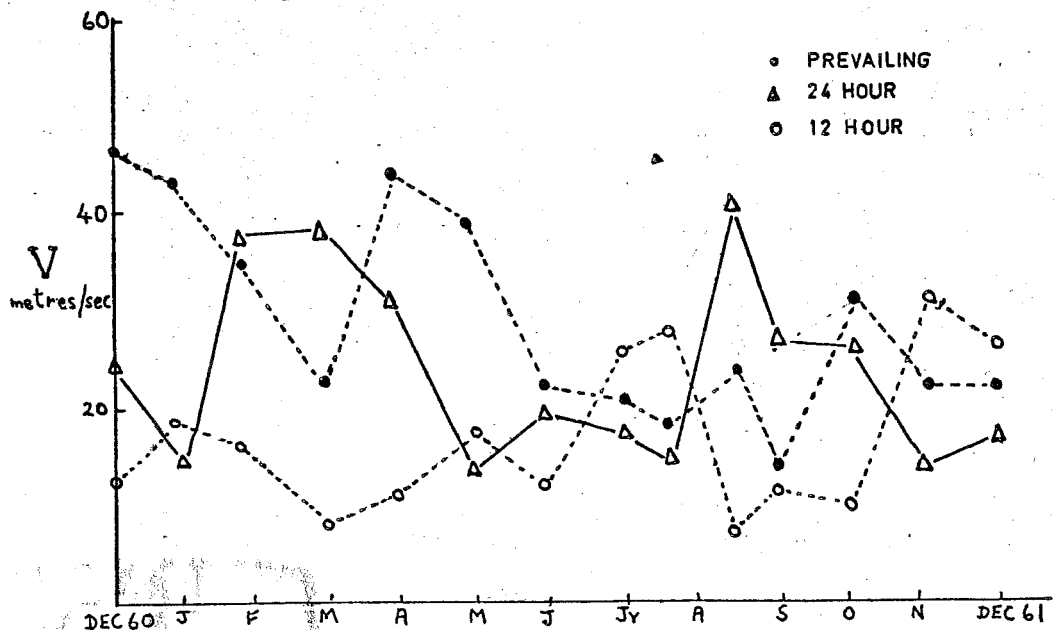


FIG. 61b COMPONENTS OF THE MEAN WIND, 85-94 KM

TABLE II

	$\epsilon$	$U_0$	$L_0$	$U_z$	$L_z$	$L_\beta$	$T_{\max}$	$T_{\text{rms}}$	$T_{\text{mean}}$	$\xi_{\text{grad}}$	$\Delta h_{\text{grad}}$	$\lambda$	$R_\lambda$	N
DEC 60	180	26	92	9	6.8	670	-3.2	0.70	0.05	5.6	7.2	2.1	$5.2 \times 10^3$	104
JAN 61	200	36	230	10	9.0	950	2.0	0.65	0.02	5.8	8.4	3.2	$1.1 \times 10^4$	88
FEB 61	360	30	74	11	6.1	790	-6.6	0.99	0.09	7.1	9.0	2.0	$6.0 \times 10^3$	77
MAR 61	398	31	70	12	6.8	800	-6.6	0.79	-0.05	7.3	8.5	1.9	$5.9 \times 10^3$	113
APR 61	363	28	55	10	5.4	720	4.6	0.75	0.12	7.1	8.9	1.8	$4.9 \times 10^3$	57
MAY 61	160	29	145	9	7.8	760	5.0	0.69	0.00	5.4	7.3	2.8	$8.2 \times 10^3$	169
JUN 61	187	27	99	9	6.8	700	2.6	0.61	0.00	5.7	7.4	2.4	$6.5 \times 10^3$	145
JULY 61	242	23	47	9	5.0	590	4.3	0.66	0.00	6.2	7.7	1.8	$4.1 \times 10^3$	194
A(1-6)	286	33	120	11	8.3	860	3.4	0.74	0.00	6.6	8.0	2.4	$8.0 \times 10^3$	80
(17-24)	326	34	118	11	6.9	890	-7.3	0.85	0.04	6.9	9.4	2.3	$8.1 \times 10^3$	124
SEP 61	345	40	177	12	8.5	1km	-4.6	0.83	-0.06	7.0	9.5	2.6	$1.0 \times 10^4$	219
OCT 61	207	32	145	10	7.8	820	-3.3	0.71	-0.04	5.9	7.9	2.7	$8.5 \times 10^3$	164
NOV 61	174	27	102	9	6.8	690	2.2	0.64	-0.05	5.6	7.3	2.5	$6.6 \times 10^3$	95
DEC 61	214	26	78	9	5.9	680	-4.0	0.74	-0.08	6.0	7.9	2.2	$5.7 \times 10^3$	107

Legend as for Table I, P178.

### The Future

As presented in the foregoing chapters, the treatment of the data obtained from the 1961 survey is by no means exhaustive. An attempt has been made to overcome the fundamental difficulty of ascribing absolute values to the turbulent dissipation rate, but this relies heavily on the existence of the proposed buoyancy subrange. A correlation analysis which endeavours to derive the characteristic scale of the buoyancy subrange from the measured velocity shears is proceeding. However, the rocket borne grenade experiment would seem to be the best for direct observation of the buoyancy subrange, and analysis of more data from this source could prove profitable.

As has been mentioned in Chapter IV Section 5 the decay time of meteor echoes can be used to measure the coefficient of ambipolar diffusion at meteor heights. Such measurements of diffusion coefficient show considerable scatter which, to date, has not been explained satisfactorily. Pressure fluctuations arising from the turbulent shears could contribute to this scatter.

An attempt is being made to apply the Hines/Long theory of gravitational waves to the motions observed during the 1961 survey. The energy spectrum predicted by Hines is generally similar to that observed, but the discrepancies are more readily explicable in terms of the Kolmogoroff spectrum

modified by the influence of shearing and buoyancy forces.

A further investigation of the random motions at these heights is contemplated, using much higher transmitter power. A pulsed transmitter with a measured peak power output of 40KW has already been built, and a 1.5KW CW transmitter is in the process of construction. At this power level, the echo rate will be several times higher than that of the 1961 survey; this should enable several velocity difference/separation histograms to be produced each "typical" day, and provide further insight into the diurnal variations of the turbulence parameters.

The use of highly directive antennae is also contemplated, since these will allow direct measurement of the time constant of the energy bearing eddies and simultaneous spacial correlations to be determined. It is proposed to incorporate a fourth station at much greater separation to verify the isotropy of the eddies of scale 6 to 7km. The results obtained from this project should also determine whether or not the seasonal change in the vertical scale is significant.

The fact that the mean flow is latitude dependent has been established by simultaneous observations using the meteor method carried out at Mawson ( $68^{\circ}$ S), Adelaide ( $35^{\circ}$ S) and Jodrell Bank ( $53^{\circ}$ N). Three-station measurements have also been made at Jodrell Bank, but not on a seasonal basis.

Under the Harvard Meteor Project a multi-station orbit survey is in progress, but unfortunately no provision has been made for wind measurement. Several multi-station experiments carried out simultaneously at different latitudes would significantly increase our knowledge of this very interesting region of the Upper Atmosphere.



APPENDIX IIBM 1620 FORTRAN programmes for Mean Wind Reduction.

List of input/output variables as they appear in  
Programme 1190A.

ERORX, ERORC	Maximum allowable errors between traces as read from Mean Wind Film.
SCALE	Allows for camera and film reader optics in conversion of distance between adjacent doppler maxima to line of sight trail drift velocity
MY	Year
MO	Month
JO	Day
LH	Local time, hours
LM	Local time, mins
UR	Echo number
KM	Midpoint echo range (St. Kilda) as read from film
WAVE	Distance between adjacent doppler maxima
MA, MB, MC, LA, LB, LC, LD	Distances measured between doppler maxima on the DF traces, subsequently used in the direction finding routine.
ML	Checks that input data is of the correct format.

LHLM Local time, hours and mins

EL, EM Direction cosines of reflection point from St. Kilda, Adelaide-St. K as Y axis.

EL3, EM3 Midpoint direction cosines, including aerial corrections and rotation to NS-EW axes

KHT Echo height, km.

LEVEL Height group, 1 75-84km  
2 85-94km  
3 95-104km

LT Local mean solar time, hours

NFL EW drift responsible for line of sight component

NFM NS ditto.

MVEL, VEL Measured line of sight drift velocity

MCS, MX Errors involved in determination of EL, EM

This is the first of three programmes designed to convert the data read from the mean wind recording display film to a mean, diurnal and semidiurnal components of the wind, at meteor heights. Data is processed in monthly batches, and the final result ( output of 1190C ) gives the winds for a typical day of each month. This first programme involves processing card by card, each input card producing one output card.

```

C   1190A UNIVERSITY ADELAIDE ROPER 313
C   UPPER ATMOSPHERE WIND ANALYSIS,           (METEOR GROU
P)
C   FILM READER DATA PROCESSOR, PUNCHES CARDS, PRINTS ERRORS
.
  READ270,ERORX,ERORC
270  FORMAT(F4.2,F4.2)
     READ272,SCALE
272  FORMAT(F6.0)
     8 READ11
     11 FORMAT(31H           )
        PRINT11
256  READ210,MY,MO,JO,LH,LM,UR,KM,WAVE,MA,MB,MC,MD,LA,LB,LC,L
D,ML
210  FORMAT(12,13,13,13,12,F6.0,14,F5.0,14,14,14,14,14,14,14
,14,12)
     LHLM=100*LH+LM
     IF(MO)301,10,262
     10 PRINT43
     43 FORMAT(41HLOAD MORE DATA, SET PAGE, THEN PUSH START)
        PAUSE
        GOTO8
301  PRINT302
302  FORMAT(1H 20HEND OF METEOR DATA  )
        STOP
262  MB=MB+LA-LB
     MC=MC+LA-LC
     MD=MD+LA-LD
     A=MA
     B=MB
     C=MC
     D=MD
     VEL=SCALE/WAVE
     ALAMB=ABSF(WAVE)
     AL=A/ALAMB
     BL=B/ALAMB
     CL=C/ALAMB
     DL=D/ALAMB
204  IF(AL-1.0)201,202,202
202  AL=AL-1.0
     GOTO204
201  IF(BL-1.0)205,203,203
203  BL=BL-1.0
     GOTO201
205  IF(CL-1.0)207,206,206
206  CL=CL-1.0
     GOTO205

```

```
207 IF(DL-1.0)203,209,209
209 DL=DL-1.0
GOTO207
203 CONTINUE
IF(CL-DL)211,212,212
211 XL=1.0+CL-DL
GOTO213
212 XL=CL-DL
213 CONTINUE
MASS=VEL
MVEL=VEL
EL=-AL
EM=-BL
XLC=(EL/2.0)-(EM/2.0)+0.5
NEXT=1
99 NIX=0
X=XL
X1=XLC
ELL=ERORX
100 CONTINUE
102 IF(X)101,103,104
101 X=X+1.0
GOTO102
104 IF(X-1.0)103,103,105
105 X=X-1.0
GOTO104
103 IF(X1)106,107,103
106 X1=X1+1.0
GOTO103
103 IF(X1-1.0)107,107,109
109 X1=X1-1.0
GOTO108
107 IF(ABSF(X-X1)-ELL)115,115,112
112 IF(ABSF(X-X1+1.0)-ELL)116,116,113
113 IF(ABSF(X-X1-1.0)-ELL)117,117,118
115 AMISS=X-X1
GOTO110
116 AMISS=X-X1+1.0
GOTO110
117 AMISS=X-X1-1.0
GOTO110
118 AMISS=X-X1
GOTO111
110 MZ=+1
GOTO114
111 MZ=-1
114 AMISS=AMISS*100.0
```

```

MISS=AMISS
IF(NIX)499,499,50
499 XL=X
    XLC=X1
    MXZ=MZ
    MX=MISS
    GOT0501
500 CL=X
    CLC=X1
    MCZ=MZ
    MCS=MISS
501 GOTO(214,219,229,237,243,251),NEXT
214 NSTAT=1
    IF(MXZ)216,215,217
216 EL=1.0-AL
    XLC=(EL/2.0)-(EM/2.0)+0.5
    NEXT=2
    GOT099
219 NSTAT=2
    IF(MXZ)218,215,221
217 IF(MASS)224,215,225
224 EL=AL
    EM=BL
    CLC=0.75-(1.25*EM)
    GOT0314
225 CLC=(1.25*EM)+0.25
314 NEXT=3
98  NIX=1
    X=CL
    X1=CLC
    ELL=ERORC
    GOT010
229 NSTAT=3
    IF(MCZ)226,215,238
226 IF(MASS)230,215,232
230 EL=AL-1.0
    EM=BL-1.0
    CLC=0.75-(1.25*EM)
    GOT0313
232 EL=1.0-AL
    EM=1.0-BL
    CLC=(1.25*EM)+0.25
313 NEXT=4
    GOT098
237 NSTAT=4
    IF(MCZ)234,215,238
221 IF(MASS)239,215,241

```

```

239 EL=AL-1.0
    EM=BL
    CLC=0.75-(1.25*EM)
    GOT0242
241 EL=1.0-AL
    EM=-BL
    CLC=(1.25*EM)+0.25
242 NEXT=5
    GOT098
243 NSTAT=5
    IF(MCZ)245,215,238
245 IF(MASS)247,215,249
247 EL=AL
    EM=BL-1.0
    CLC=0.75-(1.25*EM)
    GOT0250
249 EL=-AL
    EM=1.0-BL
    CLC=(1.25*EM)+0.25
250 NEXT=6
    GOT098
251 NSTAT=6
    IF(MCZ)252,215,238
238 CONTINUE
    RANGE=KM
    EL1=EL+0.04
    EM1=EM
    ZENTH=(EL1**2)+(EM1**2)
    IF(0.95-ZENTH)312,255,255
255 CONTINUE
    EL2=EL1*(1.0-EM1*(5.725/RANGE))
    EM2=EM1+(1.0-EM1**2)*(5.725/RANGE)
    EL3=(EL2*0.9774)-(EM2*0.2115)
    EM3=(EL2*0.2115)+(EM2*0.9774)
    EN3=SQRTF(1.0-EL3**2-EM3**2)
    HITE=2.0*RANGE*EN3
    IF(LM-45)1,2,2
  1 LT=LH
    GOT09
  2 LT=LH+1
  9 NSTAT=7
    IF(LT)215,3,4
  3 LT=LT+24
  4 KHT=HITE
    IF(KHT-75)220,12,12
 12 IF(KHT-85)14,15,15
 14 LEVEL=1

```

```

      GOTO13
15  IF(KHT-95)16,17,17
16  LEVEL=2
      GOTO13
17  IF(KHT-105)18,220,220
18  LEVEL=3
13  EL2M2=EL3**2+EM3**2
      E2L2M=1.0/EL2M2
      FL=-(VEL*EL3*E2L2M)
      FM=-(VEL*EM3*E2L2M)
      NFL=FL+0.5
      NFM=FM+0.5
PUNCH306,UR,MY,MO,JO,LHLM,EL,EM,EL3,EM3,KHT,LEVEL,LT,NFL,NFM,M
VEL,MCS,MX
306 FORMAT(F7.0,13,13,13,15,F5.2,F5.2,F5.2,F5.2,14,13,13,15,1
5,15,13,13)
      GOTO256
215 PRINT273,UR,NSTAT,LHLM
273 FORMAT(F7.0,16HFAULT FOR NSTAT=13,8H, TIME =15,4H HRS)
      GOTO256
218 PRINT275,UR,LHLM
275 FORMAT(F7.0,14HNO CHECK FOR X,8H, TIME =15,4H HRS)
      GOTO256
220 PRINT274,UR,KHT,LHLM
274 FORMAT(F7.0,25HHEIGHT UNACCEPTABLE. HT =14,10HKM, TIME
=15,4H HRS)
      GOTO256
234 PRINT280,UR,LHLM
280 FORMAT(F7.0,20HNO CHECK FOR C,L1,M1,8H, TIME =15,4H HRS)

      GOTO256
252 PRINT285,UR,LHLM
285 FORMAT(F7.0,20HNO CHECK FOR C,L2,M2,8H, TIME =15,4H HRS)

      GOTO256
312 PRINT286,UR,LHLM
286 FORMAT(F7.0,1X2OHL2+M2 EXCEEDS 0.95 ,8H, TIME =15,4H HR
S)
      GOTO256
      END

```

## Programme 1190B

The cards punched by 1190A are fed into this programme, and processed to produce a least squares fit wind for each of the 24 hours at each level. Prints output, and punches cards for processing by 1190C.

VCOS(LEVEL,LT) EW component for LEVEL at time LT

VSIN(LEVEL,LT) NS ditto.

MNO(LEVEL,LT) Number of meteors used to determine above.

## Programme 1190C

Performs a Fourier Analysis of the winds punched out by 1190B to produce mean, diurnal and semidiurnal wind components for both NS and EW directions.

NGO Control indicating type of data following

YE(J) EW wind component for hour J

YN(J) NS ditto.

AE Prevailing component

COEFE(1) Amplitude of 24 hour component

COEFE(2) Ditto for 12 hour

ALPHE(1) Phase of 24 hour component

ALPHE(2) Ditto for 12 hour

A1, A2 EW 24hr, 12hr component at given time T

B1, B2 NS ditto

Y Amplitude of the mean wind at time T



```

C 1190B UNIVERSITY ADELAIDE ROPER 313
C UPPER ATMOSPHERE WIND ANALYSIS, (METEOR GRO
UP)
C PROCESSES OUTPUT CARDS FROM 1190A TO PRODUCE WINDS (PRI
NTS OUTPUT)
DIMENSIONSUMC2(3,24),SUMS2(3,24),SUMSC(3,24),SUMFL(3,24
)
DIMENSIONSUMFM(3,24),MNO(3,24),VCOS(3,24),VSIN(3,24)
READ4
4 FORMAT(30H )
DO26LEVEL=1,3
DO26LT=1,24
SUMC2(LEVEL,LT)=0
SUMS2(LEVEL,LT)=0
SUMSC(LEVEL,LT)=0
SUMFL(LEVEL,LT)=0
SUMFM(LEVEL,LT)=0
26 MNO(LEVEL,LT)=0
1READ306,UR,MY,MO,JO,LHLM,EL,EM,EL3,EM3,KHT,LEVEL,LT,NFL,NFM,
MVEL,MCS,MX
306 FORMAT(F7.0,13,13,13,15,F5.2,F5.2,F5.2,F5.2,14,13,13,15,1
5,15,13,13)
IF(UR)8,8,13
13 EL2M2=EL3*EL3+EM3*EM3
E2L2M=1.0/EL2M2
FL=NFL
FM=NFM
C2=EL3*EL3*E2L2M
S2=EM3*EM3*E2L2M
SC=EL3*EM3*E2L2M
SUMC2(LEVEL,LT)=SUMC2(LEVEL,LT)+C2
SUMS2(LEVEL,LT)=SUMS2(LEVEL,LT)+S2
SUMSC(LEVEL,LT)=SUMSC(LEVEL,LT)+SC
SUMFL(LEVEL,LT)=SUMFL(LEVEL,LT)+FL
SUMFM(LEVEL,LT)=SUMFM(LEVEL,LT)+FM
MNO(LEVEL,LT)=MNO(LEVEL,LT)+1
GOTO1
8 DO27LEVEL=1,3
PRINT2
2 FORMAT(1X////44H SET PAGE FOR NEXT LEVEL, THEN PUSH S
TART./)
PAUSE
PRINT4
PRINT34
34 FORMAT(1X////40H LT LEVEL VCOS EAST VSIN NORTH METEOR
S//)

```

```

D027LT=1,24
IF(MNO(LEVEL,LT)-1)35,35,25
25 P=1.0/((SUMSC(LEVEL,LT)**2)-(SUMC2(LEVEL,LT)*SUMS2(L
EVEL,LT)))
VSIN(LEVEL,LT)=P*(SUMFL(LEVEL,LT)*SUMSC(LEVEL,LT))
VSIN(LEVEL,LT)=VSIN(LEVEL,LT)-P*SUMFM(LEVEL,LT)*SUMC2(L
EVEL,LT)
VCOS(LEVEL,LT)=P*SUMFM(LEVEL,LT)*SUMSC(LEVEL,LT)
VCOS(LEVEL,LT)=VCOS(LEVEL,LT)-P*SUMFL(LEVEL,LT)*SUMS2(L
EVEL,LT)
GOTO28
35 VCOS(LEVEL,LT)=999.0
VSIN(LEVEL,LT)=999.0
28 PUNCH42,LT,LEVEL,VCOS(LEVEL,LT),VSIN(LEVEL,LT),MNO(LEVE
L,LT)
27 PRINT42,LT,LEVEL,VCOS(LEVEL,LT),VSIN(LEVEL,LT),MNO(LEVE
L,LT)
42 FORMAT(14,13,6XF5.0,5XF5.0,4X14)
STOP
END

```

C 1190C UNIVERSITY OF ADELAIDE ROPER 313  
 C FOURIER ANALYSIS, 24 POINT INPUT MEAN PLUS 2 HARM  
 ONICS

DIMENSIONYE(24),YN(24)  
 DIMENSIONSE(12),DE(12),DIFE(10),SUME(4),COEFE(2),ALPHE(

2)

```

100 READ101,NGO
101 FORMAT(16)
    IF(NGO)19,21,20
    19 STOP
    21 PRINT22
    22 FORMAT(25HNEW PAGE, THEN PUSH START)
    PAUSE
    20 READ 20
200 FORMAT(38H
    D0201J=1,24
201 READ202,LEVEL,YE(J),YN(J)
202 FORMAT(4X13,6XF5.0,5XF5.0)
    IF(LEVEL-2)1,2,3
    1 LOW=75
    LHI=84
    GOT04
    2 LOW=85
    LHI=94
    GOT04
    3 LOW=95
    LHI=104
    4 NEXT=0
    GO TO 30
31 D032J=1,24
32 YE(J)=YN(J)
    NEXT=1
    PRINT22
    PAUSE
30 PRINT5,LOW,LHI
    5 FORMAT(10X34HFOURIER ANALYSIS FOR HEIGHT RANGE 14,2H -I
4,3H KM)
    PRINT20
    D06J=1,11
    K=J+12
    SE(J)=YE(J)+YE(K)
    6 DE(J)=YE(J)-YE(K)
    SE(12)=YE(24)+YE(12)
    DE(12)=YE(24)-YE(12)
    AE=0
    D07J=1,12

```

```

7 AE=AE+SE(J)
  AE=AE/24.
  DO8J=1,5
  K=12-J
  DIFE(J)=DE(J)-DE(K)
8 DIFE(J+5)=DE(J)+DE(K)
  SUME(1)=SE(1)-SE(5)-SE(7)+SE(11)
  SUME(2)=SE(2)-SE(4)-SE(8)+SE(10)
  SUME(3)=SE(1)+SE(5)-SE(7)-SE(11)
  SUME(4)=SE(2)+SE(4)-SE(8)-SE(10)
  AE24=DE(12)+0.966*DIFE(1)+0.866*DIFE(2)+0.707*DIFE(3)+0
.5*DIFE(4)
  AE24=(AE24+0.259*DIFE(5))/12.
  BE24=DE(6)+0.259*DIFE(6)+0.5*DIFE(7)+0.707*DIFE(8)+0.86
6*DIFE(9)
  BE24=(BE24+0.966*DIFE(10))/12.
  AE12=(SE(12)-SE(6)+0.866*SUME(1)+0.5*SUME(2))/12.
  BE12=(SE(3)-SE(9)+0.5*SUME(3)+0.866*SUME(4))/12.
  COEFE(1)=SQRTF(AE24*AE24+BE24*BE24)
  COEFE(2)=SQRTF(AE12*AE12+BE12*BE12)
  ALPHE(1)=ATANF(ABSF(BE24/AE24))
  ALPHE(2)=ATANF(ABSF(BE12/AE12))
  PI=3.1416
  IF(AE24)40,40,41
41 IF(BE24)43,43,42
43 ALPHE(1)=2.*PI-ALPHE(1)
  GOTO42
40 IF(BE24)44,44,45
44 ALPHE(1)=ALPHE(1)+PI
  GOTO42
45 ALPHE(1)=PI-ALPHE(1)
42 IF(AE12)50,50,51
51 IF(BE12)53,53,52
53 ALPHE(2)=2.*PI-ALPHE(2)
  GOTO52
50 IF(BE12)54,54,55
54 ALPHE(2)=ALPHE(2)+PI
  GOTO52
55 ALPHE(2)=PI-ALPHE(2)
52 IF(NEXT)34,34,35
34 PRINT11
11 FORMAT(10X25HEAST-WEST WIND COMPONENTS//)
  GOTO36
35 PRINT16
16 FORMAT(10X27HNORTH-SOUTH WIND COMPONENTS//)
36 PRINT9,AE

```

```
9  FORMAT(10X3HY =F7.1)
   ALPHE(1)=ALPHE(1)*(12./PI)
   PRINT10,COEFE(1),ALPHE(1)
10  FORMAT(13XF7.1,13H*COS(PI/12(T-F6.2,2H ))
   ALPHE(2)=ALPHE(2)*(6./PI)
   PRINT18,COEFE(2),ALPHE(2)
18  FORMAT(13XF7.1,12H*COS(PI/6(T-F6.2,2H )///)
   PRINT12
12  FORMAT(10X48HTIME A24COST B24SINT A12COS2T B12SIN2T MEA
N   Y//)
   D0301J=1,24
   T=J
   A1=AE24*COSF((PI/12.)*T)
   B1=BE24*SINF((PI/12.)*T)
   A2=AE12*COSF((PI/6.)*T)
   B2=BE12*SINF((PI/6.)*T)
   Y=AE+A1+B1+A2+B2
301 PRINT302,J,A1,B1,A2,B2,AE,Y
302 FORMAT(10X13,2XF6.1,2XF6.1,2XF6.1,3XF6.1,F7.1,F7.1)
   IF(NEXT)31,31,10
   END
```

APPENDIX IIIBM FORTRAN programmes for the Turbulence Reduction.

Programme E5 200 ( written in 7090 FORTRAN )

Reads output of Orbit Programme from tape, and punches data required for Turbulence Analysis.

NUMBRI	Echo number
MYEARI	Year
MONTHI	Month
JOURI	Day
LMSTHI	Local mean solar time, Hours
LMSTMI	Local mean solar time, Mins
IWAZ	Azimuth of reflection point from St. Kilda
IWZ	Zenith angle of ditto
IWH	Echo height
IRAZ	Azimuth of meteor radiant
IRZ	Zenith angle of ditto
IU(I)	Line of sight component of trail drift as observed from station I
IEPS1	Separation of reflection points, St. K to Sheedys farm
IEPS2	Ditto, St. Kilda to Direk
IRA	Right ascencion of meteor radiant
IDEC	Declination of ditto
IVEL	Velocity of the meteor particle

```

* E5 200 UNIVERSITY OF ADELAIDE ROPER 313
* DATE 13/8/62
* EXEQ TIME 19 MINS
* XEQ
  DIMENSION JUMP(3), ITEM(3), MOVE(3), DEVN(3), VELOC(3), TIME
(3),
  1WIND(3), TOPHAS(3)
  DIMENSION NUMBRI(1500), MYEARI(1500), MONTHI(1500), JOURI(
1500),
  1 LMSTMI(1500), IWAZ(1500), IWZ(1500), IWH(1500), IRAZ
(1500), IRZ
  2(1500), IU(3,1500), IEPS1(1500), IEPS2(1500), L
MSTHI(1500)
  DIMENSION IVEL(1500), IRA(1500), IDEC(1500)
  COMMON NUMBRI, MYEARI, MONTHI, JOURI
  PRINT 1
  1 FORMAT(1X56H THIS PROGRAMME PUNCHES APPROXIMATELY 2000
OUTPUT CARD
  1S./1X60H LOAD UNI TAPE NILSSON NO. 2. ONTO B7. PUSH STA
RT WHEN REA
  2DY////////)
  PAUSE
  CALL DNSHI(17)
  REWIND 17
  NSTART=7416
  NSTOP=76459
  2 I=0
  3 READ TAPE 17, NUMBR, MYEAR, MONTH, JOURA, LMSTH, LMSTM, JUMP, I
TEM, MOVE,
  1DEVN, VELC, VELOC, TIME, WH, SKIP, NWOP, WZ, WAZ, RZ, RAZ, EPS1, EP
S2, TLAM,
  2TMU, TLAM3, TMU3, RNEW, HAR, RA, DEC, VEL, DRAG, VGE, NQ, WIND, SMA
, E, EYE, AGP,
  3ANL, TAN, RLG, RLT, TLG, TLT, ALG, AEL, TEL, VHE, LENGTH, TOPHAS
  IF(NUMBR-NSTART)3,5,3
  4 READ TAPE 17, NUMBR, MYEAR, MONTH, JOURA, LMSTH, LMSTM, JUMP, I
TEM, MOVE,
  1DEVN, VELC, VELOC, TIME, WH, SKIP, NWOP, WZ, WAZ, RZ, RAZ, EPS1, EP
S2, TLAM,
  2TMU, TLAM3, TMU3, RNEW, HAR, RA, DEC, VEL, DRAG, VGE, NQ, WIND, SMA
, E, EYE, AGP,
  3ANL, TAN, RLG, RLT, TLG, TLT, ALG, AEL, TEL, VHE, LENGTH, TOPHAS
  IF(NUMBR-NSSTOP)5,6,5
  5 I=I+1
  NUMBRI(I)=NUMBR
  MYEARI(I)=MYEAR
  MONTHI(I)=MONTH

```

```

JOURI(1)=JOURA
LMSTHI(1)=LMSTH
LMSTMI(1)=LMSTM
IWAZ(1)=WAZ+0.5
IWZ(1)=WZ+0.5
IWH(1)=WH+0.5
IRAZ(1)=RAZ+0.5
IRZ(1)=RZ+0.5
IU(1,1)=WIND(1)
IU(2,1)=WIND(2)
IEPS1(1)=1000.0*EPS1
IEPS2(1)=1000.0*EPS2
IU(3,1)=WIND(3)
IRA(1)=RA
IDEC(1)=DEC+SIGNF(0.5,DEC)
IVEL(1)=VEL+0.5
GOTO4
6 NI=I+1
  NUMBRI(1)=NUMBR
  MYEARI(1)=MYEAR
  MONTHI(1)=MONTH
  JOURI(1)=JOURA
  LMSTHI(1)=LMSTH
  LMSTMI(1)=LMSTM
  IWAZ(1)=WAZ+0.5
  IWZ(1)=WZ+0.5
  IWH(1)=WH+0.5
  IRAZ(1)=RAZ+0.5
  IRZ(1)=RZ+0.5
  IU(1,1)=WIND(1)
  IU(2,1)=WIND(2)
  IU(3,1)=WIND(3)
  IEPS1(1)=1000.0*EPS1
  IEPS2(1)=1000.0*EPS2
  IRA(1)=RA
  IDEC(1)=DEC+SIGNF(0.5,DEC)
  IVEL(1)=VEL+0.5
  DO7I=1,NI
7 PUNCH8,NUMBRI(1),MYEARI(1),MONTHI(1),JOURI(1),LMSTHI(1)
,LMSTMI(1),
1IWAZ(1),IWZ(1),IWH(1),IRAZ(1),IRZ(1),IU(1,1),IU(2,1),IU
(3,1),
2IEPS1(1),IEPS2(1),IRA(1),IDEC(1),IVEL(1)
8 FORMAT(1X15,4I3,12,8I4,2I5,2I3,14)
N=N+1
IF(N-2)10,9,11

```



210

10 NSTART=76734  
NSTOP=76448  
GOTO2  
9 NSTART=47363  
NSTOP=63666  
GOTO2  
11 CALLEXIT  
END

```

C      1200 UNIVERSITY OF ADELAIDE ROPER 313
C      PROCESSES OUTPUT 7090 200 TO PRODUCE INPUT FOR 1200A

C      CONVERTS LINE OF SIGHT AND RADIANT AZIMUTHS AND ZENITH
C      ANGLES TO
C      DIRECTION COSINES ( L, EAST.  M, NORTH.)
1READ8,UR,MY,MO,JO,LH,LM,WAZ,WZ,WH,RAZ,RZ,IU1,IU2,IU3,EPS1,EP
S2
      8 FORMAT(F6.0,4I3,12,2F4.0,F4.0,2F4.0,3I4,2F5.3)
      1F(UR)3,4,5
      4 PRINT6
      6 FORMAT(9HMORE DATA)
      PAUSE
      GOTO1
      5 LTHM=100*LH+LM
      IWH=WH+0.5
      D1=11.45
      R=WH/COSF(WZ/57.3)
      RL=-SINF(WZ /57.3)*SINF(WAZ /57.3)
      RM=SINF(WZ /57.3)*COSF(WAZ /57.3)
      R1=SQRTF(R*R+D1*D1-2.0*RL*R*D1*0.2115+2.0*RM*R*D1*0.977

4)
      M3=1000.0*(RM*R+D1*0.9774)/R1
      L3=1000.0*(RL*R-D1*0.2115)/R1
      LR=-SINF(RZ/57.3)*SINF(RAZ/57.3)*1000.0
      MR=SINF(RZ/57.3)*COSF(RAZ/57.3)*1000.0
PUNCH9,UR,MY,MO,JO,LTHM,L3,M3,IWH,LR,MR,IU1,IU2,IU3,EPS1,EPS
2
      9 FORMAT(F7.0,3I3,15,5I4,3I4, 2F6.3)
      GOTO1
      3 STOP
      END

```

```

C      1200A UNIVERSITY OF ADELAIDE PHYSICS DEPT. ROPER 31
3
C      PROCESSES OUTPUT OF 1200 TO PUNCH INPUT TO 1200B
C      SUBTRACTS MEAN WIND COMPONENTS FROM MEASURED LINE OF SI
GHT DRIFTS
C      TO GIVE TURBULENT VELOCITIES.
C      REJECTS ECHOES WITH REFLECTION POINT HEIGHTS LYING OUTS
IDE THE
C      RANGE 74KM - 106KM.
C      INCORPORATES OPTIONAL PRINTOUT OF MEASURED AND MEAN WIN
DS OF
C      OPPOSITE SIGN.                                SWITCH 1 ON IGNORES PR
INT OUT
      DIMENSIONV(24),W(12)
      DIMENSIONVE1(24),VE2(24),VE3(24),VN1(24),VN2(24),VN3(24
)
      DIMENSIONH(3),DELH(3),X(3),U(3),UT(3),UMEAN(3)
10 READ10
100 FORMAT(48H
)
      PRINT10
      D049J=1,6
READ7,V(1),V(2),V(3),V(4),V(5),V(6),V(7),V(8),V(9),V(10),V(11
),V(12)
      7 FORMAT(12F4.0)
READ7,W(1),W(2),W(3),W(4),W(5),W(6),W(7),W(8),W(9),W(10),W(11
),W(12)
      D048K=1,12
48 V(K+12)=W(K)
      GOTO(30,31,32,33,34,35),J
30 D050K=1,24
50VE1(K)=V(K)
      GOTO49
31 D051K=1,24
51 VE2(K)=V(K)
      GOTO49
32 D052K=1,24
52VE3(K)=V(K)
      GOTO49
33 D053K=1,24
53 VN1(K)=V(K)
      GOTO49
34 D054K=1,24
54 VN2(K)=V(K)
      GOTO49
35 D055K=1,24
55 VN3(K)=V(K)
49 CONTINUE

```

```

1 READ2,UR,MY,MO,JO,LT,TL,TM,WH,RL,RM,U(1),U(2),U(3),X(1),X(2
)
2 FORMAT(F7.0,3I3,15,2F4.3,F4.0,2F4.3,3F4.0,2F6.3)
IF(UR)3,4,5
4 PRINT6
6 FORMAT(9HMORE DATA)
PAUSE
GOTO10
5 LTH=LT/10
LTM=LT-LTH*10
TN=SQRTF(1.0-TL*TL-TM*TM)
RN=SQRTF(1.0-RL*RL-RM*RM)
D08K=1,3
8 U(K)=-U(K)
IR=WH/TN+0.5
IH=WH+0.5
M=LTH+1
MP1=M+1
IF(MP1-25)45,46,46
46 MP1=1
45 TIN=LTM
TIN=TIN/60.0
DELH(1)=-X(1)*RN
DELH(2)=0
DELH(3)=-X(2)*RN
INUM=0
K=1
18 H(K)=WH+DELH(K)
IF(H(K)-74.)75,72,72
72 IF(H(K)-106.)73,73,75
75 PRINT80,UR,H(K)
80 FORMAT(F7.0,2X22HHEIGHT UNACCEPTABLE ATF6.0,3H KM)
GOTO1
73 IF(H(K)-90.)20,19,19
19 VGE=(VE3(M)-VE2(M))*0.1
VGE=VGE+(VE3(MP1)+VE2(M)-VE2(MP1)-VE3(M))*TIN*0.1
VGN=(VN3(M)-VN2(M))*0.1
VGN=VGN+(VN3(MP1)+VN2(M)-VN2(MP1)-VN3(M))*TIN*0.1
GOTO21
20 VGE=(VE2(M)-VE1(M))*0.1
VGE=VGE+(VE2(MP1)+VE1(M)-VE1(MP1)-VE2(M))*TIN*0.1
VGN=(VN2(M)-VN1(M))*0.1
VGN=VGN+(VN2(MP1)+VN1(M)-VN1(MP1)-VN2(M))*TIN*0.1
21 DELHM=H(K)-90.
VE=VE2(M)+VGE*DELHM+(VE2(MP1)-VE2(M))*TIN
VN=VN2(M)+VGN*DELHM+(VN2(MP1)-VN2(M))*TIN
UMEAN(K)=VE*TL+VN*TM

```

```
      IF(SENSESWITCH1)304,299
299  IF(UMEAN(K))300,301,301
300  IF(U(K))304,304,302
301  IF(U(K))302,304,304
302  INUM=INUM+1
      IF(INUM- 3)304,303,303
303PRINT902,UR,UMEAN(1),UMEAN(2),UMEAN(3),U(1),U(2),U(3),LTH

902  FORMAT(F7.0,2X6F5.0,15)
304  UT(K)=U(K)-UMEAN(K)
      K=K+1
      IF(K- 3)18,18,22
22   L=TL*100.
      M=TM*100.
      LR=100.0*RL
      MR=100.0*RM
      X(3)=X(2)-X(1)
PUNCH9,UR,MY,MO,JO,LT,UT(1),UT(2),UT(3),L,M,LR,MR,IR,IH,X(1
),X(2),X(3)
9   FORMAT(F7.0,3I3,15,3F5.0,4I3,2I4,3F5.2)
      GOT01
3   STOP
      END
```

```

C      1200B UNIVERSITY OF ADELAIDE PHYSICS DEPT ROPER 31
3
C      PROCESSES 1200A OUTPUT TO PRINT OUT VELOCITY DIFFERENCE
S AND
C      TURBULENT VELOCITIES AS A FUNCTION OF SEPARATION.
C      CALCULATES THE SCALE OF THE ENERGY BEARING EDDIES.
C      CALCULATES THE PARAMETERS OF REFLECTION POINT MOTION.
DIMENSIONUT(3),X(3),A(20),DELSE(20),USEP(3),SUSEP(20),ATUR(3
),BTUR(3)
DIMENSIONSATUR(20),SBTUR(20),NUM(20)
199 READ50
500 FORMAT(30H
PRINT50
D041J=1,20
A(J)=0.
DELSE(J)=0.
SUSEP(J)=0.
SATUR(J)=0.
SBTUR(J)=0.
41 NUM(J)=0
DIV=0.0
EROR1=0.0
EROR2=0.0
NO=0
ROTMA=0.
ROTME=0.
ROTSQ=0.
S=0.
SERGS=0.
SOROT=0.
SUMG=0.0
TOTUR=0.
READ400,CALEH
400 FORMAT(F6.1)
1 READ2,UR,LT,UT(1),UT(2),UT(3),R,X(1),X(2),X(3)
2 FORMAT(F7.0,9X15,3F5.0,12XF4.0,4X3F5.2)
IF(UR)3,4,5
5 USEP(1)=UT(2)-UT(1)
USEP(2)=UT(2)-UT(3)
USEP(3)=UT(1)-UT(3)
ATUR(1)=UT(2)
ATUR(2)=UT(2)
ATUR(3)=UT(1)
BTUR(1)=UT(1)
BTUR(2)=UT(3)
BTUR(3)=UT(3)
L=1

```

```

115 J=1
118 A(J)=1+2*(J-1)
      IF(ABSF(X(L))-0.1*A(J))119,119,120
119 DELSE(J)=DELSE(J)+USEP(L)**2
      SUSEP(J)=SUSEP(J)+ATUR(L)*BTUR(L)
      SATUR(J)=SATUR(J)+ATUR(L)**2
      SBTUR(J)=SBTUR(J)+BTUR(L)**2
      NUM(J)=NUM(J)+1
      TOTUR=TOTUR+UT(L)**2
      S=S+1.0
      IF(J-3)121,123,123
123 ROTIO=R*(USEP(L)/X(L))*1.E-3
      IF(ABSF(ROTMA)-ABSF(ROTIO))70,70,71
70 ROTMA=ROTIO
71 ROTME=ROTME+ROTIO
      ROTSQ=ROTSQ+ROTIO**2
      SOROT=SOROT+1.0
      GOTO121
120 J=J+1
      IF(J-20)118,118,121
121 L=L+1
      IF(L-3)115,115,122
122 NO=NO+1
      GOTO1
4 PRINT820,NO
820 FORMAT(13HNO OF METEORS16//)
      PRINT30
30 FORMAT(39HSEPARATION DELV G DELG RMSGRAD NO/)
      AMAX=0.
      DO663J=1,20
      IF(AMAX-A(J))664,664,663
664 AMAX=A(J)
663 JA=AMAX/2.0+0.55
      DO700J=1,JA
      AUM=NUM(J)
      IF(NUM(J))200,200,201
200 RMSDE=0.
      GOTO202
201 RMSDE=SQRTF(DELSE(J)/AUM)
202 C=J
      A(J)=0.1*(1.0+2.0*(C-1.0))
      GRAD=RMSDE/A(J)
      IF(NUM(J)-9)240,240,241
240 G=0.
      DELG=0.0
      GOTO70
241 G=SUSEP(J)/SQRTF(SATUR(J)*SBTUR(J))

```

```

DELG=(1.0-G*G)/SQRTF(AUM-1.0)
IF(J-12)243,243,70
243 IF(J-3)700,244,244
244 ERGS  =((RMSDE /0.83)**3)/A(J)
SERGS=SERGS+ERGS **2
SUMG=SUMG+(A(J)-0.1)/(1.0-G)**1.5
EROR1=EROR1+(A(J)-0.1)/(1.0-G+DELG)**1.5
EROR2=EROR2+(A(J)-0.1)/(1.0-G-DELG)**1.5
DIV=DIV+1.0
700 PRINT31,A(J),RMSDE,G,DELG,GRAD,NUM(J)
31 FORMAT(1XF6.2,2XF6.2,F5.2,F5.3,1XF6.2,17)
IF(DIV)199,199,755
755 ENERG=SQRTF(SERGS/DIV)
TYD=EXP(LOG(ENERG*CALEH)/3.0)*0.83
ENVEL=SQRTF(TOTUR/S)
SIZE=ENVEL**3/(ENERG *0.57)
ROTME=ROTME/SOROT
ROTSQ=SQRTF(ROTSQ/SOROT)
SUMG=SUMG/DIV
EROR1=EROR1/DIV
EROR2=EROR2/DIV
PRINT35,ENVEL,SIZE,CALEH
35FORMAT(3HUO=F8.2,6H M/SEC, //3HLO=F7.1,3H KM//6HSCALE F6.1,
3H KM//)
PRINT34,ENERG, TYD,ROTME,ROTMA,ROTSQ
34FORMAT(5HERGS=F6.0,11HERGS/GM/SEC, //3HVZ=F6.2,5HM/SEC//3
HRM=3F8.2//)
PRINT36,SUMG,EROR1,EROR2
36 FORMAT(//11HLO FROM G =F7.1,5H KM (2F7.1,2H )//)
PAUSE
GOTO199
3 STOP
END

```



```

C      1200C UNIVERSITY OF ADELAIDE PHYSICS DEPT.  ROPER  313
C      READS OUTPUT FROM 1200A AND PRINTS ZENITH DEPENDENCE
DIMENSION      ZUTUR(20),NN(20),TGZ(20),NNG(20),UT(3),X(3),US
EP(3)
199 READ50
500 FORMAT(30H          )
    PRINT50
    DO41J=1,20
    TGZ(J)=0.
    NN(J)=0
    ZUTUR(J)=0.
41  NNG(J)=0
    NIXZ=0
    NO=0
1  READ2,UR,LT,UT(1),UT(2),UT(3),WL,WM,R,X(1),X(2),X(3)
2  FORMAT(F7.0,9X15,3F5.0,2F3.2,6XF4.0,4X3F5.2)
    IF(UR)3,4,5
5  COZR=SQRTF(1.-WL*WL-WM*WM)
    USEP(1)=UT(2)-UT(1)
    USEP(2)=UT(2)-UT(3)
    USEP(3)=UT(1)-UT(3)
    J=1
128 C=J
    IF(ABSF(COZR)-0.05*C)129,129,131
129 DO130K=1,3
    ZUTUR(J)=ZUTUR(J)+UT(K)**2
130 NN(J)=NN(J)+1
    GOTO132
131 J=J+1
    IF(J-20)128,128,132
132 L=1
447 IF(ABSF(X(L))-0.3)448,449,449
448 NIXZ=NIXZ+1
    GOTO450
449 TGZ(J)=TGZ(J)+(USEP(L)/X(L))**2
    NNG(J)=NNG(J)+1
450 L=L+1
    IF(L-3)447,447,451
451 NO=NO+1
    GOTO1
4  PRINT820,NO
820 FORMAT(13HNO OF METEORS16//)
    PRINT33
33  FORMAT(22H COZR RMSVT RMSTG NO./)
    DO701J=1,20
    ANG=NNG(J)
    IF(NNG(J))250,250,251

```

```
250 RMSTG=0.  
    GOTO252  
251 RMSTG=SQRTF(TGZ(J)/ANG)  
252 CONTINUE  
    AN=NN(J)  
    IF(NN(J))210,210,211  
210 RMSZU=0.  
    GOTO212  
211 RMSZU    =SQRTF(ZUTUR (J)/AN)  
212 CONTINUE  
    C=J  
    C=C/20.0  
701 PRINT32,C,RMSZU,RMSTG,NN(J)  
    32 FORMAT(F5.2,2F6.2,14)  
    PRINT777,NIXZ  
777 FORMAT(//5HNIXZ=16//)  
    PAUSE  
    GOTO199  
    3 STOP  
    END
```

```

C      1200D UNIVERSITY OF ADELAIDE PHYSICS DEPT ROPER 313
C      PROCESSES 1200A OUTPUT TO PRODUCE HEIGHT DEPENDENCE
C      CALCULATES VERTICAL CORRELATION DISTANCE FROM THE TCHEN
SHEAR LAW
DIMENSIOND(20),HDEL(3),VLESS(3),USEP(3),HUSEP(3),DELUH(20
),VERTA(3)
DIMENSIONVERTB(3),SATH(20),SBTH(20),NUMH(20),SUH(20),UT(
3),X(3)
DIMENSIONATURB(3),BTURB(3)
10 READ10
100 FORMAT(30H
PRINT10
DO41J=1,20
DELUH(J)=0.0
SUH(J)=0.0
SATH(J)=0.0
SBTH(J)=0.0
NUMH(J)=0
-41 CONTINUE
A=0.0
DMAX=0.0
CALEM=0.0
SUMC=0.0
NO=0
READ500,ERGS,ENVEL,TYD,CALEH
500 FORMAT(4F8.1)
PRINT500,ERGS,ENVEL,TYD,CALEH
1 READ2,UR,UT(1),UT(2),UT(3),TL,TM,RL,RM,WH,X(1),X(2),X(3
)
2 FORMAT(F7.0,14X3F5.0,4F3.2,4XF4.0,3F5.2)
IF(UR)3,4,5
5 RNU=SQRTF(1.0-RL*RL-RM*RM)
COZR=SQRTF(1.0-TL*TL-TM*TM)
SNZ=SQRTF(1.0-COZR*COZR)
USEP(1)=UT(2)-UT(1)
USEP(2)=UT(2)-UT(3)
USEP(3)=UT(1)-UT(3)
ATURB(1)=UT(2)
ATURB(2)=UT(2)
ATURB(3)=UT(1)
BTURB(1)=UT(1)
BTURB(2)=UT(3)
BTURB(3)=UT(3)
L=1
85 HDEL(L)=X(L)*RNU

```

```

J=1
88 D(J)=1+2*(J-1)
   IF(ABSF(HDEL(L))- .05*D(J))89,89,90
89 DELUH(J)=DELUH(J)+ USEP(L)* USEP(L)
   VERTA(L)=(ABSF(ATURB(L))-TYD*COZR)/SNZ
   VERTB(L)=(ABSF(BTURB(L))-TYD*COZR)/SNZ
   SUH(J)=SUH(J)+VERTA(L)*VERTB(L)
   SATH(J)=SATH(J)+VERTA(L)*VERTA(L)
   SBTH(J)=SBTH(J)+VERTB(L)*VERTB(L)
   NUMH(J)=NUMH(J)+1
   GOT091
90 J=J+1
   IF(J-20)88,88,91
91 L=L+1
   IF(L- 3)85,85,92
92 NO=NO+1
   GOT01
   4 PRINT599,NO
599 FORMAT(13HNO OF METEORS16)
   PRINT60
600 FORMAT(5X20HDELH DELV GH NO/)
   DO161J=1,20
   IF(DMAX-D(J))165,165,161
165 DMAX=D(J)
161 CONTINUE
   JA=DMAX/2.0+0.55
   DO169J=1,JA
   D(J)=1+2*(J-1)
   D(J)=.05*D(J)
169 CONTINUE
   DO700J=1,JA
   AUMH =NUMH(J)
   IF(NUMH(J))400,400,401
400 RMSDH=0.
   GOT0402
401 RMSDH=SQRTF(DELUH(J)/AUMH)
402 CONTINUE
   IF(SUH(J))411,410,411
410 GH=0.
   GOT0412
411 GH =SUH(J)/SQRTF(SATH(J)*SBTH(J))
412 A=(RMSDH/(D(J)**0.6667))*AUMH+A
   SUMC=SUMC+AUMH
700PRINT8,D(J),RMSDH,GH,NUMH(J)
   8 FORMAT(2XF7.2,F6.2,F6.2,14)
   CALEM=(ENVEL*SUMC/A)**1.50
   PRINT777,CALEM,SUMC

```

```
777 FORMAT(//5X6HSCALECF7.1,3H KM,3X1H(F6.0,2H ))  
PAUSE  
GOTO10  
3 STOP  
END
```

```

C      1200E UNIVERSITY OF ADELAIDE  PHYSICS DEPT  ROPER  313

C      PROCESSES 1200A OUTPUT TO PRINT TIME DEPENDENCE
C      PRINTS AND PUNCHES NORTHERLY AND EASTERLY TURBULENT WIN
C      COMPONENTS HOUR BY HOUR.
DIMENSION SUTE(24), SUTN(24)
DIMENSION UT(3), X(3), AM(24), BM(24), STGM(24), USEP(3)
199 READ 50
500 FORMAT(30H )
PRINT 50
DO 10 M=1, 24
  BM(M)=0.0
  SUTE(M)=0.0
  SUTN(M)=0.0
  STGM(M)=0.0
  10 AM(M)=0.0
  NME=0
  NO=0
  READ 100, ERGS, ENVEL, TYD, CALEH
100 FORMAT(4F8.1)
  PRINT 100, ERGS, ENVEL, TYD, CALEH
  1 READ 2, UR, LT, UT(1), UT(2), UT(3), TL, TM, X(1), X(2), X(3)
  2 FORMAT(F7.0, 9X15, 3F5.0, 2F3.2, 14X3F5.2)
  IF(UR) 3, 4, 5
  5 M=LT/10
  M=M+1
  USEP(1)=UT(2)-UT(1)
  USEP(2)=UT(2)-UT(3)
  USEP(3)=UT(1)-UT(3)
  COZR=SQRTF(1.0-TL*TL-TM*TM)
  SNZ=SQRTF(1.0-COZR*COZR)
  DO 469 K=1, 3
  SUTE(M)=SUTE(M)+(((ABSF(UT(K))-TYD*COZR)/SNZ)*(TL/SNZ))
**2
  SUTN(M)=SUTN(M)+(((ABSF(UT(K))-TYD*COZR)/SNZ)*(TM/SNZ))
**2
  BM(M)=BM(M)+1.
469 CONTINUE
  L=1
470 IF(ABSF(X(L))-0.3) 471, 472, 472
471 NME=NME+1
  GOTO 473
472 STGM(M)=STGM(M)+(USEP(L)/X(L))**2
  AM(M)=AM(M)+1.
473 L=L+1

```

```
IF(L- 3)470,470,120
120 NO=NO+1
GOTO1
4 PRINT820,NO
820 FORMAT(//13HNO OF METEORS16//)
PRINT9
9 FORMAT(8X4OHTIME RMSVTE RMSVTN NO RMSTG NO/)

D0700M=1,24
NBM =BM(M)+0.5
IF(BM(M))480,480,481
480 RSUTE=0.0
RSUTN=0.0
GOTO482
481 RSUTE=SQRTF(SUTE(M)/BM(M))
RSUTN=SQRTF(SUTN(M)/BM(M))
482 CONTINUE
NAM =AM(M)+0.5
IF(AM(M))483,483,484
483 RSTGM=0.0
GOTO485
484 RSTGM =SQRTF(STGM(M)/AM(M))
485 CONTINUE
NIM=(M-1)*10
NIME=M*10
LEVEL=2
PUNCH202,LEVEL,RSUTE,RSUTN
202 FORMAT(4X13,6XF5.0,5XF5.0)
700 PRINT8,NIM,NIME,RSUTE,RSUTN,NBM,RSTGM,NAM
8 FORMAT(2X15,2H -15,F7.2,F8.2,16,F7.2,16)
PRINT777,NME
777 FORMAT(//5X3HNME17)
PAUSE
GOTO199
3 STOP
END
```

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